

# **INDEPENDENT ORBITER ASSESSMENT**

## **ASSESSMENT OF THE LANDING/DECELERATION SUBSYSTEM**

**18 MARCH 1988**



MCDONNELL DOUGLAS ASTRONAUTICS COMPANY  
HOUSTON DIVISION

SPACE TRANSPORTATION SYSTEM ENGINEERING AND OPERATIONS SUPPORT

WORKING PAPER NO. 1.0-WP-VA88003-43

INDEPENDENT ORBITER ASSESSMENT  
ASSESSMENT OF THE LANDING/DECELERATION (LDG/DEC)  
SUBSYSTEM FMEA/CIL

18 MARCH 1988

This Working Paper is Submitted to NASA under  
Task Order No. VA86001, Contract NAS 9-17650

PREPARED BY: Robert A. O'Donnell  
R. O'Donnell  
Analyst  
Independent Orbiter  
Assessment

PREPARED BY: William A. Weissinger  
D. Weissinger  
Lead Analyst  
Independent Orbiter  
Assessment

APPROVED BY: J.M. Compton  
J.M. Compton  
Section Manager-FMEA/CIL  
Independent Orbiter  
Assessment

APPROVED BY: G.W. Knori  
G.W. Knori  
Technical Manager  
Independent Orbiter  
Assessment

APPROVED BY: J.I. McPherson  
J.I. McPherson  
Project Manager  
STSEOS





<b>CONTENTS</b>		<b>Page</b>
1.0	EXECUTIVE SUMMARY	1
2.0	INTRODUCTION	3
2.1	Purpose	3
2.2	Scope	3
2.3	Analysis Approach	3
2.4	Ground Rules and Assumptions	4
3.0	SUBSYSTEM DESCRIPTION	5
3.1	Design and Function	5
3.2	Interfaces and Locations	8
3.3	Hierarchy	8
4.0	ASSESSMENT RESULTS	16
5.0	REFERENCES	23
APPENDIX A	ACRONYMS	A-1
APPENDIX B	DEFINITIONS, GROUND RULES, AND ASSUMPTIONS	B-1
B.1	Definitions	B-2
B.2	Project Level Ground Rules and Assumptions	B-4
B.3	Subsystem Specific Ground Rules and Assumptions	B-6
APPENDIX C	ASSESSMENT WORKSHEETS	C-1
APPENDIX D	CRITICAL ITEMS	D-1
APPENDIX E	ANALYSIS WORKSHEETS	E-1
APPENDIX F	NASA FMEA TO IOA WORKSHEET CROSS REFERENCE/RECOMMENDATIONS	F-1

## **List of Figures**

	<b>Page</b>
Figure 1 - LDG/DEC FMEA/CIL ASSESSMENT	2
Figure 2 - LDG/DEC SUBSYSTEM OVERVIEW	9
Figure 3 - NOSE LANDING GEAR - STOWED POSITION	10
Figure 4 - NOSE LANDING GEAR - EXTENDED POSITION	11
Figure 5 - MAIN LANDING GEAR - STOWED POSITION	12
Figure 6 - MAIN LANDING GEAR - EXTENDED POSITION	13
Figure 7 - BRAKE/SKID CONTROL SYSTEM OVERVIEW	14
Figure 8 - LANDING GEAR CONTROL SYSTEM OVERVIEW	15

## **List of Tables**

	<b>Page</b>
Table I - SUMMARY OF IOA FMEA ASSESSMENT	20
Table II - SUMMARY OF IOA CIL ASSESSMENT	20
Table III- SUMMARY OF IOA RECOMMENDED FAILURE CRITICALITIES	21
Table IV - SUMMARY OF IOA RECOMMENDED CRITICAL ITEMS	21
Table V - IOA WORKSHEET NUMBERS	22

Independent Orbiter Assessment  
Assessment of the Landing/Deceleration FMEA/CIL

**1.0 EXECUTIVE SUMMARY**

The McDonnell Douglas Astronautics Company (MDAC) was selected in June 1986 to perform an Independent Orbiter Assessment (IOA) of the Failure Modes and Effects Analysis (FMEA) and Critical Items List (CIL). Direction was given by the STS Orbiter and GFE Projects Office to perform the hardware analysis using the instructions and ground rules defined in NSTS 22206, Instructions for Preparation of FMEA and CIL, 10 October 1986.

The IOA effort first completed an analysis of the Landing / Deceleration (LDG/DEC) hardware, generating draft failure modes and potential critical items. To preserve independence, this analysis was accomplished without reliance upon the results contained within the NASA FMEA/CIL documentation. The IOA results were then compared to the NASA FMEA/CIL baseline with proposed Post 51-L updates included. A resolution of each discrepancy from the comparison is provided through additional analysis as required. This report documents the results of that comparison for the Orbiter LDG/DEC hardware.

The IOA product for the LDG/DEC analysis consisted of 259 failure mode "worksheets" that resulted in 124 potential critical items being identified. Comparison was made to the NASA baseline (as of 19 November 1986) which consisted of 267 FMEA's and 120 CIL items. The comparison determined if there were any results which had been found by the IOA but were not in the NASA baseline. This comparison produced agreement on all but 75 FMEA's which caused differences in 51 CIL items. Figure 1 presents a comparison of the proposed Post 51-L NASA baseline, with the IOA recommended baseline, and any issues.

The issues arose due to differences between the NASA and IOA FMEA/CIL preparation instructions. NASA had used an older ground rules document which has since been superseded by the NSTS 22206 used by the IOA. After comparison, there were no discrepancies found that were not already identified by NASA, and the remaining issues may be attributed to differences in ground rules.

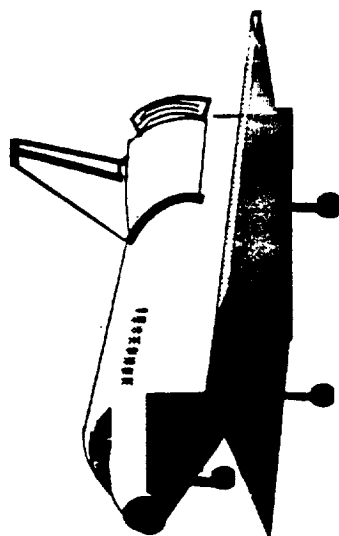
LDG/DEC ASSESSMENT OVERVIEW #			
	IOA	NASA	ISSUES
FMEA	259	267	75
CIL	124	120	51

B & AS		
	IOA	NASA ISSUES
FMEA	28	28
CIL	14	15

EPD&C		
	IOA	NASA ISSUES
FMEA	122	135
CIL	38	44

FLIGHT CONTROLS		
	IOA	NASA ISSUES
FMEA	3	1
CIL	2	1

FMEA COUNT INCLUDES CIL'S



HYD		
	IOA	NASA ISSUES
FMEA	40	42
CIL	16	10

MLG		
	IOA	NASA ISSUES
FMEA	29	40
CIL	23	33

NLG		
	IOA	NASA ISSUES
FMEA	31	15
CIL	25	11

PYRO		
	IOA	NASA ISSUES
FMEA	6	6
CIL	6	6

Figure 1 - LDG/DEC FMEA/CIL ASSESSMENT

## **2.0 INTRODUCTION**

### **2.1 Purpose**

The 51-L Challenger accident prompted the NASA to readdress safety policies, concepts, and rationale being used in the National Space Transportation System (NSTS). The NSTS Office has undertaken the task of re-evaluating the FMEA/CIL for the Space Shuttle design. The MDAC is providing an independent assessment of the proposed Post 51-L Orbiter FMEA/CIL for completeness and technical accuracy.

### **2.2 Scope**

The scope of the independent FMEA/CIL assessment activity encompasses those Shuttle Orbiter subsystems and GFE hardware identified in the Space Shuttle Independent FMEA/CIL Assessment Contractor Statement of Work. Each subsystem analysis addresses hardware, functions, internal and external interfaces, and operational requirements for all mission phases.

### **2.3 Analysis Approach**

The independent analysis approach is a top-down analysis utilizing as-built drawings to breakdown the respective subsystem into components and low-level hardware items. Each hardware item is evaluated for failure mode, effects, and criticality. These data are documented in the respective subsystem analysis report, and are used to assess the proposed Post 51-L NASA and Prime Contractor FMEA/CIL. The IOA analysis approach is summarized in the following Steps 1.0 through 3.0. Step 4.0 summarizes the assessment of the NASA and Prime Contractor FMEA/CIL which is documented in this report.

#### **Step 1.0 Subsystem Familiarization**

- 1.1 Define subsystem functions
- 1.2 Define subsystem components
- 1.3 Define subsystem specific ground rules and assumptions

#### **Step 2.0 Define subsystem analysis diagram**

- 2.1 Define subsystem
- 2.2 Define major assemblies
- 2.3 Develop detailed subsystem representations

#### **Step 3.0 Failure events definition**

- 3.1 Construct matrix of failure modes
- 3.2 Document IOA analysis results

- Step 4.0 Compare IOA analysis data to NASA FMEA/CIL
- 4.1 Resolve differences
  - 4.2 Review in-house
  - 4.3 Document assessment issues
  - 4.4 Forward findings to Project Manager

## **2.4 Ground Rules and Assumptions**

The ground rules and assumptions used in the IOA are defined in Appendix B. The subsystem specific ground rules were defined to limit the analysis to single-failed parts for each failure.

### 3.0 SUBSYSTEM DESCRIPTION

#### 3.1 Design and Function

The Landing / Deceleration Subsystem consists of the hardware required to perform landing and rollout to a safe stop (Figure 2). In addition, the landing / deceleration system performs the function of transporting the Orbiter during the landing phase and towing during post mission operations. The Landing / Deceleration

Subsystem consists of the following components:

1. The Nose Landing Gear Shock Strut Assembly (NGSSA.) is the assembly that supports the nose of the Orbiter during landing and ground handling operations. The NGSSA consists of the Shock Strut, Axle, Steering / Damping Actuator, Torque Arms, Drag Brace, Lock Brace, and attaching hardware (Figures 3 & 4).
2. The Nose Landing Gear Doors and Uplock / Release Mechanisms (Figures 3 & 4) consists of the following components that function when the landing gear deploy switch is activated:
  - o Extend / Retract Hydraulic Strut Actuator
  - o Door Extend Retract Mechanism
  - o Door Over-Center Bungee
  - o Gear Uplock Hook
  - o Door Hooks
  - o Door Hook Actuation Linkage
  - o NLG Uplock Release Hydraulic Actuator
  - o Backup Pyro Uplock Release Actuator
  - o NLG Extension Booster Pyro Actuator
  - o Door Bungee Assist Assembly
3. The data for the Nose Landing Gear Wheels and Tires are not currently available for use in the evaluation of the wheels or the tires. B. F. Goodrich drawings were requested through NASA, Rockwell International - Downey Operations, and through B. F. Goodrich - in Troy, Ohio. Some analysis has been performed using the Rockwell Procurement Specifications which were available through NASA.
4. The two Main Landing Gear Shock Strut Assemblies (MGSSA) support the aft portion of the Orbiter during landing and ground handling activities. The MGSSA consists of the Shock Strut, Axle, Torque Arms, Drag Brace, Lock Brace, and attaching hardware (Figures 5 & 6).

5. The Main Landing Gear Doors and Uplock Mechanisms (Figures 5 & 6) consists of the following components that function when the landing gear deploy switch is activated:
  - o Extend / Retract Hydraulic Strut Actuator
  - o Door Extend Retract Mechanism
  - o Door Over-Center Bungee
  - o Gear Uplock Hook
  - o Door Hooks
  - o Door Hook Actuation Linkage
  - o MLG Uplock Release Hydraulic Actuator
  - o Backup Pyro Uplock Release Actuator
  - o Door Bungee Assist Mechanism
6. The data for the Main Landing Gear Wheels and Tires are not currently available for use in the evaluation of the wheels or the tires. B. F. Goodrich drawings were requested through NASA, Rockwell International - Downey Operations, and through B. F. Goodrich - in Troy, Ohio. Some analysis has been performed using the Rockwell Procurement Specifications which were available through NASA.
7. The data for the Brake and Anti-Skid Controls are limited and the assessment was performed using the data available in the Space Shuttle Systems Handbook, the Shuttle Flight Operations Manual - Volume 10D, the Rockwell Procurement Specification, Brake / Skid Control Subsystem, Wheel Brakes - Main Landing Gear, Orbiter, and the NASA Training Document on Landing / Deceleration systems, LNDG/DECEL 2102. Data were requested on the Mark III Skid Control System, but we were informed that the data were proprietary and that the data would not be made available for the assessment. Current data were requested through NASA and Rockwell International - Downey Operations. The Brake and Anti-Skid Controls consist of the Rudder / Brake Pedal Assembly and the Brake / Skid Control System as identified in Figure 7.
8. The data for the Brake System are not currently available for use in the evaluation of the system. B. F. Goodrich drawings were requested through NASA, Rockwell International - Downey Operations, and through B. F. Goodrich - in Troy, Ohio. Some analysis has been performed using the Rockwell Procurement Specifications which were available through NASA. Some data were found on the Orbiter braking system through Lockheed, in Clear Lake through the notes from the AD HOC COMMITTEE - ORBITER BRAKING SYSTEM ASSESSMENT documents. The brake system consists of four electro-hydraulic disc braking systems. Each assembly has nine discs: four rotors and five stators. The rotors are



splined to the inside of the wheel and they rotate with the wheel. The stators are splined to the outside of the axle assembly and they do not rotate. When the brakes are applied, eight hydraulic actuators in the brake assembly press the discs together, thus providing the braking torque. The hydraulic brake actuators are distributed evenly around the discs. four of the actuators are manifolded into a brake chamber and are powered by a single hydraulic system. The remaining four are manifolded into a second braking chamber and are powered by a different hydraulic system.

9. The Rudder / Brake Pedal Assembly is the mechanical assembly that allows the crew to make manual inputs into the Landing / Deceleration Subsystems. The R/BPA converts the manual inputs into electrical data that is transmitted to the flight control systems, the brake controls and the nose wheel steering. Each rudder / brake pedal assembly contains two brake pedal transducer units called the Rudder Pedal Transducer Assemblies (RPTA). Each unit has four Linear Variable Differential Transducers (LVDT) which output 0-5 VDC brake signals to the brake / skid control boxes A and B. Each of the transducer units output four separate braking signals for the respective left / right brake control for the associated braking system.
10. The Electrical Power Distribution and Control (EPD&C) consists of two subsystems within the Landing / Deceleration Subsystem: Landing Gear Control and Brake and Antiskid. The Landing Gear Control system provides power to the Nose and Main Landing Gear Doors and Uplock Release Mechanisms on the orbiter (Figure 8). The Brake and Antiskid subsystem transfers brake and skid control power to the Brake/Skid Control Boxes A and B (Figure 7). Power is also provided to the hydraulic brake-line heater coils for orbiter hydraulic fluid heating. EPD&C powers the electronics for sensing and monitoring the discrete position of moving parts and assemblies within the Landing / Deceleration subsystem.
11. The responsibility for the Nose Wheel Steering system has been assigned to the NWS Group. The FMEA's for the NWS were originally included in the Landing / Deceleration Subsystem. The mechanical linkage portions of the NWS are still included in the Landing / Deceleration IOA reports as a portion of the Nose Landing Gear reports.

12. The Hydraulics Actuators on the Landing / Deceleration Subsystem consists of six actuators. Three actuators activate the mechanism to release the uplock mechanism to deploy the landing gear mechanisms, and three actuators perform the task of extending or retracting the landing gear. Landing gear retraction can only be performed while the vehicle is being supported by Ground Support Equipment (GSE), a landing gear retraction cannot be supported on orbit. These actuators are actually components of the subsystems listed in sections 2 and 5 of this paragraph. However, for this report they are broken out separately for the purpose of clarity.

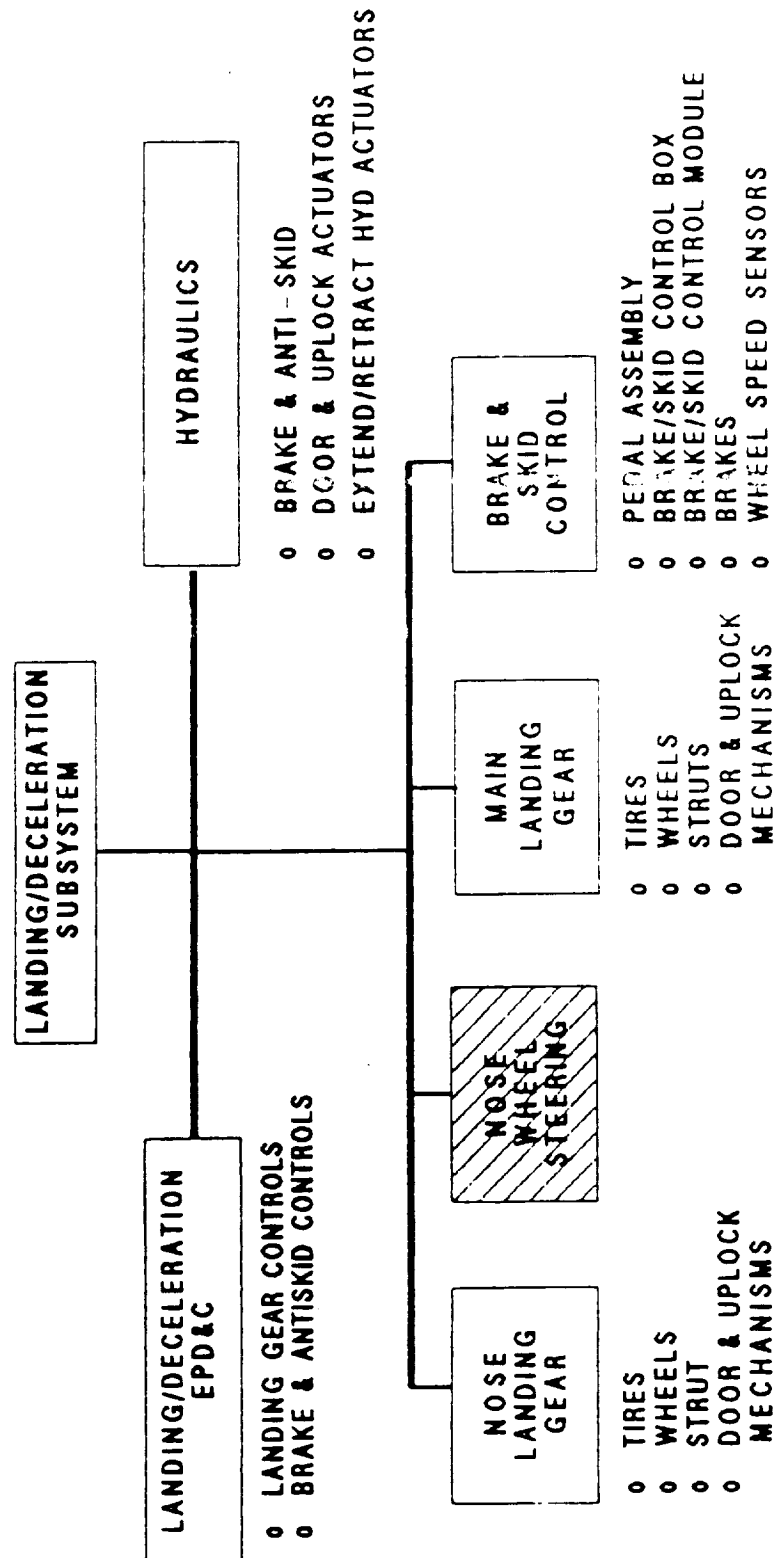
### **3.2 Interfaces and Locations**

The Landing / Deceleration hardware consists of six major subcomponents located in the Orbiter's cabin area and on the under carriage of the Orbiter. The interfaces for the subsystem are relatively simple in that there is a limited number of interfaces with the other subsystems on the Orbiter. The interfaces with the other subsystems are basically limited to interfaces with the Data Processing System (DPS) for backup flight control purposes (NWS) and for instrumentation on the Landing / Deceleration hardware, and the Hydraulics System (HYD) for hydraulics system pressure for the subsystem actuators and for the brakes. The remainder of the subsystem is capable of direct control, via wire, from the control device to the subsystem hardware.

### **3.3 Hierarchy**

Figure 2 illustrates the hierarchy of the Landing / Deceleration Subsystem hardware and the corresponding subcomponents. Figures 3 through 8 comprise the detailed system representations.

# LANDING/DECELERATION OVERVIEW



LANDING/DECELERATION SUBSYSTEM  
NOT CONSIDERED IN THIS REPORT



Figure 2 - LDG/DEC SUBSYSTEM OVERVIEW

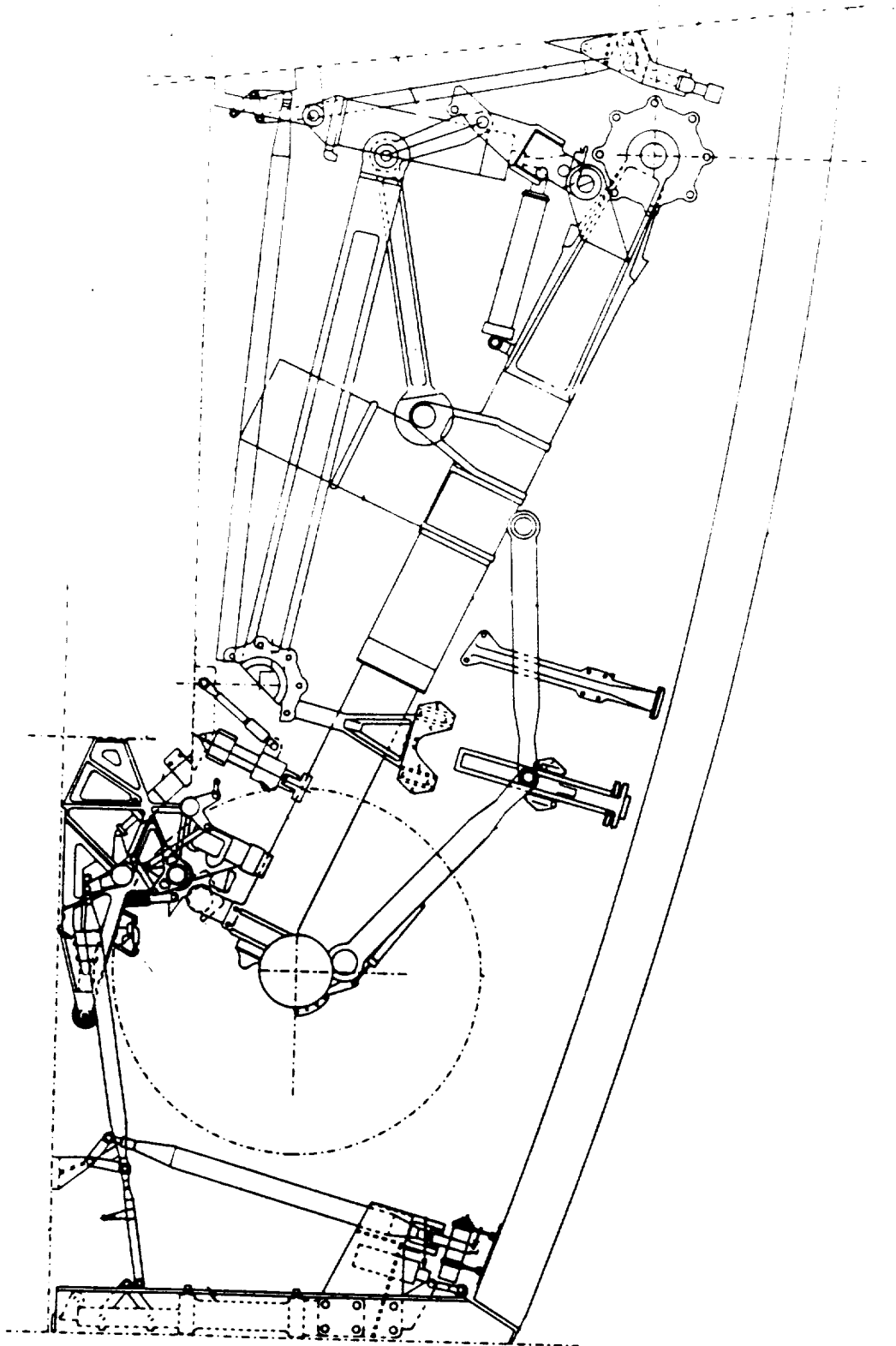
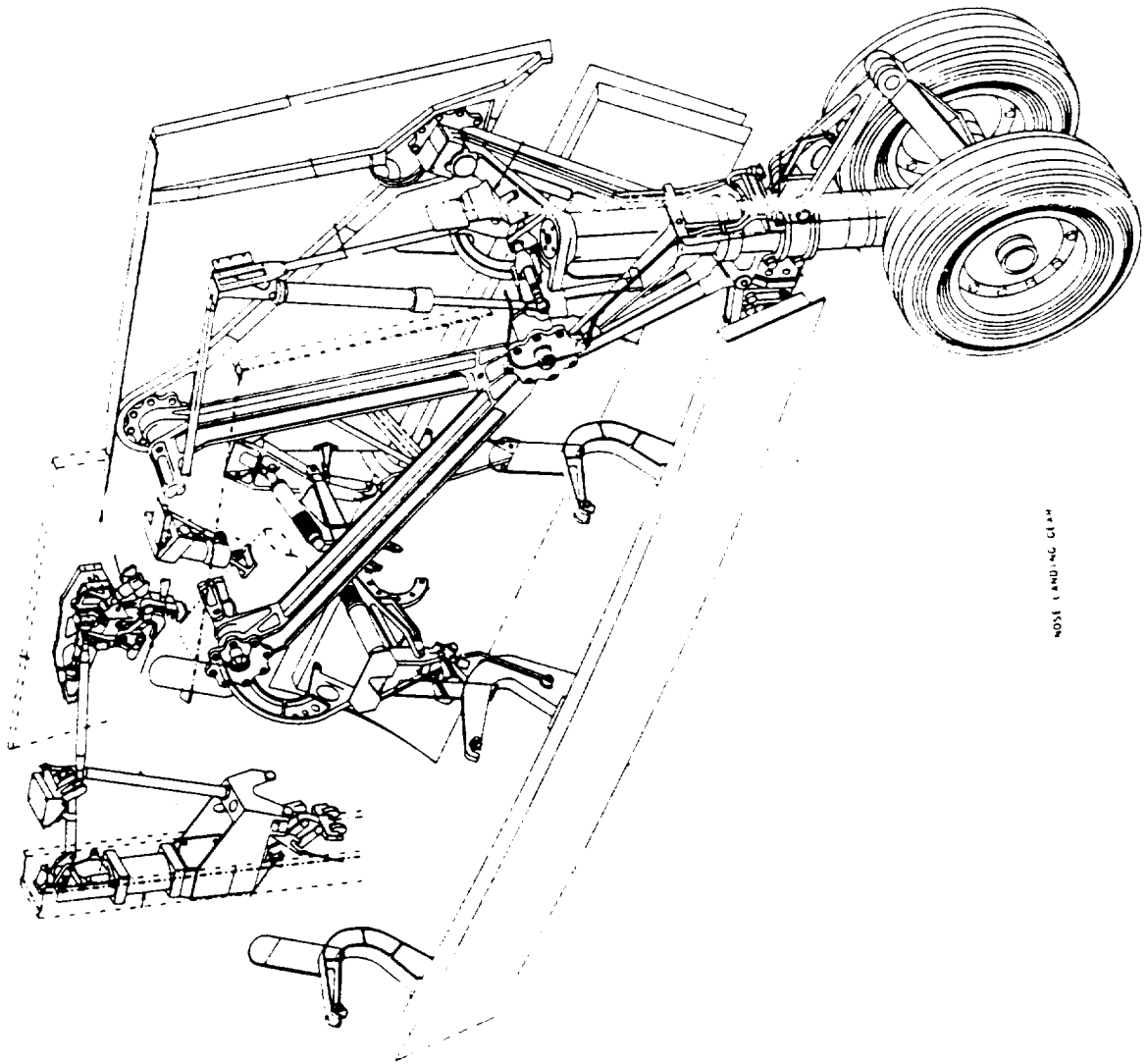
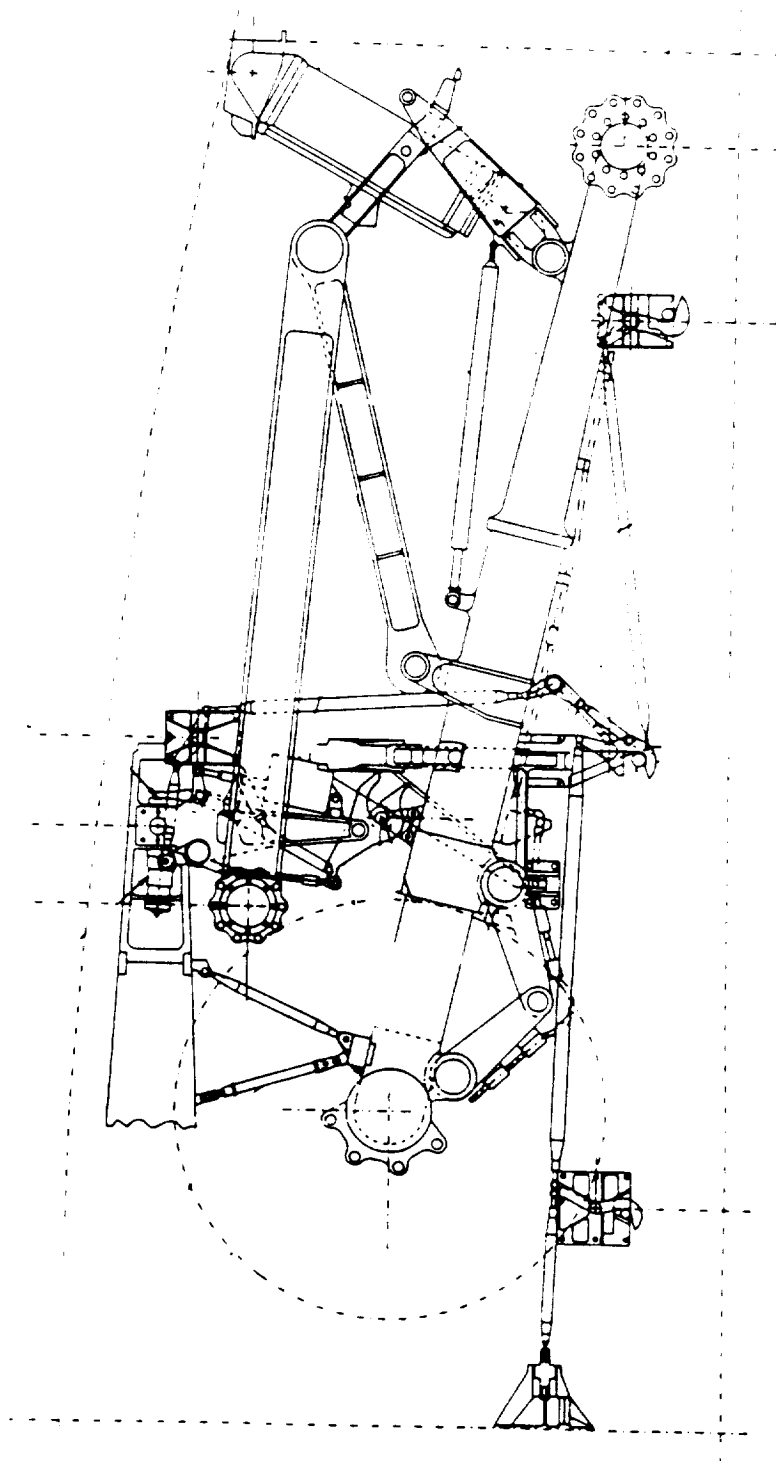


Figure 3 - NOSE LANDING GEAR - STOWED POSITION



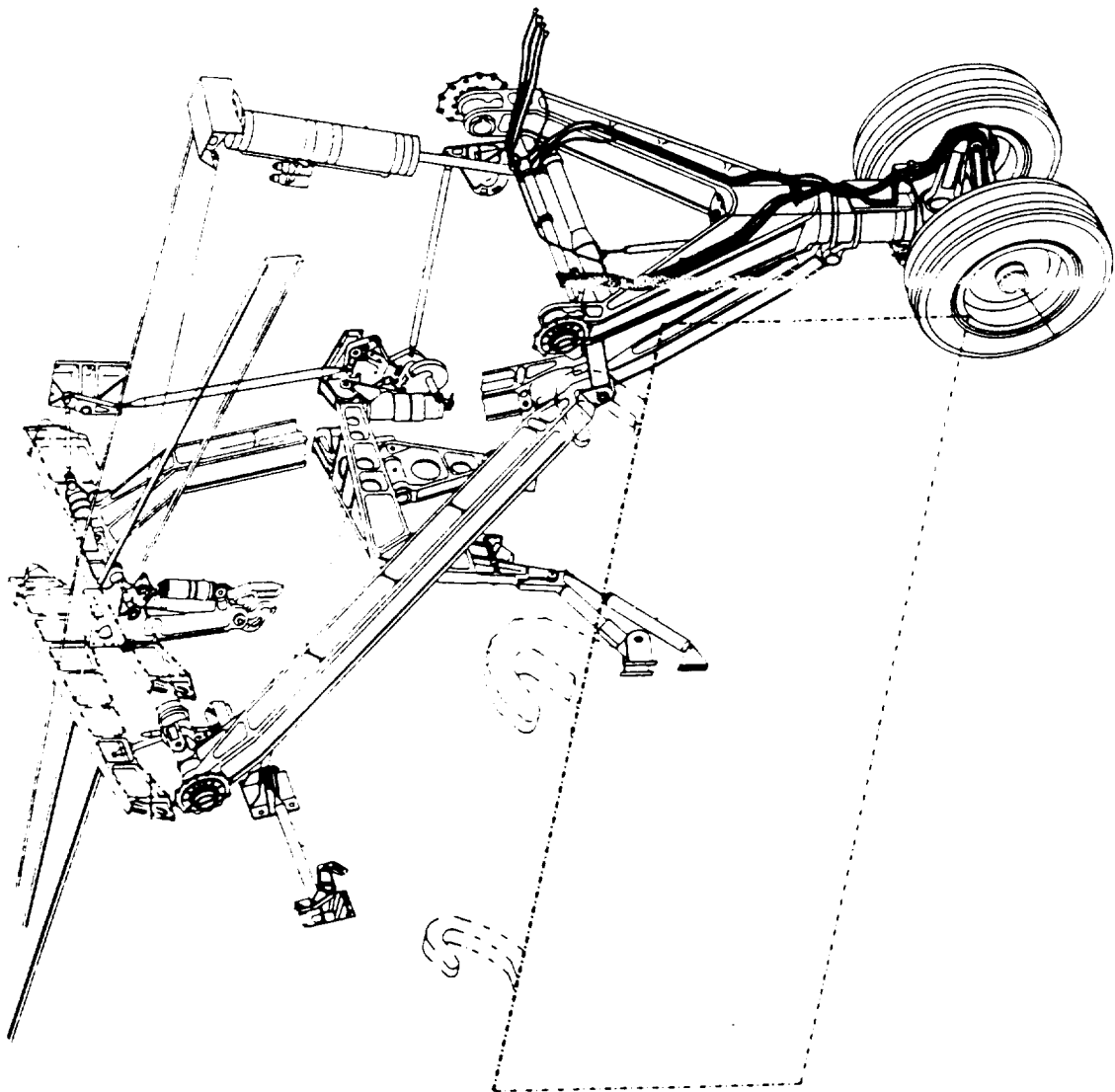
NOSE LANDING GEAR

Figure 4 - NOSE LANDING GEAR - EXTENDED POSITION



MAIN LANDING GEAR STOWED

Figure 5 - MAIN LANDING GEAR - STOWED POSITION



MAIN LANDING GEAR

Figure 6 - MAIN LANDING GEAR - EXTENDED POSITION

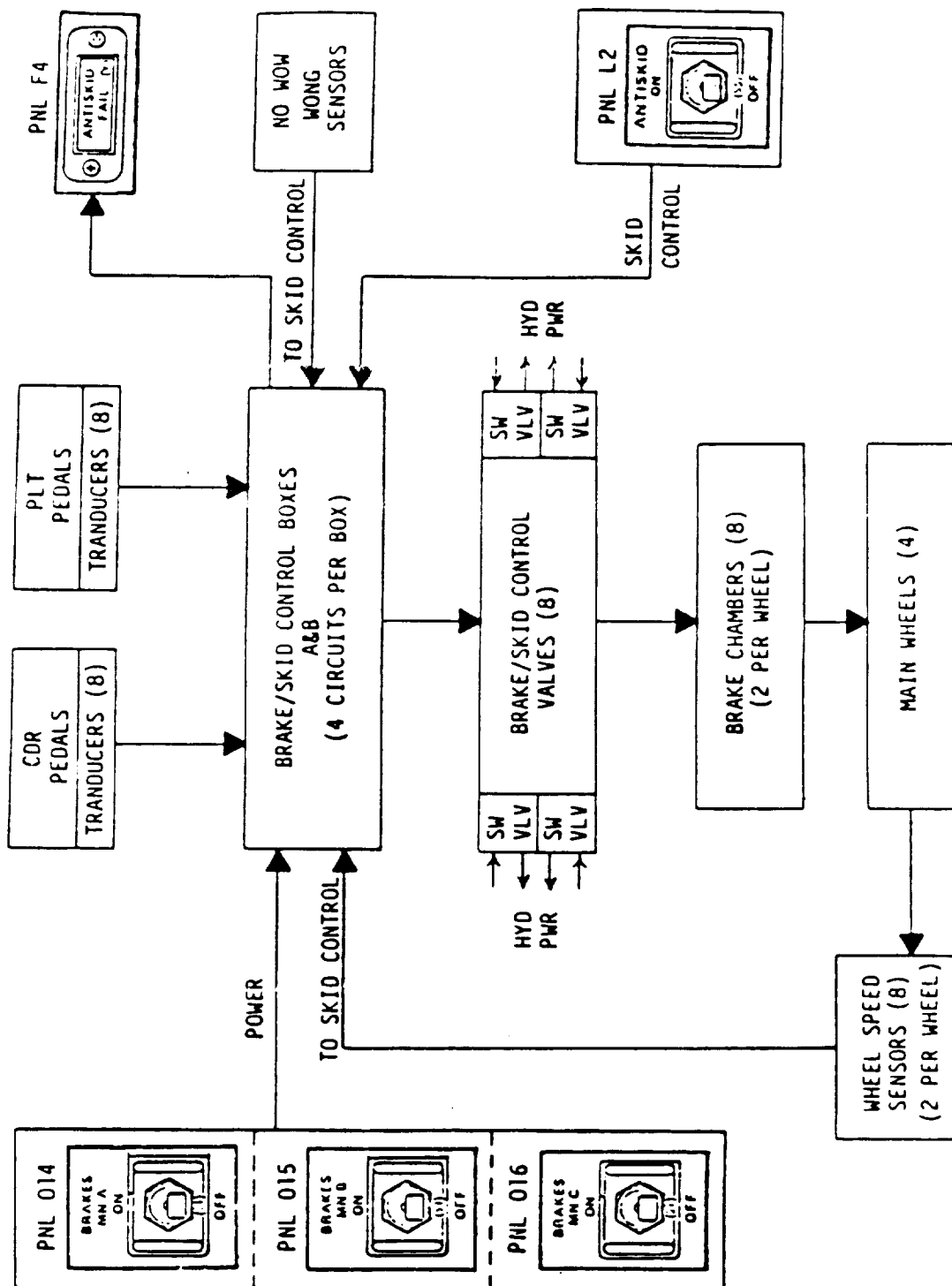


Figure 7 - BRAKE / SKID CONTROL SYSTEM OVERVIEW



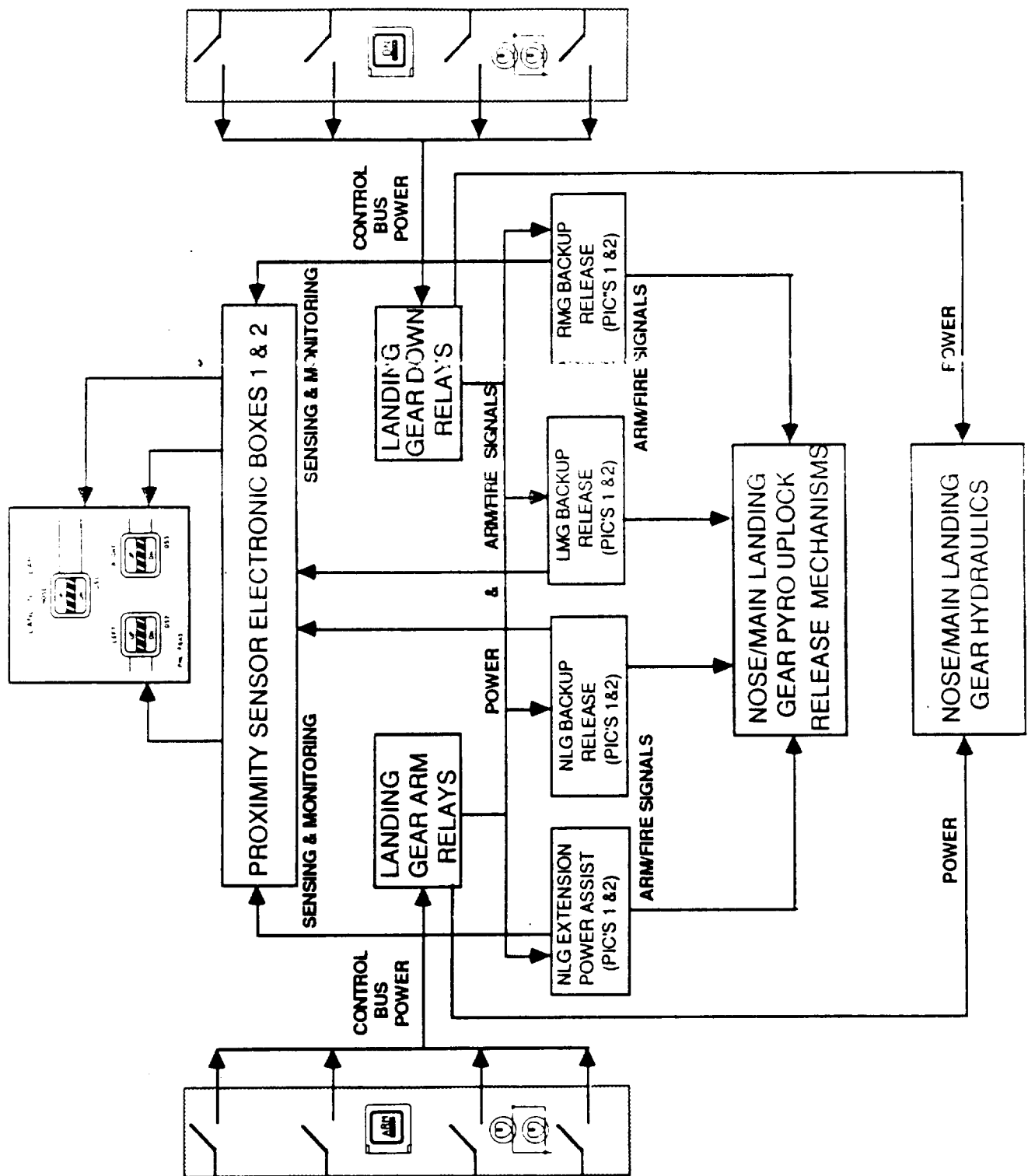


Figure 8 - LANDING GEAR CONTROL SYSTEM OVERVIEW

#### 4.0 ASSESSMENT RESULTS

The IOA analysis of the Landing / Deceleration hardware initially generated 256 failure mode worksheets and identified 124 Potential Critical Items (PCIs) before starting the assessment process. The IOA supported the first NASA sponsored Landing / Deceleration hardware working group meeting on 20-24 January 1987. During the meeting a great deal of work was done to support the development of the OMRSD data and to support the tracking of the OMRSD data with the FMEA/CIL data. Meeting minutes and reports were requested during the meeting, but no minutes or reports were received by the IOA for an analysis. During May 1987, the IOA personnel went directly to the Subsystem Manager, and subsequently to his supervisor in order to have data released. Rough data was obtained and the data that the IOA personnel received was stamped as preliminary data and verbal instructions were given that the data not be used for a comparison of FMEA/CIL documentation. Additional attempts to obtain hardware data were unsuccessful, success was achieved in obtaining preliminary EPD&C in October 1987. The hardware portion of the NASA analysis was not completed until January 1988, there was insufficient time available to perform an analysis and prepare a comparison of the results.

In the analysis report, the Landing / Deceleration Subsystem was divided into six separate functional areas according to hardware and function. Difficulty was encountered in the hardware analysis due to the large amounts of proprietary data or unobtainable data on the brakes and skid control hardware, the tires and wheels, and many of the mechanisms of the landing gear and the hydraulics systems. The initial NASA document, STS 82-0013, consisted of five separate functional areas which included one hundred eighteen (118) FMEA/CIL's. After the initial definition of the subsystem the thirty two (32) NWS FMEA's were removed and a separate group was initiated to prepare the analysis for that subsystem. A decision was made to include the EPD&C data for the subsystem and one hundred twenty two (122) Electrical FMEA's were added to the subsystem. In November 1986 forty four (44) Hydraulics FMEA's were added to the subsystem. After the initial IOA Analasys was completed in January 1987, a decision was made to remove the pyrotechnic devices from the subsystem, which removed six FMEA's from the NLG and MLG subsystems. The seven areas of the Landing / Deceleration analysis that have been encompassed in this report and there status are as follows:

##### BRAKES AND ANTI-SKID (B&AS)

Forty two (42) FMEA's on the Brakes and Anti-Skid were included in the original data package. The study of the Anti-Skid system was accomplished utilizing data that was available through NASA. Difficulty was encountered in the preparation of the data on the actual brake hardware and the wheels and tires, for there was no data available through NASA to study or review, a large amount of the data involved

is proprietary. Data on the brakes and tires is controlled by B.F. Goodrich, and the data on the antiskid subsystem is maintained by Crane Industries. An analysis on the Brakes, Wheels and Tires was not fully accomplished due to the lack of data available.

#### ELECTRICAL POWER DISTRIBUTION AND CONTROL (EPD&C)

EPD&C provides power to the Landing Gear Control Subsystem, Brake and Anti-Skid subsystem, and to the sensing and monitoring functions within the Landing / Deceleration Subsystem. The IOA Analysis generated 114 assessment worksheets associated with credible failure modes and defined criticality. Of the identified failure modes eleven (11) are criticality 2/1R, forty-six (46) are criticality 3/1R, and fifty seven (57) are criticality 3/3. Thirty one (31) failure modes are identified as PCI's. These PCI's are listed in Appendix D. The assessment between the IOA EPD&C worksheets and NASA Post 51-L FMEA/CIL (PRCB Review Presentation 1/25/88) Produced forty (40) issues. IOA recommends downgrading the criticality of nine (9) FMEA's, three (3) of which will be removed from the CIL. IOA recommends that fifteen (15) of the NASA baseline FMEA's be deleted because they represent non-credible failure modes for those particular components. These deletions would remove seven (7) items from the CIL. IOA recommends changes in the redundancy screens for six (6) NASA FMEA's. Resulting in two new additional CIL's. IOA also recommends the addition of fifteen (15) FMEA's to the NASA baseline for failure modes not present in the NASA baseline. One (1) of the additions will require a new CIL. IOA generated one (1) new analysis sheet to correspond to a NASA baseline FMEA not covered in the original IOA analysis. The IOA analysis agreed with the NASA failure mode and criticality evaluation.

#### FLIGHT CONTROLS

Originally, only one (1) FMEA was written on the flight controls and that was written against the Pedal Assembly.

#### HYDRAULICS (HYD)

Originally, no Hydraulics FMEAs were included in the Landing / Deceleration subsystem. Forty-four (44) FMEA's from the Hydraulics system were added to the subsystem in order to conform with the NASA configuration. The information about the addition of the Hydraulics data was transmitted to the IOA Subsystem manager in late November 1986 just prior to the initial submission of the data to the IOA Data Management Group. The Landing / Deceleration subsystem assumed responsibility for the actuators and the hydraulic

lines from the point where the hydraulic lines entered the landing gear compartment.

#### MAIN LANDING GEAR (MLG)

Twenty five (25) FMEA's were written against the MLG. Two (2) Pyrotechnic FMEA's were removed from this system after the IOA analysis was completed, these analysis have remained in this report.

#### NOSE LANDING GEAR (NLG)

Nineteen (19) FMEA's were written against the NLG. Four (4) Pyrotechnic FMEA's were removed from this system after the IOA analysis was completed, these analysis have remained in this report.

#### NOSE WHEEL STEERING (NWS)

Thirty two (32) FMEA's on the NWS were originally included in the Landing / Deceleration subsystem, these FMEA's were transferred to the NWS analysis group in order to align with the NASA configuration.

The following Pyrotechnic (PYRO) data was covered by the initial Independent Orbiter Assessment, and the data has been included in the initial assessment report. The data is not covered in the NASA Landing / Deceleration report.

FMEA NUMBER	IOA ASSESSMENT NUMBER
02-1-015-1	11102
02-1-015-2	11101
02-1-097-1	21101
02-1-097-2	21102
02-1-104-1	11202
02-1-104-2	11201

The data required to perform an analysis was not available for the following FMEA's, so no analysis was performed rather than preparing data with no documentation to support a position.

FMEA NUMBER	
02-1-011-1	CLAMP FLEX LINE BRAKES
02-1-031-1	DISPLACEMENT LIMITER
02-1-056-1	THERMAL RELIEF PLUG MLG
02-1-066-1	WHEEL BRAKE ASSEMBLY
02-1-067-1	WHEEL TIE BOLTS, MLG
02-1-068-1	WHEEL ASSEMBLY, MLG
02-1-068-2	WHEEL ASSEMBLY, MLG
02-1-068-3	WHEEL ASSEMBLY, MLG
02-1-069-1	WHEEL ASSEMBLY OVERINFLATION PLUG MLG
02-1-070-1	WHEEL TIE BOLTS - NLG
02-1-071-1	WHEEL - NLG
02-1-071-2	WHEEL - NLG
02-1-105-1	NLG THERMAL RELIEF PLUG
02-1-110-2	NLG TIRES

The following Nose Wheel Steering (NWS) data were removed from the Landing / Deceleration Subsystem responsibility in order to align with the activities at NASA. A separate IOA team evaluated the NWS data.

02-1-084-1	02-1-090-2	02-1-096-2
02-1-086-1	02-1-091-1	02-1-100-1
02-1-086-2	02-1-091-2	02-1-100-2 CIL ONLY
02-1-087-1	02-1-092-1	02-1-101-1 CIL ONLY
02-1-087-2	02-1-092-2	02-1-101-2 CIL ONLY
02-1-088-1	02-1-093-1	02-1-106-2 CIL ONLY
02-1-088-2	02-1-093-2	02-1-SPA-1
02-1-089-1	02-1-094-1	02-1-SPA-2
02-1-089-2	02-1-094-2	02-1-SPT-1
02-1-089-3	02-1-095-1	02-1-SPT-2
02-1-090-1	02-1-096-1	

The IOA analysis of the LDG/DEC hardware generated 259 failure mode worksheets and identified 124 Potential Critical Items (PCI's). These analysis results were compared to the proposed NASA Post 51-L baseline of 267 FMEA's including 120 CIL items, which were generated using the NSTS 22206 FMEA/CIL instructions. Upon completion of the assessment, there were 75 issues with 51 pertaining to CIL items.

A summary of the quantity of NASA FMEAs assessed, versus the recommended IOA baseline, and any issues identified is presented in Table I.

Table I SUMMARY OF IOA FMEA ASSESSMENT			
Component	NASA	IOA	Issues
B&AS	28	28	9
EPD&C	135	122	40
Flight Controls	1	3	-
HYD	42	40	2
MLG	40	29	11
NLG	15	31	13
PYRO	6	6	-
TOTAL	267	259	75

FMEA count includes CIL's

A summary of the quantity of NASA CIL items assessed, versus the recommended IOA baseline, and any issues identified is presented in Table II.

Table II SUMMARY OF IOA CIL ASSESSMENT			
Component	NASA	IOA	Issues
B&AS	15	14	9
EPD&C	44	38	16
Flight Controls	1	2	-
HYD	10	16	2
MLG	33	23	11
NLG	11	25	13
PYRO	6	6	-
TOTAL	120	124	51

Appendix C presents the detailed assessment worksheets for each failure mode identified and assessed. Appendix D highlights the NASA Critical Items and corresponding IOA worksheet ID. Appendix E contains IOA analysis worksheets supplementing previous analysis results reported in Space Transportation System Engineering and Operations Support (STSEOS) Working Paper No. 1.0-WP-VA86001-25, Analysis of the Landing/Deceleration Subsystem (19 January 1987). Appendix F provides a cross reference between the NASA FMEA and corresponding IOA worksheet(s). IOA recommendations are also summarized.

Table III presents a summary of the IOA recommended failure criticalities for the Post 51-L FMEA baseline. Further discussion of each of these subdivisions and the applicable failure modes is provided in subsequent paragraphs.

Table III SUMMARY OF IOA RECOMMENDED FAILURE CRITICALITIES							
Criticality:	1/1	2/1R	2/2	3/1R	3/2R	3/3	TOTAL
B&AS	5	7	-	9	-	7	28
EPD&C	-	11	-	52	-	59	122
Flight Cntrls	2	-	-	-	-	1	3
HYD	6	10	-	-	-	24	40
MLG	22	-	-	1	-	6	29
NLG	23	-	-	2	-	6	31
PYRO	6	-	-	-	-	0	6
TOTAL	64	28	-	64	-	103	259

FMEA count includes CIL's

Of the failure modes analyzed, 124 were determined to be critical items. A summary of the IOA recommended critical items is presented in Table IV.

Table IV SUMMARY OF IOA RECOMMENDED FAILURE CRITICALITIES							
Criticality:	1/1	2/1R	2/2	3/1R	3/2R	3/3	TOTAL
B&AS	5	7	-	2	-	-	14
EPD&C	-	11	-	27	-	-	38
Flight Cntrls	2	-	-	-	-	-	2
HYD	6	10	-	-	-	-	16
MLG	22	-	-	1	-	-	23
NLG	23	-	-	2	-	-	25
PYRO	6	-	-	-	-	-	6
TOTAL	64	28	-	32	-	-	124

The scheme for assigning IOA assessment (Appendix C) and analysis (Appendix E) worksheet numbers is shown in Table V.

Table V IOA WORKSHEET NUMBERS	
Component	IOA ID Number
B&AS	LDGDEC - 30000
EPD&C	LDGDEC - 31000
Flight Controls	LDGDEC - 30000
HYD	LDGDEC - 10000, 11000, 20000, 21000
MLG	LDGDEC - 20000, 21000
NLG	LDGDEC - 10000, 11000
PYRO	LDGDEC - 11000, 21000



## 5.0 REFERENCES

Reference documentation available from NASA and Rockwell was used in the analysis. The documentation used included the following:

1. AD HOC COMMITTEE - ORBITER BRAKING SYSTEM ASSESSMENT - Volume I, Data Book, January 1984.
2. AD HOC COMMITTEE - ORBITER BRAKING SYSTEM ASSESSMENT - Volume II, Assessment, March 1984.
3. AD HOC COMMITTEE - ORBITER BRAKING SYSTEM ASSESSMENT - Volume III, Interim Review, July 1985.
4. THIRD AD HOC COMMITTEE MEETING - ORBITER BRAKE DEVELOPMENT - July 1986.
5. JSC-12770 Shuttle Flight Operations Manual - Volume 10D. Preliminary, 15-12-78.
6. JSC-18341 Mechanical Systems Console Handbook - Volume II Systems Briefs, Basic, Revision A, PCN-3, 07-02-86.
7. NSTS-22206 Instructions For Preparation Of Failure Modes And Effects Analysis (FMEA) And Critical Items List (CIL), 10 October 1986.
8. MC197-0007 Rockwell Procurement Specification, Tires - Orbiter, 22 February 1984.
9. MC325-0006 Rockwell Procurement Specification, Thruster Assembly, Pyrotechnic, Emergency Nose Gear Uplock Release, 29 February 1984.
10. MC325-0019 Rockwell Procurement Specification, Thruster Assembly, Main Landing Gear - Gear Uplock Release, 6 June 1986.
11. MC621-0011 Rockwell Procurement Specification, Shock Strut Assembly - Main Landing Gear - Orbiter, 8 July 1985.
12. MC621-0012 Rockwell Procurement Specification, Shock Strut Assembly - Nose Landing Gear - Orbiter, 7 May 1981.
13. MC621-0043 Rockwell Procurement Specification, Space Shuttle Flight Control Subsystem, 4 October 1985.
14. MC621-0050 Rockwell Procurement Specification, Wheel Assembly, Nose Landing Gear - Orbiter, 30 March 1984.
15. MC621-0051 Rockwell Procurement Specification, Wheel and Brake Assembly - Main Landing Gear - Orbiter, 21 February 1986.

16. MC621-0055 Rockwell Procurement Specification, Brake / Skid Control Subsystem, Wheel Brakes - Main Landing Gear - Orbiter, 13 January 1986.
17. MC621-0058 Rockwell Procurement Specification, Steering and Damping Subsystem - Nose Landing Gear, 24 October 1985.
18. VO70-510001 Main Landing Gear - Installation. Revision B-13, 28 July 1986.
19. VO70-510101 Booster Assembly, Door Extension - Main Landing Gear. Revision B-08, 10 December 1985.
20. VO70-510201 Mechanical Installation - Main Landing Gear. Revision D-10, 8 July 1986.
21. VO70-510202 Assembly of the Uplock Arm - Main Landing Gear. Revision ?, 10 September 1974.
22. VO70-510300 Fitting Installation - Main Landing Gear. Revision C-05, 1 March 1984.
23. VO70-510301 Uplock Assembly - Main Landing Gear. Revision C-10, 30 November 1984.
24. VO70-510302 Fitting, Uplock, Assembly of, Main Landing Gear. Revision C-06, 20 September 1985.
25. VO70-510346 Hook, Center Door - Assembly of, Main Landing Gear. Revision B, 1 August 1978.
26. VO70-510400 Hook - Door, Assembly of, Main Landing Gear. Revision ?, 5 November 1980.
27. VO70-510476 Fitting, Inboard Trunnion, Assembly of, Main Landing gear. Revision ?, 26 April 1986.
28. VO70-510501 Installation - Nose Landing Gear. Revision C-05, 30 July 1986.
29. VO70-510502 Chassis Assembly - Nose Landing Gear. Revision E-09, 29 July 1986.
30. VO70-510550 Uplock Assembly - Nose Landing Gear. Revision B-10, 7 November 1985.
31. VO70-510601 Doors and Mechanical Installation - Nose Landing Gear. Revision E-24, 19 February 1985.
32. VO70-510711 Lock Assembly - Aft Door, Nose Landing Gear. Revision A-03, 9 February 1978.
33. VO70-510751 Bungee Assembly - Thruster, Nose Landing Gear. Revision B-05, 12 February 1982.

34. VO70-552001 Cartridge Installation - Nose Landing Gear Thrusters.
35. VO70-573001 Mechanical Installation - Yaw & Brake Control Pedals. Revision C-11, 25 October 1985.
36. VS70-510109 Schematic Diagram - Landing Gear Control Subsystem. Revision E-01, 6 June 1983.
37. VS70-510209 Schematic Diagram - Nose Wheel Steering Subsystem. Revision F-01, 23 August 1985.
38. VS70-520109 Schematic Diagram - Brake and Skid Control Subsystem. Revision E-01, 22 August 1985.
39. VS70-790149 Schematic Diagram - Rudder Pedal Transducer Assembly - Flight Control Subsystem. Revision ?, 1 December 1984.
40. 1170100 MENASCO - Shock Strut Assembly - Main Landing Gear - Orbiter. Revision 2-H, Date Unreadable.
41. 1170101 MENASCO - Cylinder Assembly, Shock Strut - Main Landing Gear - Orbiter. Revision D, Date Unreadable.
42. 1170114 MENASCO - Pin Meetering, Shock Strut - Main Landing Gear - Orbiter. Revision C, Date Unreadable.
43. 1170182 MENASCO - Axle Assembly, Shock Strut - Main Landing Gear - Orbiter. Revision A. 7 November 1984. (Reference Only)
44. 1170300 MENASCO - Drag Brace Assembly - Main Landing Gear - Orbiter. Revision D, Date Unreadable.
45. 1170301 MENASCO - Drag Brace Assembly - Lower - Main Landing Gear - Orbiter. Revision A, Date Unreadable.
46. 1170350 MENASCO - Lock Brace Assembly - Main Landing Gear - Orbiter. Revision E, 20 July 1976.
47. 1170493 MENASCO - Layout - Shock Strut - Main Landing Gear - Orbiter. Revision F, Date Unreadable.
48. Landing / Deceleration - LDG/DECEL 2102 - Training Document. 22-02-83
49. MC287-0034 Rockwell Procurement Specification, Actuator Strut, Landing Gear, Hydraulic. Revision H-6, 1 August 1983.



## APPENDIX A ACRONYMS

AOA	- Abort-Once-Around
ATO	- Abort-To-Orbit
B&AS	- Brakes and Antiskid
BFC	- Backup Flight Control
BFS	- Backup Flight System
BITE	- Built-In Test Equipment
C&W	- Caution and Warning
CIL	- Critical Items List
CPU	- Central Processing Unit
CRT	- Cathode-Ray Tube
D/A	- Digital to Analog
DPS	- Data Processing System (Subsystem)
EPD&C	- Electrical Power Distribution and Control
EVA	- Extravehicular Activity
FMEA	- Failure Modes and Effects Analysis
GFE	- Government Furnished Equipment
GPC	- General Purpose Computer
GSE	- Ground Support Equipment
HDC	- Hybrid Driver Controller
HYD	- Hydraulics
IOA	- Independent Orbiter Assessment
LCA	- Load Controller Assembly
LDG/DEC	- Landing/Deceleration
LVDT	- Linear Variable Differential Transformer
MDAC	- McDonnell Douglas Astronautics Company
MGSSA	- Main Gear Shock Strut Assembly
MLG	- Main Landing Gear
NA	- Not Applicable
NASA	- National Aeronautics and Space Administration
NGSSA	- Nose Landing Gear Shock Strut Assembly
NLG	- Nose Landing Gear
NO	- Number
NSTS	- National Space Transportation System
NWS	- Nose-Wheel Steering
OMRSD	- Operational Maintenance Requirements and Specifications Document
OPS	- Operations Sequence
PCA	- Power Control Assembly
PCI	- Potential Critical Item
PIC	- Pyro Initiator Controller
PYRO	- Pyrotechnic
R/BPA	- Rudder/Pedal Brake Assembly
REG	- Regulate, Regulator
RI	- Rockwell International
RPTA	- Rudder Pedal Transducer Assembly
RTLS	- Return-to-Landing Site

## ACRONYMS

SFTWE	- Software
STS	- Space Transportation System
STSEOS	- Space Transportation System Engineering and Operations Support
TAL	- Transatlantic Abort Landing
TD	- Touch Down
THC	- Thruster Hand Controller
VAC	- Volts, ac
VDC	- Volts, dc
WONG	- Weight on Nose Gear
WOW	- Weight on Wheels

## **APPENDIX B**

### **DEFINITIONS, GROUND RULES, AND ASSUMPTIONS**

- B.1 Definitions
- B.2 Project Level Ground Rules and Assumptions
- B.3 Subsystem-Specific Ground Rules and Assumptions

**APPENDIX B**  
**DEFINITIONS, GROUND RULES, AND ASSUMPTIONS**

**B.1 Definitions**

Definitions contained in NSTS 22206, Instructions For Preparation of FMEA/CIL, 10 October 1986, were used with the following amplifications and additions.

**INTACT ABORT DEFINITIONS:**

RTLS - begins at transition to OPS 6 and ends at transition to OPS 9, post-flight

TAL - begins at declaration of the abort and ends at transition to OPS 9, post-flight

AOA - begins at declaration of the abort and ends at transition to OPS 9, post-flight

ATO - begins at declaration of the abort and ends at transition to OPS 9, post-flight

CREDIBLE (CAUSE) - an event that can be predicted or expected in anticipated operational environmental conditions. Excludes an event where multiple failures must first occur to result in environmental extremes

CONTINGENCY CREW PROCEDURES - procedures that are utilized beyond the standard malfunction procedures, pocket checklists, and cue cards

EARLY MISSION TERMINATION - termination of onorbit phase prior to planned end of mission

EFFECTS/RATIONALE - description of the case which generated the highest criticality

HIGHEST CRITICALITY - the highest functional criticality determined in the phase-by-phase analysis

MAJOR MODE (MM) - major sub-mode of software operational sequence (OPS)

MC - Memory Configuration of Primary Avionics Software System (PASS)

MISSION - assigned performance of a specific Orbiter flight with payload/objective accomplishments including orbit phasing and altitude (excludes secondary payloads such as GAS cans, middeck P/L, etc.)



MULTIPLE ORDER FAILURE - describes the failure due to a single cause or event of all units which perform a necessary (critical) function

OFF-NOMINAL CREW PROCEDURES - procedures that are utilized beyond the standard malfunction procedures, pocket checklists, and cue cards

OPS - software operational sequence

PRIMARY MISSION OBJECTIVES - worst case primary mission objectives are equal to mission objectives

PHASE DEFINITIONS:

PRELAUNCH PHASE - begins at launch count-down Orbiter power-up and ends at moding to OPS Major Mode 102 (liftoff)

LIFTOFF MISSION PHASE - begins at SRB ignition (MM 102) and ends at transition out of OPS 1 (Synonymous with ASCENT)

ONORBIT PHASE - begins at transition to OPS 2 or OPS 8 and ends at transition out of OPS 2 or OPS 8

DEORBIT PHASE - begins at transition to OPS Major Mode 301 and ends at first main landing gear touchdown

LANDING/SAFING PHASE - begins at first main gear touchdown and ends with the completion of post-landing safing operations

**APPENDIX B**  
**DEFINITIONS, GROUND RULES, AND ASSUMPTIONS**

**B.2 IOA Project Level Ground Rules and Assumptions**

The philosophy embodied in NSTS 22206, Instructions for Preparation of FMEA/CIL, 10 October 1986, was employed with the following amplifications and additions.

1. The operational flight software is an accurate implementation of the Flight System Software Requirements (FSSRs).

RATIONALE: Software verification is out-of-scope of this task.

2. After liftoff, any parameter which is monitored by system management (SM) or which drives any part of the Caution and Warning System (C&W) will support passage of Redundancy Screen B for its corresponding hardware item.

RATIONALE: Analysis of on-board parameter availability and/or the actual monitoring by the crew is beyond the scope of this task.

3. Any data employed with flight software is assumed to be functional for the specific vehicle and specific mission being flown.

RATIONALE: Mission data verification is out-of-scope of this task.

4. All hardware (including firmware) is manufactured and assembled to the design specifications/drawings.

RATIONALE: Acceptance and verification testing is designed to detect and identify problems before the item is approved for use.

5. All Flight Data File crew procedures will be assumed performed as written, and will not include human error in their performance.

RATIONALE: Failures caused by human operational error are out-of-scope of this task.

6. All hardware analyses will, as a minimum, be performed at the level of analysis existent within NASA/Prime Contractor Orbiter FMEA/CIL's, and will be permitted to go to greater hardware detail levels but not lesser.

RATIONALE: Comparison of IOA analysis results with other analyses requires that both analyses be performed to a comparable level of detail.

7. Verification that a telemetry parameter is actually monitored during AOS by ground-based personnel is not required.

RATIONALE: Analysis of mission-dependent telemetry availability and/or the actual monitoring of applicable data by ground-based personnel is beyond the scope of this task.

8. The determination of criticalities per phase is based on the worst case effect of a failure for the phase being analyzed. The failure can occur in the phase being analyzed or in any previous phase, whichever produces the worst case effects for the phase of interest.

RATIONALE: Assigning phase criticalities ensures a thorough and complete analysis.

9. Analysis of wire harnesses, cables, and electrical connectors to determine if FMEAs are warranted will not be performed nor FMEAs assessed.

RATIONALE: Analysis was substantially complete prior to NSTS 22206 ground rule redirection.

10. Analysis of welds or brazed joints that cannot be inspected will not be performed nor FMEA's assessed.

RATIONALE: Analysis was substantially complete prior to NSTS 22206 ground rule redirection.

11. Emergency system or hardware will include burst discs and will exclude the EMU Secondary Oxygen Pack (SOP), pressure relief valves and the landing gear pyrotechnics.

RATIONALE: Clarify definition of emergency systems to ensure consistency throughout IOA project.

**APPENDIX B**  
**DEFINITIONS, GROUND RULES, AND ASSUMPTIONS**

**B.3 LDG/DEC-Specific Ground Rules and Assumptions**

The IOA analysis was performed to the component or assembly level of the Landing and Deceleration (LDG/DEC) Subsystem. The analysis considered the worst case effects of the hardware or functional failure on the subsystem, mission, and crew and vehicle safety.

1. Pyrotechnic devices were not considered as emergency devices that were to be used in contingency operations. The Pyrotechnic devices were evaluated according to the usage or the demand for usage, if the devices were demanded and they failed to perform. (Pyrotechnic devices were removed from the NASA Analysis after the completion of the IOA Assessment).

RATIONALE: The NLG extension Booster Pyro Actuator functions every time that the NLG is operated to insure that the system is able to overcome any wind forces that are acting on the landing gear doors.

RATIONALE: The Backup Pyro Uplock Release Actuator is a backup or redundant actuator that operates two seconds after the deploy command is issued, if it does not receive a signal that the Uplock Release Hook has functioned properly.

2. The Landing / Deceleration Subsystem considers that all NORMAL and INTACT ABORT LANDINGS will be initiated during the Deorbit Phase and terminated post landing at the time of vehicle egress.

RATIONALE: Under the IOA specific rules all landings will fall under two phase definitions, the deorbit Phase and the Landing/Safing Phase. The intent of this rule is to simplify the development of the analysis by not requiring an analysis for both operational phases.

3. Component age life will not be considered in the analysis.

RATIONALE: Component age analysis is beyond the scope of this task.

## APPENDIX C DETAILED ASSESSMENT

This section contains the IOA assessment worksheets generated during the assessment of this subsystem. The information on these worksheets facilitates the comparison of the NASA FMEA/CIL (Pre and Post 51-L) to the IOA detailed analysis worksheets included in Appendix E. Each of these worksheets identifies the NASA FMEA being assessed, corresponding MDAC Analysis Worksheet ID (Appendix E), hardware item, criticality, redundancy screens, and recommendations. For each failure mode, the highest assessed hardware and functional criticality is compared and discrepancies noted as "N" in the compare row under the column where the discrepancy occurred.

### LEGEND FOR IOA ASSESSMENT WORKSHEETS

-----

#### Hardware Criticalities:

- 1 = Loss of life or vehicle
- 2 = Loss of mission or next failure of any redundant item (like or unlike) could cause loss of life/vehicle
- 3 = All others

#### Functional Criticalities:

- 1R = Redundant hardware items (like or unlike) all of which, if failed, could cause loss of life or vehicle
- 2R = Redundant hardware items (like or unlike) all of which, if failed, could cause loss of mission

#### Redundancy Screens A, B and C:

- P = Passed Screen
- F = Failed Screen
- NA = Not Applicable

#### NASA Data :

- Baseline = NASA FMEA/CIL
- New = Baseline with Proposed Post 51-L Changes

#### CIL Item :

- X = Included in CIL

#### Compare Row :

- N = Non compare for that column (deviation)

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20201	BASELINE [ X ]
NASA FMEA #: 02-1-001-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             20201  
ITEM:                SHOCK STRUT STRUCTURE

LEAD ANALYST:       W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]
COMPARE	[   /   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[ X ]

## REMARKS:

THE NASA FMEA/CIL ONLY COVERS THE SHOCK STRUT PISTON INNER AND OUTER CYLINDER AND NOT THE REMAINING STRUCTURE OF THE MLGSSA

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20202  
NASA FMEA #: 02-1-001-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20202  
ITEM: SHOCK STRUT PISTON ASSEMBLY

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

CRITICALITY FLIGHT HDW/FUNC		REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[ ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ N /N ]	[ ]	[ ]	[ ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ X ]

## REMARKS:

THE NASA FMEA COVERS ONLY THE LOSS OF NITROGEN.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20203	BASELINE [ X ]
NASA FMEA #: 02-1-001-2	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             20203  
ITEM:                SHOCK STRUT PISTON ASSEMBLY

LEAD ANALYST:       W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 3 /3 ]	[ NA ]	[ NA ]	[ NA ]	[   ] *
IOA	[ 3 /1R ]	[ NA ]	[ NA ]	[ NA ]	[ X ]
COMPARE	[   /N ]	[   ]	[   ]	[   ]	[ N ]

RECOMMENDATIONS:   (If different from NASA)

[ 3 /1R ]	[ NA ]	[ NA ]	[ NA ]	[ A ] (ADD/DELETE)
-----------	--------	--------	--------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

- NASA FMEA ASSUMES LOSS OF NITROGEN ELASTIC MEDIUM ONLY.
- HYD FLUID IS CONSIDERED AS CAPABLE OF ABSORBING A LANDING SHOCK PER MC621-0011.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20101  
NASA FMEA #: 02-1-002-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20101  
ITEM: TIRES, MLG TYPE I

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ ]
INADEQUATE	[ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20206  
NASA FMEA #: 02-1-003-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20206  
ITEM: LOWER DRAG BRACE STRUT

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

## RECOMMENDATIONS: (If different from NASA)

[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ A ]
				(ADD/DELETE)

## \* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

ALSO SEE 20219.

- NASA FMEA CONSIDERS ONLY THE ASSEMBLY AND DOES NOT CONSIDER THE INDIVIDUAL CRITICAL PARTS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20207  
NASA FMEA #: 02-1-003-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20207  
ITEM: UPPER DRAG BRACE TRUNIONS (2 EA)

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

CRITICALITY		REDUNDANCY SCREENS			CIL
FLIGHT					ITEM
HDW/FUNC		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

ALSO SEE 20219.

- NASA FMEA CONSIDERS ONLY THE ASSEMBLY AND DOES NOT CONSIDER THE INDIVIDUAL CRITICAL PARTS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20219	BASELINE [ X ]
NASA FMEA #: 02-1-003-1	NEW [   ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20219  
ITEM: UPPER DRAG BRACE STRUTS (2 EA)

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]
COMPARE	[   /   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ A ] (ADD/DELETE)
----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[ X ]

## REMARKS:

SEE 20206, 20220, 20221 & 20223.

- NASA FMEA 02-1-003-1 DEALS WITH THE DRAG BRACE ASSEMBLY AS A SINGLE PART. THE ASSEMBLY CONSISTS OF SEVERAL COMPONENTS MOST OF WHICH ARE CRITICAL.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20220  
NASA FMEA #: 02-1-003-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20220  
ITEM: CENTER DRAG BRACE TRUNION (AT LOCK BRACE)

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ X ]

## REMARKS:

SEE 20219 & 20221.

- NASA FMEA 02-1-003-1 DEALS WITH THE DRAG BRACE ASSEMBLY AS A SINGLE PART. THE ASSEMBLY CONSISTS OF SEVERAL COMPONENTS MOST OF WHICH ARE CRITICAL.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
 ASSESSMENT ID: LDGDEC-20221  
 NASA FMEA #: 02-1-003-1

NASA DATA:  
 BASELINE [ X ]  
 NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
 MDAC ID: 20221  
 ITEM: LOWER DRAG BRACE TRUNION (ATTACHES TO SHOCK STRUT)

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ ]
INADEQUATE	[ X ]

REMARKS:

- NASA FMEA 02-1-003-1 DEALS WITH THE DRAG BRACE ASSEMBLY AS A SINGLE PART. THE ASSEMBLY CONSISTS OF SEVERAL COMPONENTS MOST OF WHICH ARE CRITICAL.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20204  
NASA FMEA #: 02-1-004-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20204  
ITEM: TORQUE ARM ASSEMBLY

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ] *
IOA	[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20208	BASELINE [ X ]
NASA FMEA #: 02-1-005-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             20208  
ITEM:                LOCK BRACE ASSEMBLY

LEAD ANALYST:       W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]
COMPARE	[   /   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
(ADD/DELETE)				

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

REMARKS:



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20222  
NASA FMEA #: 02-1-005-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20222  
ITEM: LOCK BRACE CENTER TRUNION

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

NASA CONSIDERS THE ASSEMBLY AS A COMPONENT NOT AS THE (5)  
FIVE SINGLE POINT FAILURES THAT COULD OCCUR DUE TO THE THREE  
TRUNIONS AND THE TWO ARMS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20216	BASELINE [ X ]
NASA FMEA #: 02-1-006-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             20216  
ITEM:                UPLOCK ROLLER RETAINING ASSEMBLY

LEAD ANALYST:       W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20211  
NASA FMEA #: 02-1-007-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20211  
ITEM: MLG DOWN AND LOCK SENSORS

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ] *
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

REMARKS:

ALSO SEE 20212.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20209	BASELINE [ X ]
NASA FMEA #: 02-1-008-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             20209  
ITEM:                 DOWN LOCK BUNGEE

LEAD ANALYST:       W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ NA ]	[ P ]	[   ] *
IOA	[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ]
COMPARE	[ N /N ]	[ N ]	[   ]	[ N ]	[ N ]

## RECOMMENDATIONS:   (If different from NASA)

[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ A ] (ADD/DELETE)
----------	--------	--------	--------	-----------------------

## \* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

UNTIL THE SAFETY PIN IS INSTALLED IN THE LOCK BRACE THERE IS A MAJOR PROBLEM. FROM THE TIME THE HYD SYS 1 IS SHUTDOWN UNTIL THE SAFETY IS INSTALLED THERE IS AN IMINENT THREAT OF COLLAPSE.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20210  
NASA FMEA #: 02-1-008-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20210  
ITEM: DOWN LOCK BUNGEE

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ NA]	[ P ]	[ ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ N /N ]	[ N ]	[ ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ ]
INADEQUATE	[ ]

REMARKS:

SEE IOA EFFECTS/RATIONALE.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20212  
NASA FMEA #: 02-1-009-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20212  
ITEM: MLG DOWN AND LOCK SENSORS

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ] *
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20213  
NASA FMEA #: 02-1-009-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20213  
ITEM: MLG DOWN AND LOCK SENSORS

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ] *
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20217	BASELINE [ X ]
NASA FMEA #: 02-1-010-1	NEW [   ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20217  
ITEM: TORQUE TUBE ASSEMBLY

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ] *
IOA	[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

REMARKS:



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20501  
NASA FMEA #: 02-1-012-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20501  
ITEM: DOOR EXTEND / RETRACT MECH

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ] *
IOA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

THIS ASSEMBLY WAS CALLED OUT BY NAME ON THE DRAWING. THE NASA FMEA'S CALL OUT THE VARIOUS RODS, FITTINGS, HOOKS AND TUBES WHICH MAKE UP OUR ASSEMBLIES.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20701	BASELINE [ X ]
NASA FMEA #: 02-1-012-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             20701  
ITEM:                MLG UPLOCK HOOK ASSEMBLY

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]
COMPARE	[   /   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

## REMARKS:

THIS ASSEMBLY WAS CALLED OUT BY NAME ON THE DRAWING.

THE NASA FMEA'S CALL OUT THE VARIOUS RODS, FITTINGS, HOOKS  
AND TUBES WHICH MAKEUP OUR ASSEMBLIES.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20901  
NASA FMEA #: 02-1-012-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20901  
ITEM: DOOR HOOK ACTUATION LINKAGE

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

THIS ASSEMBLY WAS CALLED OUT BY NAME ON THE DRAWING.

THE NASA FMEA'S CALL OUT THE VARIOUS RODS, FITTINGS, HOOKS  
AND TUBES WHICH MAKEUP OUR ASSEMBLIES.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20501A	BASELINE [ X ]
NASA FMEA #: 02-1-013-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             20501  
ITEM:                 DOOR EXTEND / RETRACT MECH

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A           B           C		
NASA	[ 1 /1 ]	[ NA ]   [ NA ]   [ NA ]		[ X ] *
IOA	[ 1 /1 ]	[ NA ]   [ NA ]   [ NA ]		[ X ]
COMPARE	[   /   ]	[   ]   [   ]   [   ]		[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]	
					(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

## REMARKS:

THIS ASSEMBLY WAS CALLED OUT BY NAME ON THE DRAWING. THE NASA FMEA'S CALL OUT THE VARIOUS RODS, FITTINGS, HOOKS AND TUBES WHICH MAKE UP OUR ASSEMBLIES.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20701A  
NASA FMEA #: 02-1-013-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20701  
ITEM: MLG UPLOCK HOOK ASSEMBLY

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

THIS ASSEMBLY WAS CALLED OUT BY NAME ON THE DRAWING.

THE NASA FMEA'S CALL OUT THE VARIOUS RODS, FITTINGS, HOOKS  
AND TUBES WHICH MAKEUP OUR ASSEMBLIES.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20901A	BASELINE [ X ]
NASA FMEA #: 02-1-013-1	NEW [   ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20901  
ITEM: DOOR HOOK ACTUATUON LINKAGE

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]
COMPARE	[   /   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
-----------	-------	-------	-------	-------

(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

## REMARKS:

THIS ASSEMBLY WAS CALLED OUT BY NAME ON THE DRAWING.

THE NASA FMEA'S CALL OUT THE VARIOUS RODS, FITTINGS, HOOKS AND TUBES WHICH MAKEUP OUR ASSEMBLIES.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20501B  
NASA FMEA #: 02-1-014-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20501  
ITEM: DOOR EXTEND / RETRACT MECH

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

CRITICALITY		REDUNDANCY SCREENS			CIL
FLIGHT					ITEM
HDW/FUNC		A	B	C	
NASA	[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ] *
IOA	[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

THIS ASSEMBLY WAS CALLED OUT BY NAME ON THE DRAWING. THE NASA FMEA'S CALL OUT THE VARIOUS RODS, FITTINGS, HOOKS AND TUBES WHICH MAKE UP OUR ASSEMBLIES.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20701B	BASELINE [ X ]
NASA FMEA #: 02-1-014-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             20701  
ITEM:                MLG UPLOCK HOOK ASSEMBLY

LEAD ANALYST:       J. COMPTON

ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]
IOA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

REMARKS:

THIS ASSEMBLY WAS CALLED OUT BY NAME ON THE DRAWING.

THE NASA FMEA'S CALL OUT THE VARIOUS RODS, FITTINGS, HOOKS  
AND TUBES WHICH MAKEUP OUR ASSEMBLIES.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20901B  
NASA FMEA #: 02-1-014-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20901  
ITEM: DOOR HOOK ACTUATION LINKAGE

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ] *
IOA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

THIS ASSEMBLY WAS CALLED OUT BY NAME ON THE DRAWING.

THE NASA FMEA'S CALL OUT THE VARIOUS RODS, FITTINGS, HOOKS  
AND TUBES WHICH MAKEUP OUR ASSEMBLIES.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86 NASA DATA:  
ASSESSMENT ID: LDGDEC-21101 BASELINE [ X ]  
NASA FMEA #: 02-1-015-1 NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 21101  
ITEM: MLG PYRO UPLOCK RELEASE

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-21102  
NASA FMEA #: 02-1-015-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 21102  
ITEM: MLG PYRO UPLOCK RELEASE

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ N /N ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 2 /1R ] [ N ] [ N ] [ N ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ X ]

## REMARKS:

THIS SYSTEM IS NEVER CALLED UPON TO FUNCTION UNLESS THERE IS AN INITIAL HYDRAULICS/MECHANICAL SYSTEM MALFUNCTION THAT INITIATES THE PYRO BACKUP.

ACCORDING TO THE REDUNDANCY RULES IN 22206, THIS SYSTEM IS A 2/1R CRITICALITY, BUT A FAILED HYDRAULICS SYSTEM ACTIVATES THIS SYSTEM. THIS SYSTEMS FAILURE WILL NOT ACTIVATE THE HYDRAULICS. THERE IS A LINEAR OPERATION HERE THAT WILL NOT ALLOW REVERSAL OF THE ROLES.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20218	BASELINE [ X ]
NASA FMEA #: 02-1-017-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             20218  
ITEM:                 SHOCK STRUT ATTACHING TRUNIONS

LEAD ANALYST:       W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]
IOA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20223  
NASA FMEA #: 02-1-018-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20223  
ITEM: SUPPORT BEAM

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

THE SUPPORT BEAM IS THE COMPONENT THAT HOLDS THE TRUNIONS OF THE DRAG BRACE ASSEMBLY (AT THE ORBITER ATTACHING POINTS) IN ALIGNMENT. THE SUPPORT BEAM IS A COMPONENT OF THE DRAG BRACE ASSEMBLY.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20501C	BASELINE [ X ]
NASA FMEA #: 02-1-019-1	NEW [   ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20501  
ITEM: DOOR EXTEND / RETRACT MECH

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ] *
IOA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

## REMARKS:

THIS ASSEMBLY WAS CALLED OUT BY NAME ON THE DRAWING. THE NASA FMEA'S CALL OUT THE VARIOUS RODS, FITTINGS, HOOKS AND TUBES WHICH MAKE UP OUR ASSEMBLIES.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20701C  
NASA FMEA #: 02-1-019-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20701  
ITEM: MLG UPLOCK HOOK ASSEMBLY

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ] (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

THIS ASSEMBLY WAS CALLED OUT BY NAME ON THE DRAWING.

THE NASA FMEA'S CALL OUT THE VARIOUS RODS, FITTINGS, HOOKS  
AND TUBES WHICH MAKEUP OUR ASSEMBLIES.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20901C	BASELINE [ X ]
NASA FMEA #: 02-1-019-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             20901  
ITEM:                DOOR HOOK ACTUATUON LINKAGE

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS	CIL ITEM
	A	B	C
NASA [ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]
IOA [ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]
COMPARE [ / ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[ / ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

## REMARKS:

THIS ASSEMBLY WAS CALLED OUT BY NAME ON THE DRAWING.

THE NASA FMEA'S CALL OUT THE VARIOUS RODS, FITTINGS, HOOKS  
AND TUBES WHICH MAKEUP OUR ASSEMBLIES.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20501D	BASELINE [ X ]
NASA FMEA #: 02-1-020-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             20501  
ITEM:                DOOR EXTEND / RETRACT MECH

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS	CIL ITEM
	A           B           C	
NASA [ 1 /1 ]	[ NA]   [ NA]   [ NA]	[ X ] *
IOA [ 1 /1 ]	[ NA]   [ NA]   [ NA]	[ X ]
COMPARE [   /   ]	[   ]   [   ]   [   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

## REMARKS:

THIS ASSEMBLY WAS CALLED OUT BY NAME ON THE DRAWING. THE NASA FMEA'S CALL OUT THE VARIOUS RODS, FITTINGS, HOOKS AND TUBES WHICH MAKE UP OUR ASSEMBLIES.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20701D	BASELINE [ X ]
NASA FMEA #: 02-1-020-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             20701  
ITEM:                MLG UPLOCK HOOK ASSEMBLY

LEAD ANALYST:       J. COMPTON

ASSESSMENT:

CRITICALITY FLIGHT HDW/FUNC		REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

REMARKS:

THIS ASSEMBLY WAS CALLED OUT BY NAME ON THE DRAWING.

THE NASA FMEA'S CALL OUT THE VARIOUS RODS, FITTINGS, HOOKS AND TUBES WHICH MAKEUP OUR ASSEMBLIES.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20901D	BASELINE [ X ]
NASA FMEA #: 02-1-020-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             20901  
ITEM:                DOOR HOOK ACTUATUON LINKAGE

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]
IOA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

## REMARKS:

THIS ASSEMBLY WAS CALLED OUT BY NAME ON THE DRAWING.

THE NASA FMEA'S CALL OUT THE VARIOUS RODS, FITTINGS, HOOKS  
AND TUBES WHICH MAKEUP OUR ASSEMBLIES.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20501E	BASELINE [ X ]
NASA FMEA #: 02-1-021-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             20501  
ITEM:                DOOR EXTEND / RETRACT MECH

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ] (ADD/DELETE)
-----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

## REMARKS:

THIS ASSEMBLY WAS CALLED OUT BY NAME ON THE DRAWING. THE NASA FMEA'S CALL OUT THE VARIOUS RODS, FITTINGS, HOOKS AND TUBES WHICH MAKE UP OUR ASSEMBLIES.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20701E  
NASA FMEA #: 02-1-021-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20701  
ITEM: MLG UPLOCK HOOK ASSEMBLY

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ] *
IOA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ] (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

THIS ASSEMBLY WAS CALLED OUT BY NAME ON THE DRAWING.

THE NASA FMEA'S CALL OUT THE VARIOUS RODS, FITTINGS, HOOKS  
AND TUBES WHICH MAKEUP OUR ASSEMBLIES.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20901E	BASELINE [ X ]
NASA FMEA #: 02-1-021-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             20901  
ITEM:                DOOR HOOK ACTUATUON LINKAGE

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS	CIL ITEM
		A           B           C	
NASA	[ 1 /1 ]	[ NA]   [ NA]   [ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]   [ NA]   [ NA]	[ X ]
COMPARE	[   /   ]	[   ]   [   ]   [   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ] (ADD/DELETE)
-----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

## REMARKS:

THIS ASSEMBLY WAS CALLED OUT BY NAME ON THE DRAWING.

THE NASA FMEA'S CALL OUT THE VARIOUS RODS, FITTINGS, HOOKS  
AND TUBES WHICH MAKEUP OUR ASSEMBLIES.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-30114  
NASA FMEA #: 02-1-022-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30114  
ITEM: DISPLACEMENT LIMITER, HYD MODULE ASSY

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

CRITICALITY FLIGHT HDW/FUNC		REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-30113	BASELINE [ X ]
NASA FMEA #: 02-1-022-2	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             30113  
ITEM:                DISPLACEMENT LIMITER, HYD MODULE ASSY

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[   ]
COMPARE	[ N /   ]	[   ]	[   ]	[   ]	[ N ]

## RECOMMENDATIONS:   (If different from NASA)

[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ D ] (ADD/DELETE)
-----------	-------	-------	-------	-----------------------

## \* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

LOSS OF ONE DISPLACEMENT LIMITER WILL CAUSE LOSS OF ONE BRAKE LINE. NEXT FAILURE WILL LOSE ANOTHER BRAKE LINE, BUT CREW SHOULD STILL ROLLOUT SAFELY IF RUNWAY LONG ENOUGH.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-30119  
NASA FMEA #: 02-1-023-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30119  
ITEM: SELECTOR VALVE, HYD MODULE ASSY

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[ ]
COMPARE	[ /N ]	[ N ]	[ N ]	[ N ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[ ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

WITH VALVE OPEN, FLUID IS AVAILABLE ASSUMING HYDRAULIC  
SYSTEM OKAY. CONTROL VALVE WILL STILL REGULATE PRESSURE TO  
BRAKES BASED ON DEMAND.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-30117	BASELINE [ X ]
NASA FMEA #: 02-1-023-2	NEW [   ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30117  
ITEM: SELECTOR VALVE, HYD MODULE ASSY

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]
IOA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-30118  
NASA FMEA #: 02-1-023-3

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30118  
ITEM: SELECTOR VALVE, HYD MODULE ASSY

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ X ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IF VALVE JAMS CLOSED BEFORE BRAKES APPLIED, BRAKING WILL NOT BE AVAILABLE THROUGH THIS SYSTEM (HALF BRAKING TO BOTH WHEELS IN WHEEL-WELL). IF BRAKES APPLIED, FLUID WILL BE TRAPPED IN LINES WITH CONTINUOUS PRESSURE.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-30101  
NASA FMEA #: 02-1-024-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30101  
ITEM: ANTI-SKID SELECT SWITCH

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[ ]
COMPARE	[ /N ]	[ N ]	[ N ]	[ N ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

SWITCH IS REDUNDANT, BUT VEHICLE CAN LAND SAFELY WITHOUT ANTISKID.

NO NASA CIL AVAILABLE ON THIS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-30106  
NASA FMEA #: 02-1-025-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30106  
ITEM: BRAKE CIRCUIT

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ] (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-30105	BASELINE [ X ]
NASA FMEA #: 02-1-025-2	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             30105  
ITEM:                 BRAKE CIRCUIT

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A           B           C		
NASA	[ 2 /1R ]	[ P ]   [ F ]   [ P ]		[ X ] *
IOA	[ 1 /1 ]	[ NA ]   [ NA ]   [ NA ]		[ X ]
COMPARE	[ N /N ]	[ N ]   [ N ]   [ N ]		[   ]

RECOMMENDATIONS:   (If different from NASA)

[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ A ] (ADD/DELETE)
----------	--------	--------	--------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[ X ]

## REMARKS:

WITH BRAKE PRESSURE BEING APPLIED AT TOUCHDOWN, TIRE ON THAT WHEEL WILL PROBABLY BLOW RIGHT AFTER TOUCHDOWN CAUSING POSSIBLE LOSS OF VEHICLE.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-30108  
NASA FMEA #: 02-1-026-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30108  
ITEM: SKID CIRCUIT

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ ] *
IOA	[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[ ]
COMPARE	[ /N ]	[ N ]	[ N ]	[ N ]	[ ]

## RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[ ]
				(ADD/DELETE)

## \* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

NO CIL AVAILABLE. WILL RESULT IN LOSS OF 12.5% BRAKING  
CAPABILITY. SYSTEM IS NOT REDUNDANT.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-30107  
NASA FMEA #: 02-1-026-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30107  
ITEM: SKID CIRCUIT

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ X ] *
IOA	[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[ ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ N ]

## RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[ A ]
				(ADD/DELETE)

## \* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ X ]

## REMARKS:

LOSS OF ANTISKID PROTECTION TO ONE BRAKE. SHOULD NOT  
PRESENT A PROBLEM SINCE CREW NOTIFIED OF ANTISKID PROBLEM AND  
SHOULD AVOID HEAVY BRAKING. CIL NOT NEEDED.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-30109  
NASA FMEA #: 02-1-027-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30109  
ITEM: ANTI-SKID FAIL CIRCUIT

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ X ] *
IOA	[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[ ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ X ]

## REMARKS:

CIL NOT REQUIRED. FAIL CIRCUIT MONITORS SYSTEM AND IN WORST CASE DESELECT A GOOD CIRCUIT RESULTING IN REDUCED BRAKING CAPABILITY. CREW NOTIFIED OF A PROBLEM AND SHOULD AVOID HEAVY BRAKING.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-30109A  
NASA FMEA #: 02-1-027-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30109  
ITEM: ANTI-SKID FAIL CIRCUIT

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ X ] *
IOA	[ 3 /3 ]	[ NA ]	[ NA ]	[ NA ]	[ ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ]	[ NA ]	[ NA ]	[ NA ]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ X ]

## REMARKS:

CIL NOT REQUIRED. FAIL CIRCUIT MONITORS SYSTEM AND IN WORST CASE DESELECT A GOOD CIRCUIT RESULTING IN REDUCED BRAKING CAPABILITY. CREW NOTIFIED OF A PROBLEM AND SHOULD AVOID HEAVY BRAKING.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-30122  
NASA FMEA #: 02-1-028-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30122  
ITEM: BRAKE / SKID CONTROL VALVE

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

REMARKS:

NO CIL AVAILABLE.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-30121	BASELINE [ X ]
NASA FMEA #: 02-1-028-2	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             30121  
ITEM:                BRAKE / SKID CONTROL VALVE

LEAD ANALYST:       J. COMPTON

ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]
IOA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-30120  
NASA FMEA #: 02-1-028-3

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30120  
ITEM: BRAKE / SKID CONTROL VALVE

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

REMARKS:  
NO CIL AVAILABLE.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-30115  
NASA FMEA #: 02-1-029-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30115  
ITEM: BY - PASS VALVE, HYD MODULE ASSY

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ] *
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-30116  
NASA FMEA #: 02-1-029-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30116  
ITEM: BY - PASS VALVE, HYD MODULE ASSY

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

CRITICALITY		REDUNDANCY SCREENS			CIL
FLIGHT					ITEM
HDW/FUNC		A	B	C	
NASA	[ 3 /3 ]	[ NA ]	[ NA ]	[ NA ]	[ ] *
IOA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ N ]

## RECOMMENDATIONS: (If different from NASA)

[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ ]
				(ADD/DELETE)

## \* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

THIS VALVE HAS VERY LITTLE VALUE SINCE HYD. SYS. 1 IS SHUTDOWN AND LINE CLOSED OFF DURING FLIGHT. FLUID NOT AVAILABLE TO VALVE UNTIL JUST PRIOR TO LANDING. IF BOTH PRIMARY AND STANDBY SYSTEMS FAIL TO OPEN BOTH SYSTEMS MUST BE SO SLUGGISH THAT THE BRAKES ON THIS CONTROL MODULE WON'T FUNCTION. ALSO SEE 30131.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-30131	BASELINE [ X ]
NASA FMEA #: 02-1-029-2	NEW [   ]
SUBSYSTEM: LANDING/DECELERATION SYSTEMS	
MDAC ID: 30131	
ITEM: BY - PASS VALVE, HYD MODULE ASSY (SYS 2&3)	
LEAD ANALYST: J. COMPTON	

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 3 /3 ]	[ NA ]	[ NA ]	[ NA ]
IOA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]

## RECOMMENDATIONS: (If different from NASA)

[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ A ]
				(ADD/DELETE)

## \* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

SEE 30116. SINCE CIRC PUMPS ARE ON FOR THESE SYSTEMS ONORBIT, THIS FAILURE COULD BE DETECTED INFLIGHT.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-30112  
NASA FMEA #: 02-1-030-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30112  
ITEM: INLET FILTER, HYD MODULE ASSY

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ]
COMPARE	[ N / ]	[ ]	[ ]	[ ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

NO CIL AVAILABLE. SHOULD BE 2/1R BECAUSE IF STANDBY FILTER GETS CLOGGED, HALF BRAKING CAPABILITY TO BRAKES IN THAT WHEEL WELL WILL BE LOST. SEE 30130.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-30130  
NASA FMEA #: 02-1-030-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30130  
ITEM: INLET FILTER, HYD MODULE ASSY (SYS 1)

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ N / ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ 2 /1R ] [ P ] [ F ] [ P ] [ A ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

SEE 30112 - SYS 1 DOES NOT PASS REDUNDANCY SCREEN B.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-30103  
NASA FMEA #: 02-1-032-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30103  
ITEM: BRAKE PEDAL TRANSDUCER

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

NASA CIL NOT AVAILABLE.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-30102  
NASA FMEA #: 02-1-032-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30102  
ITEM: BRAKE PEDAL TRANSDUCER

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-30110  
NASA FMEA #: 02-1-033-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30110  
ITEM: HYDRAULIC PRESSURE REGULATOR

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-30111	BASELINE [ X ]
NASA FMEA #: 02-1-033-2	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             30111  
ITEM:                HYD PRESS REG (SYS 2 & 3)

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[   ] *
IOA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ]
COMPARE	[ N /   ]	[   ]	[   ]	[   ]	[ N ]

RECOMMENDATIONS:   (If different from NASA)

[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

SEE 30129. SHOULD BE A 2 BECAUSE IF STANDBY SYSTEM HAD SOME FAILURE VERY LITTLE BRAKING WOULD BE AVAILABLE - ONLY FROM LAST REMAINING SYSTEM.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-30129  
NASA FMEA #: 02-1-033-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30129  
ITEM: HYD PRESS REG (SYS 1)

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ N / ]	[ ]	[ N ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ X ]

## REMARKS:

SEE 30111 - DIFFERENT BECAUSE THIS DOESN'T PASS REDUNDANCY  
SCREEN B.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-21301	BASELINE [ X ]
NASA FMEA #: 02-1-034-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             21301  
ITEM:                 MLG DOOR BOOSTER BUNGEE

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]
COMPARE	[   /   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

REMARKS:



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-30125  
NASA FMEA #: 02-1-044-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30125  
ITEM: RUDDER / BRAKE PEDAL ASSEMBLY

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ] *
IOA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-30126	BASELINE [ X ]
NASA FMEA #: 02-1-044-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             30126  
ITEM:                 RUDDER / BRAKE PEDAL ASSEMBLY

LEAD ANALYST:       W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ] *
IOA	[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ] (ADD/DELETE)
-----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-30123  
NASA FMEA #: 02-1-050-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30123  
ITEM: EXCITER RING - WHEEL SENSOR

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[ ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ X ]

## REMARKS:

EXCITER RING USED TO FEED WHEEL SPEED INFORMATION BACK TO CIRCUIT CONTROL BOX. IF INPUT NOT RECEIVED THEN ANTISKID FUNCTION FOR THAT CIRCUIT INOPERATIVE. CREW SHOULD GET FAILURE LIGHT AND AVOID HEAVY BRAKING.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-30123A	BASELINE [ X ]
NASA FMEA #: 02-1-051-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             30123  
ITEM:                EXCITER RING - WHEEL SENSOR

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[ . ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS:   (If different from NASA)

[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[ A ] (ADD/DELETE)
----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[ X ]

## REMARKS:

EXCITER RING USED TO FEED WHEEL SPEED INFORMATION BACK TO CIRCUIT CONTROL BOX. IF INPUT NOT RECEIVED THEN ANTISKID FUNCTION FOR THAT CIRCUIT INOPERATIVE. CREW SHOULD GET FAILURE LIGHT AND AVOID HEAVY BRAKING.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-30123B  
NASA FMEA #: 02-1-051-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30123  
ITEM: EXCITER RING - WHEEL SENSOR

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

CRITICALITY		REDUNDANCY SCREENS			CIL
FLIGHT					ITEM
HDW/FUNC		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[ ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ X ]

## REMARKS:

EXCITER RING USED TO FEED WHEEL SPEED INFORMATION BACK TO CIRCUIT CONTROL BOX. IF INPUT NOT RECEIVED THEN ANTISKID FUNCTION FOR THAT CIRCUIT INOPERATIVE. CREW SHOULD GET FAILURE LIGHT AND AVOID HEAVY BRAKING.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-30123C	BASELINE [ X ]
NASA FMEA #: 02-1-053-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             30123  
ITEM:                EXCITER RING - WHEEL SENSOR

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]
IOA	[ 3 /3 ]	[ NA]	[ NA]	[ NA]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS:   (If different from NASA)

[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[ A ] (ADD/DELETE)
----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[ X ]

## REMARKS:

EXCITER RING USED TO FEED WHEEL SPEED INFORMATION BACK TO  
CIRCUIT CONTROL BOX. IF INPUT NOT RECEIVED THEN ANTISKID  
FUNCTION FOR THAT CIRCUIT INOPERATIVE. CREW SHOULD GET FAILURE  
LIGHT AND AVOID HEAVY BRAKING.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-30124  
NASA FMEA #: 02-1-066-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30124  
ITEM: STATORS, ROTORS, CLIPS

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 1 /1 ]	[ P ]	[ P ]	[ P ]	[ X ]
COMPARE	[ N /N ]	[ ]	[ ]	[ ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ 1 /1 ]	[ P ]	[ P ]	[ P ]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IF LOCKUP OCCURS AT HIGH SPEED, TIRE WILL BLOW CAUSING  
POSSIBLE LOSS OF CREW AND VEHICLE.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-10215	BASELINE [ X ]
NASA FMEA #: 02-1-075-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             10215  
ITEM:                 SHOCK STRUT

LEAD ANALYST:        W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]
COMPARE	[   /   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

THE NGSSA CAN LOSE THE NITROGEN PRESSURE DOWN TO 1 ATMOSPHERE AND STILL PERFORM A SAFE LANDING, BUT ONCE A LOSS OF HYDRAULIC FLUID OCCURS THE SHOCK ATTENUATION CAPABILITY OF THE NGSSA IS DEGRADED BEYOND THE REQUIREMENTS FOR A SAFE LANDING.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-10216  
NASA FMEA #: 02-1-075-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10216  
ITEM: SHOCK STRUT

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 3 /1R ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ N /N ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /1R ]	[ NA]	[ NA]	[ NA]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ X ]

## REMARKS:

ALSO SEE IOA 10215.

THIS ASSESSMENT RELATES DIRECTLY TO THE LOSS OF THE NITROGEN  
ELASTIC MEDIUM.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-10211	BASELINE [ X ]
NASA FMEA #: 02-1-076-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             10211  
ITEM:                 TORQUE ARM ASSEMBLY

LEAD ANALYST:       W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[   ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS:   (If different from NASA)

[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ A ] (ADD/DELETE)
----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[ X ]

## REMARKS:

A NOSE WHEEL SLAPDOWN WHERE THE NOSE WHEEL ROTATES BEYOND A SAFE ANGLE OF ATTACK WILL CAUSE AN IMMEDIATE COLLAPSE OF THE NLG.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-10202  
NASA FMEA #: 02-1-077-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10202  
ITEM: DRAG BRACE

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

## RECOMMENDATIONS: (If different from NASA)

[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ A ]
				(ADD/DELETE)

## \* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ X ]

## REMARKS:

ALSO SEE 10203, 10202, 10221  
CONCUR WITH THE BASIC ANALYSIS BUT THE ANALYSIS IS NOT  
COMPREHENSIVE ENOUGH TO COVER SINGLE POINT FAILURES WITHIN THE  
COMPONENT.  
NASA FMEA TREATS THE DRAG BRACE AS A SINGLE COMPONENT WHEN  
THERE ARE SEVERAL PARTS THAT ARE CRITICAL INDIVIDUALLY.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE:	12/15/86	NASA DATA:
ASSESSMENT ID:	LDGDEC-10203	BASELINE [ X ]
NASA FMEA #:	02-1-077-1	NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10203  
ITEM: DRAG BRACE TRUNION

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ] *
IOA	[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

## RECOMMENDATIONS: (If different from NASA)

[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ A ]
				(ADD/DELETE)

## \* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ ]
INADEQUATE	[ X ]

## REMARKS:

THE DRAG BRACE TRUNIONS WERE NOT CONSIDERED AS AN INDIVIDUAL COMPONENT.  
SEE LDGDEC-10203

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-10221  
NASA FMEA #: 02-1-077-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10221  
ITEM: DRAG BRACE

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ] *
IOA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

ALSO SEE 10202, 10203

FMEA 02-1-077-1 COVERS THE NLG DRAG BRACE ASSEMBLY BUT IT  
DOES NOT COVER THE CRITICAL PARTS INDIVIDUALLY.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-10222  
NASA FMEA #: 02-1-077-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10222  
ITEM: DRAG BRACE TRUNION

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

## RECOMMENDATIONS: (If different from NASA)

[ 1 /1 ]	[ NA]	[ NA]	[ AN]	[ A ]
				(ADD/DELETE)

## \* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

ALSO SEE 10202, 10203

FMEA 02-1-077-1 COVERS THE NLG DRAG BRACE ASSEMBLY BUT IT DOES NOT COVER THE CRITICAL PARTS INDIVIDUALLY.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-10223  
NASA FMEA #: 02-1-077-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10223  
ITEM: DRAG BRACE TRUNION

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

## RECOMMENDATIONS: (If different from NASA)

[ 1 /1 ]	[ NA]	[ NA]	[ AN]	[ A ]
				(ADD/DELETE)

## \* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

ALSO SEE 10202, 10203

FMEA 02-1-077-1 COVERS THE NLG DRAG BRACE ASSEMBLY BUT IT DOES NOT COVER THE CRITICAL PARTS INDIVIDUALLY.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-10224	BASELINE [ X ]
NASA FMEA #: 02-1-077-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             10224  
ITEM:                 SUPPORT BEAM

LEAD ANALYST:       W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A           B           C		
NASA	[ 1 /1 ]	[ NA]   [ NA]   [ NA]		[ X ] *
IOA	[ 1 /1 ]	[ NA]   [ NA]   [ NA]		[ X ]
COMPARE	[   /   ]	[   ]   [   ]   [   ]		[   ]

## RECOMMENDATIONS:   (If different from NASA)

[ 1 /1 ]	[ NA]	[ NA]	[ AN]	[ A ] (ADD/DELETE)
----------	-------	-------	-------	-----------------------

## \* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

## REMARKS:

ALSO SEE 10202, 10203

FMEA 02-1-077-1 COVERS THE NLG DRAG BRACE ASSEMBLY BUT IT DOES NOT COVER THE CRITICAL PARTS INDIVIDUALLY.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-10204  
NASA FMEA #: 02-1-078-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10204  
ITEM: LOCK BRACE ASSEMBLY

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-10205	BASELINE [ X ]
NASA FMEA #: 02-1-079-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             10205  
ITEM:                 DOWNLOCK BUNGEE

LEAD ANALYST:       W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ NA ]	[ P ]	[   ] *
IOA	[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ]
COMPARE	[ N /N ]	[ N ]	[   ]	[ N ]	[ N ]

## RECOMMENDATIONS:   (If different from NASA)

[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ A ] (ADD/DELETE)
----------	--------	--------	--------	-----------------------

## \* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

ALSO SEE 10206

THE DOWNLOCK BUNGEE IS A MECHANICAL DEVICE THAT IF BENT OR JAMMED IN THE EXTENDED POSITION COULD CAUSE A FORCE THAT WOULD UNLOCK THE LOCK BRACE.

HYDRAULICS - THE EXTEND/RETRACT HYD ACTUATOR IS THE ONLY REDUNDANT ITEM. WHEN THE VEHICLE IS SHUT DOWN POST LANDING THERE IS NO REDUNDANCY. THE NASA FMEA/CIL DOES NOT CONSIDER APU SHUTDOWN OCCURRING BEFORE CREW EGRESS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-10206	BASELINE [ X ]
NASA FMEA #: 02-1-079-1	NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10206  
ITEM: DOWNLOCK BUNGEE

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ NA ]	[ P ]	[ ] *
IOA	[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ]
COMPARE	[ N /N ]	[ N ]	[ ]	[ N ]	[ N ]

## RECOMMENDATIONS: (If different from NASA)

[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ A ] (ADD/DELETE)
----------	--------	--------	--------	-----------------------

## \* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ ]
INADEQUATE	[ ]

## REMARKS:

SEE 10205.

THERE WILL BE NO COMPLICATIONS THROUGHOUT THE LANDING UNTIL AFTER THE VEHICLE IS SHUTDOWN. ONCE THE HYDRAULICS SYSTEM IS DEACTIVATED THERE IS NO SYSTEM TO HOLD THE LOCK BRACE IN POSITION, AND A GUST OF WIND, AN IMPACT FROM APPROACHING VEHICLES OR MOVEMENT INSIDE THE VEHICLE COULD CAUSE NLG COLLAPSE. A COLLAPSE OF THE NLG WOULD CAUSE STRUCTURAL DAMAGE AND A POSSIBLE LOSS OF LIFE.

THIS SITUATION CAN BE BYPASSED BY INSTALLING THE LANDING GEAR SAFETY PINS IN THE LOCK BRACE PRIOR TO HYDRAULICS SYSTEM 1 SHUTDOWN.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-10214	BASELINE [ X ]
NASA FMEA #: 02-1-080-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             10214  
ITEM:                WEIGHT ON WHEELS SENSORS - NLG

LEAD ANALYST:       W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ NA ]	[ P ]	[   ] *
IOA	[ 3 /3 ]	[ P ]	[ P ]	[ P ]	[   ]
COMPARE	[   /N ]	[   ]	[ N ]	[   ]	[   ]

## RECOMMENDATIONS:   (If different from NASA)

[ 3 /3 ]	[ P ]	[ P ]	[ P ]	[ A ] (ADD/DELETE)
----------	-------	-------	-------	-----------------------

## \* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

2 SWITCHES ARE ON THE NLG, EITHER OF WHICH WILL ACTIVATE NWS AND B&A'S. THE SWITCHES CAN ALSO BE BYPASSED VIA THE ET SEP SWITCH/SRB SEP SWITCH.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-10218  
NASA FMEA #: 02-1-080-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10218  
ITEM: WEIGHT ON WHEELS SENSORS - NLG

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ NA ]	[ P ]	[ ] *
IOA	[ 3 /3 ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ /N ]	[ ]	[ N ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ]	[ P ]	[ P ]	[ P ]	[ ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ X ]

## REMARKS:

- SEE 10214, 10219.
- SWITCHES CAN BE BYPASSED BY THE ET/SRB SEP SWITCH.
  - DIFFERENTIAL BRAKING CAN BE USED AS BACKUP STEERING.
  - NASA FMEA ASSUMES THE FAILURE OF BOTH SENSOR SWITCHES.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-10219	BASELINE [ X ]
NASA FMEA #: 02-1-080-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             10219  
ITEM:                WEIGHT ON WHEELS SENSORS - NLG

LEAD ANALYST:       W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS	CIL ITEM
		A           B           C	
NASA	[ 3 /1R ]	[ P ]   [ NA ]   [ P ]	[   ] *
IOA	[ 3 /3 ]	[ P ]   [ P ]   [ P ]	[   ]
COMPARE	[   /N ]	[   ]   [ N ]   [   ]	[   ]

## RECOMMENDATIONS:   (If different from NASA)

[ 3 /3 ]	[ P ]	[ P ]	[ P ]	[   ]
				(ADD/DELETE)

## \* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

SEE 10214.

- SENSOR CAN BE BYPASSED WITH THE ET/SRB SEP SWITCH.

- DIFFERENTIAL BRAKING CAN BE USED AS THE ALTERNATE STEERING METHOD.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-10207  
NASA FMEA #: 02-1-081-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10207  
ITEM: NLG - DOWN AND LOCK SENSOR

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ] *
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-10208	BASELINE [ X ]
NASA FMEA #: 02-1-081-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             10208  
ITEM:                NLG - DOWN AND LOCK SENSOR

LEAD ANALYST:       W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

MECHANICAL FAILURE CAUSING A LOSS OF SIGNAL.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-10217  
NASA FMEA #: 02-1-082-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10217  
ITEM: UPLOCK ROLLER RETAINING ASSEMBLY

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

NASA FMEA 02-1-083-1 COVERS THE SAME ASSY BUT THE FAILURE  
MODES ARE NOT ALIKE.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86 NASA DATA:  
ASSESSMENT ID: LDGDEC-10701 BASELINE [ X ]  
NASA FMEA #: 02-1-082-1 NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10701  
ITEM: NLG UPLOCK HOOK ASSEMBLY

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-10217A  
NASA FMEA #: 02-1-083-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10217  
ITEM: UPLOCK ROLLER RETAINING ASSEMBLY

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

NASA FMEA 02-1-082-1 COVERS THE SAME ASSY BUT THE FAILURE MODES ARE NOT ALIKE.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-10209	BASELINE [ X ]
NASA FMEA #: 02-1-085-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             10209  
ITEM:                STEERING COLLAR ASSEMBLY

LEAD ANALYST:       W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]
IOA	[ 3 /1R ]	[ NA ]	[ NA ]	[ NA ]
COMPARE	[   /   ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]	[   ]	[   ]	[   ]
							(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

## REMARKS:

THE ONLY PORTION OF THE FLIGHT THAT THIS FUNCTION CAN BE OBSERVED IS DURING THE LANDING ROLLOUT AND AT THAT POINT THE INFORMATION IS VIRTUALLY USELESS. IN ORDER TO RECEIVE A FAILURE SIGNAL THE NLG MUST HAVE WEIGHT ON THE NLG WHICH MEANS THAT THE VEHICLE IS IN THE ROLLOUT PHASE OF THE LANDING. IF A FAILURE OCCURS THE PILOT WOULD HAVE TAKEN OVER MANUALLY BEFORE HE WOULD RECOGNIZE THE FAILURE SIGNAL ON PANEL F3. THE VEHICLE WILL STOP BEFORE A COMPLETE ANALYSIS OF THE SITUATION IS PERFORMED.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-11102  
NASA FMEA #: 02-1-097-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 11102  
ITEM: NLG B/U PYRO UPLOCK RELEASE MECH

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ N /N ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ X ]

## REMARKS:

SYSTEM IS NOT USED UNLESS HYDRAULIC UPLOCK RELEASE SYSTEM FAILS. IF THIS SYSTEM FAILS WHEN CALLED ON TO FUNCTION, THERE IS NO OTHER BACKUP.

ACCORDING TO THE REDUNDANCY RULES IN 22206, THIS SYSTEM IS A 2/1R CRITICALITY BUT, A FAILED HYDRAULICS SYSTEM ACTIVATES THIS SYSTEM. THIS SYSTEMS FAILURE WILL NOT ACTIVATE THE HYDRAULICS. THERE IS A LINEAR OPERATION HERE THAT WILL NOT ALLOW REVERSAL OF THE ROLES.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-11101	BASELINE [ X ]
NASA FMEA #: 02-1-097-2	NEW [   ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 11101  
ITEM: NLG B/U PYRO UPLOCK RELEASE MECH

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ] (ADD/DELETE)
-----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-10501	BASELINE [ X ]
NASA FMEA #: 02-1-098-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             10501  
ITEM:                NLG DOOR EXTEND / RETRACT MECHANISM

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ] *
IOA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ] (ADD/DELETE)
-----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-10901	BASELINE [ X ]
NASA FMEA #: 02-1-098-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             10901  
ITEM:                NLG DOOR HOOK ACT LINKAGE

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ] (ADD/DELETE)
-----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

REMARKS:



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-10501A  
NASA FMEA #: 02-1-099-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10501  
ITEM: NLG DOOR EXTEND / RETRACT MECHANISM

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

DOORS COULD IMPACT AND DAMAGE ELECTRICAL AND HYDRAULIC  
COMPONENTS OR OTHER ITEMS LOCATED ON THE SHOCK STRUT ASSEMBLY

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-10901A	BASELINE [ X ]
NASA FMEA #: 02-1-099-1	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             10901  
ITEM:                NLG DOOR HOOK ACT LINKAGE

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ A ] (ADD/DELETE)
----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-11301  
NASA FMEA #: 02-1-102-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 11301  
ITEM: NLG DOOR BUNGEE ASSIST ASSY

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

CRITICALITY		REDUNDANCY SCREENS			CIL ITEM
FLIGHT HDW/FUNC		A	B	C	
NASA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ] *
IOA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-11202	BASELINE [ X ]
NASA FMEA #: 02-1-104-1	NEW [   ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 11202  
ITEM: NLG EXTENSION BOOSTER PYRO ACT

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-11201  
NASA FMEA #: 02-1-104-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 11201  
ITEM: NLG EXTENSION BOOSTER PYRO ACT

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
 ASSESSMENT ID: LDGDEC-10201  
 NASA FMEA #: 02-1-109-1

NASA DATA:  
 BASELINE [ X ]  
 NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
 MDAC ID: 10201  
 ITEM: NOSE LANDING GEAR TRUNION

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ] *
IOA	[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 1 /1 ] [ NA ] [ NA ] [ NA ] [ A ]  
 (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
 INADEQUATE [ X ]

## REMARKS:

ALSO SEE 10203, 10222, 10223

THE NASA FMEA DOES NOT COVER ANYTHING BUT THE NLG SHOCK STRUT TRUNIONS. THERE ARE SEVERAL OTHER TRUNNIONS IN THE NLG AND THEY ARE LOCATED IN THE NLG DRAGBRACE, THE NLG LOCKBRACE AND THE SUPPORT BEAM

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-10101  
NASA FMEA #: 02-1-110-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10101  
ITEM: TIRES, NLG TYPE II

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-21005	BASELINE [ X ]
NASA FMEA #: 02-6-G08-A01	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             21005  
ITEM:                 MLG UPLOCK ACTUATOR

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[   ] *
IOA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[   /   ]	[   ]	[ N ]	[   ]	[ N ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[ F ]	[   ]	[ A ] (ADD/DELETE)
-----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[ X ]

## REMARKS:

REDUNDANCY SCREEN B FAILS BECAUSE HYD. SYS. 1 FLUID IS NOT CIRCULATED TO THIS ACTUATOR ONORBIT, THUS FAILURE NOT DETECTED.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-21003  
NASA FMEA #: 02-6-G08-A02

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 21003  
ITEM: MLG UPLOCK ACTUATOR

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ N ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ]	[ ]	[ F ]	[ ]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ X ]

## REMARKS:

POSSIBLE LOSS OF HYDRAULICS SYSTEM 1. IF SYSTEM FAILS, THEN THE ORBITER IS ONE FAILURE AWAY FROM LOSS OF LIFE OR VEHICLE. PYRO BACKUP. HYDRAULIC FLUID IS NOT CIRCULATED TO THIS ACTUATOR ONORBIT, THUS FAILURE CANNOT BE DETECTED - FAILS REDUNDANCY SCREEN B.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-21006	BASELINE [ X ]
NASA FMEA #: 02-6-G08-A03	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             21006  
ITEM:                 MLG UPLOCK ACTUATOR

LEAD ANALYST:        J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20404  
NASA FMEA #: 02-6-G09-A01

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20404  
ITEM: MLG EXTEND / RETRACT HYD STRUT ACT

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ] *
IOA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ] (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

SAME FOR NOSE LANDING GEAR - SEE LDGDEC - 10404.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20402	BASELINE [ X ]
NASA FMEA #: 02-6-G09-A02	NEW [   ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20402  
ITEM: MLG EXTEND / RETRACT HYD STRUT ACT

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS	CIL ITEM
	A	B	C
NASA [ 3 /3 ]	[ NA ]	[ NA ]	[ NA ]
IOA [ 2 /1R ]	[ P ]	[ F ]	[ P ]
COMPARE [ N /N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

POSSIBLE LOSS OF HYDRAULICS SYSTEM 1. IF SYSTEM FAILS, THEN THE ORBITER IS ONE FAILURE AWAY FROM LOSS OF LIFE OR VEHICLE. THE GEAR HAS A PYRO BACKUP TO UNLOCK THE GEAR. IF IT FAILS, THE GEAR WILL NOT DEPLOY.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20403  
NASA FMEA #: 02-6-G09-A03

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20403  
ITEM: MLG EXTEND / RETRACT HYD STRUT ACT

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ] *
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20416	BASELINE [ X ]
NASA FMEA #: 02-6-G09-A04	NEW [   ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20416  
ITEM: MLG EXTEND / RETRACT HYD STRUT ACT

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]
IOA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]
COMPARE	[   /   ]	[   ]	[ N ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ A ] (ADD/DELETE)
-----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[ X ]

## REMARKS:

REDUNDANCY SCREEN B FAILS BECAUSE HYD SYS 1 FLUID IS NOT CIRCULATED TO THIS ACTUATOR ONORBIT, THUS, THE FAILURE IS NOT DETECTED.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20412  
NASA FMEA #: 02-6-G09-B01

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20412  
ITEM: MLG EXTEND / RETRACT HYD STRUT ACT

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ] *
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20413	BASELINE [ X ]
NASA FMEA #: 02-6-G09-B02	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             20413  
ITEM:                MLG EXTEND / RETRACT HYD STRUT ACT

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

REMARKS:



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20411	BASELINE [ X ]
NASA FMEA #: 02-6-G09-C01	NEW [   ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20411  
ITEM: MLG EXTEND / RETRACT HYD STRUT ACT

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS	CIL ITEM
	A	B	C
NASA [ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]
IOA [ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]
COMPARE [   /   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20409	BASELINE [ X ]
NASA FMEA #: 02-6-G09-D01	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             20409  
ITEM:                MLG EXTEND / RETRACT HYD STRUT ACT

LEAD ANALYST:       J. COMPTON

ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A           B           C		
NASA	[ 3 / 3 ]	[ NA ]   [ NA ]   [ NA ]	[   ]	*
IOA	[ 3 / 3 ]	[ NA ]   [ NA ]   [ NA ]	[   ]	
COMPARE	[   /   ]	[   ]   [   ]   [   ]	[   ]	

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86                      NASA DATA:  
 ASSESSMENT ID: LDGDEC-20410                      BASELINE [ X ]  
 NASA FMEA #: 02-6-G09-D02                      NEW [   ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
 MDAC ID: 20410  
 ITEM: MLG EXTEND / RETRACT HYD STRUT ACT

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[   ] *
IOA	[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[   ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]      [   ]      [   ]      [   ]      [   ]  
 (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [   ]  
 INADEQUATE [   ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20408	BASELINE [ X ]
NASA FMEA #: 02-6-G09-E01	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             20408  
ITEM:                MLG EXTEND / RETRACT HYD STRUT ACT

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[   ] *
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[   ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ] (ADD/DELETE)
-----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20408A  
NASA FMEA #: 02-6-G09-E02

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20408  
ITEM: MLG EXTEND / RETRACT HYD STRUT ACT

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ] *
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20414	BASELINE [ X ]
NASA FMEA #: 02-6-G09-F01	NEW [   ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20414  
ITEM: MLG EXTEND / RETRACT HYD STRUT ACT

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ] *
IOA	[ 1 / 1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86 NASA DATA:  
ASSESSMENT ID: LDGDEC-20406 BASELINE [ X ]  
NASA FMEA #: 02-6-G09-G01 NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20406  
ITEM: MLG EXTEND / RETRACT HYD STRUT ACT

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[ ] *
IOA	[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20415	BASELINE [ X ]
NASA FMEA #: 02-6-G09-H01	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             20415  
ITEM:                MLG EXTEND / RETRACT HYD STRUT ACT

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[   ] *
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[   ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ] (ADD/DELETE)
-----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

REMARKS:



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20415A  
NASA FMEA #: 02-6-G09-H02

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20415  
ITEM: MLG EXTEND / RETRACT HYD STRUT ACT

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ] *
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20407	BASELINE [ X ]
NASA FMEA #: 02-6-G09-J01	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             20407  
ITEM:                MLG EXTEND / RETRACT HYD STRUT ACT

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]	[   ]	
						(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20407A  
NASA FMEA #: 02-6-G09-J02

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20407  
ITEM: MLG EXTEND / RETRACT HYD STRUT ACT

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ] *
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ] (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20405	BASELINE [ X ]
NASA FMEA #: 02-6-G09-K01	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             20405  
ITEM:                MLG EXTEND / RETRACT HYD STRUT ACT

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

	ADEQUATE [   ]
	INADEQUATE [   ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-30128  
NASA FMEA #: 02-6-G12-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30128  
ITEM: BRAKE HYDRAULIC LINE HEATERS

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ] (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

CIL NOT AVAILABLE.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
 ASSESSMENT ID: LDGDEC-10404  
 NASA FMEA #: 02-6-H01-A01

NASA DATA:  
 BASELINE [ X ]  
 NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
 MDAC ID: 10404  
 ITEM: NLG EXTEND / RETRACT HYD STRUT ACTUATOR

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
 (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
 INADEQUATE [ ]

## REMARKS:

SAME FOR MAIN LANDING GEAR - SEE LDGDEC 20404.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-10402  
NASA FMEA #: 02-6-H01-A02

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10402  
ITEM: NLG EXTEND / RETRACT HYD STRUT ACT

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /3 ]	[ NA ]	[ NA ]	[ NA ]	[ ] *
IOA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ A ]
-----------	-------	-------	-------	-------

(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ X ]

## REMARKS:

POSSIBLE LOSS OF HYDRAULICS SYSTEM 1. IF THE SYSTEM FAILS, THEN THE ORBITER IS ONE FAILURE AWAY FROM LOSS OF LIFE OR VEHICLE. THE GEAR HAS A PYRO BACKUP TO UNLOCK THE GEAR. IF IT FAILS, THE GEAR WILL NOT DEPLOY.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE:	12/15/86	NASA DATA:
ASSESSMENT ID:	LDGDEC-10403	BASELINE [ X ]
NASA FMEA #:	02-6-H01-A03	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             10403  
ITEM:                NLG EXTEND / RETRACT HYD STRUT ACTUATOR

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[   ] *
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[   ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]	(ADD/DELETE)
-----------	-------	-------	-------	-------	--------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

REMARKS:



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-10416  
NASA FMEA #: 02-6-H01-A04

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10416  
ITEM: NLG EXTEND / RETRACT HYD STRUT ACTUATOR

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ N ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

REDUNDANCY SCREEN B BECAUSE HYD SYS 1 FLUID IS NOT  
CIRCULATED TO THIS ACTUATOR ONORBIT, THUS THE FAILURE IS NOT  
DETECTED.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-10412	BASELINE [ X ]
NASA FMEA #: 02-6-H01-B01	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             10412  
ITEM:                NLG EXTEND / RETRACT HYD STRUT ACTUATOR

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 3 /3 ]	[ NA]	[ NA]	[ NA]
IOA	[ 3 /3 ]	[ NA]	[ NA]	[ NA]
COMPARE	[   /   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86                      NASA DATA:  
 ASSESSMENT ID: LDGDEC-10413                      BASELINE [ X ]  
 NASA FMEA #: 02-6-H01-B02                      NEW [   ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
 MDAC ID: 10413  
 ITEM: NLG EXTEND / RETRACT HYD STRUT ACTUATOR

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ]      [ ]      [ ]      [ ]      [ ]  
 (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
 INADEQUATE [   ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-10411	BASELINE [ X ]
NASA FMEA #: 02-6-H01-C01	NEW [   ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10411  
ITEM: NLG EXTEND / RETRACT HYD STRUT ACTUATOR

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[   ] *
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[   ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ] (ADD/DELETE)
-----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-10409  
NASA FMEA #: 02-6-H01-D01

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10409  
ITEM: NLG EXTEND / RETRACT HYD STRUT ACTUATOR

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ] *
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ] (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-10410	BASELINE [ X ]
NASA FMEA #: 02-6-H01-D02	NEW [   ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10410  
ITEM: NLG EXTEND / RETRACT HYD STRUT ACTUATOR

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[   ] *
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[   ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ] (ADD/DELETE)
-----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

	ADEQUATE [   ]
REMARKS:	INADEQUATE [   ]

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86                      NASA DATA:  
 ASSESSMENT ID: LDGDEC-10408                      BASELINE [ X ]  
 NASA FMEA #: 02-6-H01-E01                      NEW [   ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
 MDAC ID: 10408  
 ITEM: NLG EXTEND / RETRACT HYD STRUT ACTUATOR

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[   ] *
IOA	[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[   ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]      [   ]      [   ]      [   ]      [   ]  
 (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [   ]  
 INADEQUATE [   ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-10408A	BASELINE [ X ]
NASA FMEA #: 02-6-H01-E02	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             10408  
ITEM:                NLG EXTEND / RETRACT HYD STRUT ACTUATOR

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[   ] *
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[   ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

REMARKS:



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86 NASA DATA:  
ASSESSMENT ID: LDGDEC-10414 BASELINE [ X ]  
NASA FMEA #: 02-6-H01-F01 NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10414  
ITEM: NLG EXTEND / RETRACT HYD STRUT ACTUATOR

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ] (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-10406	BASELINE [ X ]
NASA FMEA #: 02-6-H01-G01	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             10406  
ITEM:                NLG EXTEND / RETRACT HYD STRUT ACTUATOR

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]	[   ]	[   ]
(ADD/DELETE)						

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-10415  
NASA FMEA #: 02-6-H01-H01

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10415  
ITEM: NLG EXTEND / RETRACT HYD STRUT ACTUATOR

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ] *
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-10415A  
NASA FMEA #: 02-6-H01-H02

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10415  
ITEM: NLG EXTEND / RETRACT HYD STRUT ACTUATOR

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ] *
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-10407  
NASA FMEA #: 02-6-H01-J01

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10407  
ITEM: NLG EXTEND / RETRACT HYD STRUT ACTUATOR

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

CRITICALITY		REDUNDANCY SCREENS			CIL
FLIGHT					ITEM
HDW/FUNC		A	B	C	
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ] *
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-10407A	BASELINE [ X ]
NASA FMEA #: 02-6-H01-J02	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             10407  
ITEM:                 NLG EXTEND / RETRACT HYD STRUT ACTUATOR

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A           B           C		
NASA	[ 3 / 3 ]	[ NA ]   [ NA ]   [ NA ]		[   ] *
IOA	[ 3 / 3 ]	[ NA ]   [ NA ]   [ NA ]		[   ]
COMPARE	[   /   ]	[   ]   [   ]   [   ]		[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-10405	BASELINE [ X ]
NASA FMEA #: 02-6-H01-K01	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             10405  
ITEM:                NLG EXTEND / RETRACT HYD STRUT ACTUATOR

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A           B           C		
NASA	[ 3 / 3 ]	[ NA ]   [ NA ]   [ NA ]		[   ] *
IOA	[ 3 / 3 ]	[ NA ]   [ NA ]   [ NA ]		[   ]
COMPARE	[   /   ]	[   ]   [   ]   [   ]		[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]	(ADD/DELETE)
-----------	-------	-------	-------	-------	--------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-11005  
NASA FMEA #: 02-6-H03-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 11005  
ITEM: NLG UPLOCK ACTUATOR

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ N ]	[ ]	[ ]

## RECOMMENDATIONS: (If different from NASA)

[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ A ]
				(ADD/DELETE)

## \* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

REDUNDANCY SCREEN B FAILS BECAUSE HYD SYS. 1 FLUID IS NOT CIRCULATED TO THIS ACTUATOR ONORBIT, THUS FAILURE NOT DETECTED.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-11003  
NASA FMEA #: 02-6-H03-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 11003  
ITEM: NLG UPLOCK ACTUATOR

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ NA ]	[ NA ]	[ NA ]	[ ] *
IOA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ / ]	[ N ]	[ N ]	[ N ]	[ N ]

## RECOMMENDATIONS: (If different from NASA)

[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ A ]
				(ADD/DELETE)

## \* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

POSSIBLE LOSS OF HYDRAULICS SYSTEM 1. IF SYSTEM FAILS, THEN THE ORBITER IS ONE FAILURE AWAY FROM LOSS OF LIFE OR VEHICLE. PYRO BACKUP. HYDRAULIC FLUID IS NOT CIRCULATED TO THIS ACTUATOR ONORBIT, THUS FAILURE CANNOT BE DETECTED - FAILS REDUNDANCY SCREEN B.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-11006	BASELINE [ X ]
NASA FMEA #: 02-6-H03-3	NEW [   ]

SUBSYSTEM:           LANDING/DECELERATION SYSTEMS  
MDAC ID:             11006  
ITEM:                 NLG UPLOCK ACTUATOR

LEAD ANALYST:       J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS:   (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31147  
NASA FMEA #: 05-06BA-2401-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31147  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] .
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ] (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF FMEA 2401-1.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31147A	BASELINE [ X ]
NASA FMEA #: 05-06BA-2401-2	NEW [   ]

SUBSYSTEM: EPD&C  
MDAC ID: 31147  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[   ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[   ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ] (ADD/DELETE)
-----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF FMEA 2401-2, AND IOA RECOMMENDS THAT FMEA 2401-2 SHOULD COMBINED WITH 2401-1. NASA REEVALUATION COMBINED FMEA 2401-2 IN WITH 2401-1.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31224  
NASA FMEA #: 05-2106-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31224  
ITEM: TOGGLE SWITCH, 3PST

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ P ] [ P ] [ P ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE "MAIN C" TOGGLE SWITCH.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31155	BASELINE [ X ]
NASA FMEA #: 05-2409-2	NEW [   ]

SUBSYSTEM: EPD&C  
MDAC ID: 31155  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 3)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS	CIL ITEM
	A	B	C
NASA [ 3 / 3 ]	[   ]	[   ]	[   ]
IOA [ 3 / 1R ]	[ P ]	[ F ]	[ P ]
COMPARE [   / N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE TYPE 3 HDC'S.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31163  
NASA FMEA #: 05-6BA-200200-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31163  
ITEM: HYBRID DRIVER CONTROLLER (TYPE II)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ N ]	[ ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ] (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION.  
NASA GENERATED SEPARATE FMEA'S TO COVER THE HDC'S FAILURE MODES  
INSTEAD OF LUMPING THEM TOGETHER WITH UNLIKE COMPONENTS IN FMEA  
200200-1. REFER TO LDGDEC-31164.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31164  
NASA FMEA #: 05-6BA-200200-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31164  
ITEM: HYBRID DRIVER CONTROLLER (TYPE II)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ N ]	[ ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

NASA GENERATED SEPARATE FMEA'S TO COVER THE FAILURE MODES INSTEAD OF LUMPING THEM TOGETHER WITH UNLIKE COMPONENTS IN FMEA 200200-1. FAILURE MODES EVALUATED IN FMEAS 05-6BA-2209-1, 2209-2, 2356-1, 2356-2, 2408-1, 2408-2, 2409-1, 2409-2, 2410-1, 2410-2, 2575-1, AND 2575-2.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31165  
NASA FMEA #: 05-6BA-200200-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31165  
ITEM: HYBRID DRIVER CONTROLLER (TYPE III)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ N ]	[ ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ] (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

NASA INCORPORATED FMEA INTO OTHER FMEA. SEE ASSESSMENT LDGDEC-31164.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31166  
NASA FMEA #: 05-6BA-200200-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31166  
ITEM: HYBRID DRIVER CONTROLLER (TYPE III)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ N ]	[ ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ F ] [ ] [ A ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

NASA INCORPORATED FMEA INTO OTHER FMEAS. SEE ASSESSMENT LDGDEC-31164.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31181  
NASA FMEA #: 05-6BA-200200-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31181  
ITEM: MAIN GEAR BRAKE UPLOCK RELEASE CIRCUITS 1 & 2

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA RECOMMENDS: (1) DELETING THIS FMEA AND COVERING EACH TYPE OF COMPONENT WITH ITS OWN FMEA (E.G. MDAC ID'S 31120, 31128, 31129, 31156, 31157, 31163 THROUGH 31166 COVER ALL THE COMPONENTS LISTED IN THIS FMEA) INTO OTHER FMEAS, REFER TO MDAC ASSESSMENTS NOTED ABOVE.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31119  
NASA FMEA #: 05-6BA-2113-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31119  
ITEM: EVENT INDICATORS (6)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE LANDING GEAR EVENT INDICATORS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31113  
NASA FMEA #: 05-6BA-2115-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31113  
ITEM: PUSHBUTTON SWITCH (2), LANDING GEAR DOWN

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ N / ]	[ ]	[ N ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

PROVIDES REDUNDANT MANUAL "ON" CONTROL FROM CONTROL BUS TO LATCHING RELAYS FOR LANDING GEAR DOWN CIRCUIT. AFTER FURTHER REVIEW/ANALYSIS IOA DOES CONCUR FULLY WITH NASA'S EVALUATION.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31114A	BASELINE [ X ]
NASA FMEA #: 05-6BA-2115-2	NEW [   ]

SUBSYSTEM: EPD&C  
MDAC ID: 31114  
ITEM: PUSHBUTTON SWITCH (2), LANDING GEAR DOWN

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A            B            C		
NASA	[   /   ]	[   ]   [   ]   [   ]		[   ]   *
IOA	[ 3 / 3 ]	[   ]   [   ]   [   ]		[   ]
COMPARE	[ N / N ]	[   ]   [   ]   [   ]		[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

PROVIDES REDUNDANT MANUAL "ON" CONTROL FROM CONTROL BUS TO LATCHING RELAYS FOR LANDING GEAR DOWN CIRCUIT. IOA DOES CONCUR WITH NASA'S EVALUATION THAT FMEA 2115-2 (SHORTS TO GROUND) IS NOT A CREDIBLE FAILURE AND THAT IT BE DELETED.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31114  
NASA FMEA #: 05-6BA-2115-3

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31114  
ITEM: PUSHBUTTON SWITCH (2), LANDING GEAR DOWN

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

CRITICALITY		REDUNDANCY SCREENS			CIL
FLIGHT					ITEM
HDW/FUNC		A	B	C	
NASA	[ 1 /1 ]	[ ]	[ ]	[ ]	[ X ] *
IOA	[ 3 /3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ N /N ]	[ ]	[ ]	[ ]	[ N ]

## RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ]	[ ]	[ ]	[ ]	[ D ]
				(ADD/DELETE)

## \* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

PROVIDES REDUNDANT MANUAL "ON" CONTROL FROM CONTROL BUS TO LATCHING RELAYS FOR LANDING GEAR DOWN CIRCUIT. IOA DOES NOT CONCUR WITH NASA'S EVALUATION AND IOA RECOMMENDS DOWNGRADING FMEA 2115-3 CRITICALITY TO 3/3.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31115  
NASA FMEA #: 05-6BA-2116-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31115  
ITEM: LANDING GEAR TOGGLE SWITCH, S13

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCUR WITH NASA'S EVALUATION, FAILURE HAS NO EFFECT ON  
SUBSYSTEM. CB60 REMAINS "OFF" UNTIL NEEDED.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31115A  
NASA FMEA #: 05-6BA-2116-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31115  
ITEM: LANDING GEAR TOGGLE SWITCH, S13

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ] (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA RECOMMENDS THAT THE FAILURE MODE OF FMEA 2116-2 BE DELETED, BECAUSE IT IS NOT A CREDIBLE FAILURE FOR THIS COMPONENT. NASA EVALUATION DELETED NASA 2116-2.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31115B  
NASA FMEA #: 05-6BA-2116-3

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31115  
ITEM: LANDING GEAR TOGGLE SWITCH, S13

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ X ] *
IOA	[ 3 /3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ] [ ] [ ] [ ] [ D ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES NOT CONCUR WITH NASA'S EVALUATION, FAILURE HAS NO EFFECT ON SUBSYSTEM. CB60 REMAINS "OFF" UNTIL NEEDED. IOA RECOMMENDS DOWNGRADING THE CRITICALITY AND COMBINING FMEA'S 2116-1 AND 2116-3 TOGETHER, SINCE THEY ARE BOTH CRITICALITY 3/3, TO CONFORM TO NSTS 22206.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31117  
NASA FMEA #: 05-6BA-2117-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31117  
ITEM: PUSHBUTTON SWITCH, LDG GR ARM, 4PDT, ILLUMINATED

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ N / ]	[ ]	[ N ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /1R ] [ P ] [ F ] [ P ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

PROVIDES REDUNDANT MANUAL "ON" CONTROL FROM CONTROL BUS TO LATCHING RELAYS FOR LANDING GEAR ARM CIRCUIT. IOA DOES NOT CONCUR FULLY WITH NASA'S EVALUATION AND IOA RECOMMENDS: (1) CHANGING THE REDUNDANCY SCREENS, IT FAILS REDUNDANCY SCREEN B, AND LOWERING THE CRITICALITY TO 3/1R.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31118A	BASELINE [ X ]
NASA FMEA #: 05-6BA-2117-2	NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31118  
ITEM: PUSHBUTTON SWITCH, LDG GR ARM, 4 PDT,  
ILLUMINATED

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ NA ]	[ NA ]	[ NA ]	[ ] *
IOA	[ 3 /3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ /N ]	[ N ]	[ N ]	[ N ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ]	[ ]	[ ]	[ ]	[ ]	[ ]	[ ]	[ ]
(ADD/DELETE)							

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ ]
INADEQUATE	[ ]

## REMARKS:

PROVIDES REDUNDANT MANUAL "ON" CONTROL FROM CONTROL BUS TO LATCHING RELAYS FOR LANDING GEAR ARM CIRCUIT. NASA FMEA 2117-2 (SHORTS TO GROUND) IS NOT A CREDIBLE FAILURE AND IOA RECOMMENDS THAT IT BE DELETED FROM THE CIL.  
NASA DELETED THIS FMEA FROM THE BASELINE

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31118  
NASA FMEA #: 05-6BA-2117-3

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31118  
ITEM: PUSHBUTTON SWITCH, LDG GR ARM, 4 PDT,  
ILLUMINATED

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 3 /3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ N ]

## RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ]	[ ]	[ ]	[ ]	[ D ]
----------	-----	-----	-----	-------

(ADD/DELETE)

## \* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

PROVIDES REDUNDANT MANUAL "ON" CONTROL FROM CONTROL BUS TO LATCHING RELAYS FOR LANDING GEAR ARM CIRCUIT. IOA DOES NOT CONCUR WITH NASA'S EVALUATION AND IOA RECOMMENS DOWNGRADING FMEA 2117-3 CRITICALITY TO 3/3, AND DELETING IT FROM THE CIL.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31101  
NASA FMEA #: 05-6BA-2118-4

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31101  
ITEM: PROXIMITY SENSOR BOX (2)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE PROXIMITY SENSOR BOXES  
FOR THE FAILURE MODE: INADVERTENT OUTPUT.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31100  
NASA FMEA #: 05-6BA-2204-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31100  
ITEM: ISOLATION DIODE (12), 1 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE ISOLATION DIODES.  
LOSS OF EVENT INDICATION NOT CRITICAL TO FLIGHT.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31100A  
NASA FMEA #: 05-6BA-2204-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31100  
ITEM: ISOLATION DIODE (12), 1 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE ISOLATION DIODES.  
LOSS OF EVENT INDICATION NOT CRITICAL TO FLIGHT.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31105  
NASA FMEA #: 05-6BA-2205-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31105  
ITEM: TRANSIENT SUPPRESSOR DIODE (4), 3 AMPS

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF FMEA 2205-1 FOR THE  
TRANSIENT SUPPRESSOR DIODES.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
 ASSESSMENT ID: LDGDEC-31105A  
 NASA FMEA #: 05-6BA-2205-2

NASA DATA:  
 BASELINE [ X ]  
 NEW [ ]

SUBSYSTEM: EPD&C  
 MDAC ID: 31105  
 ITEM: TRANSIENT SUPPRESSOR DIODE (4), 3 AMPS

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ F ]	[ F ]	[ P ]	[ X ] *
IOA	[ 3 /3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ] [ ] [ ] [ ] [ D ]  
 (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
 INADEQUATE [ ]

## REMARKS:

IOA DOES NOT CONCUR WITH NASA'S EVALUATION AND IOA RECOMMENDS  
 DOWNGRADING THE CRITICALITY AND REMOVING THIS ITEM FROM CIL.  
 LOSS OF TWO DIODES IS LOSS OF A HYDRAULIC SYSTEM WHICH THEN  
 RESULTS IN 3/1R CRITICALITY.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31107  
NASA FMEA #: 05-6BA-2206-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31107  
ITEM: BLOCKING DIODE (2) 12 AMP, 400V

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

CRITICALITY FLIGHT HDW/FUNC		REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ] (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION AND RATIONALE, LOSS OF ELECTRICAL POWER MIGHT LEAD TO LOSS OF CREW/VEHICLE.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31108  
NASA FMEA #: 05-6BA-2206-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31108  
ITEM: BLOCKING DIODE (2) 12 AMP, 400V

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE BLOCKING DIODES. IF A  
BLOCKING DIODE FAILS SHORTED, POSSIBLE LOSS OF RETURN BUS  
ISOLATION MIGHT OCCUR.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31109  
NASA FMEA #: 05-6BA-2207-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31109  
ITEM: BLOCKING DIODE (2) 12 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION AND RATIONALE, LOSS OF ELECTRICAL POWER FROM AN RPC MIGHT CAUSE LOSS OF CREW/VEHICLE.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31110  
NASA FMEA #: 05-6BA-2207-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31110  
ITEM: BLOCKING DIODE (2) 12 AMP, 400V

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION, POSSIBLE LOSS OF UNIT  
CAPABILITY TO ISOLATE THE TWO RPC OUTPUTS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31122  
NASA FMEA #: 05-6BA-2208-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31122  
ITEM: BLOCKING DIODE (2) 1 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE BLOCKING DIODES.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31122A  
NASA FMEA #: 05-6BA-2208-2

NASA DATA:  
BASELINE [ ]  
NEW [ X ]

SUBSYSTEM: EPD&C  
MDAC ID: 31122  
ITEM: BLOCKING DIODE (2) 1 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE BLOCKING DIODES.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31121  
NASA FMEA #: 05-6BA-2209-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31121  
ITEM: BLOCKING DIODE (6) 3 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

CRITICALITY FLIGHT HDW/FUNC		REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

## RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ] (ADD/DELETE)

## \* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31120  
NASA FMEA #: 05-6BA-2209-2

NASA DATA:  
BASELINE [ ]  
NEW [ X ]

SUBSYSTEM: EPD&C  
MDAC ID: 31120  
ITEM: BLOCKING DIODE (6) 3 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH THE NASA FMEA EVALUATION.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31111  
NASA FMEA #: 05-6BA-2243-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31111  
ITEM: CIRCUIT BREAKERS (2), LG SENSORS

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

REMARKS:  
IOA DOES CONCUR WITH NASA'S EVALUATION.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31112  
NASA FMEA #: 05-6BA-2243-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31112  
ITEM: CIRCUIT BREAKERS (2), LG SENSORS

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION, FAILURE HAS NO EFFECT ON SUBSYSTEM, CIRCUIT BREAKERS ARE NORMALLY IN A CLOSED POSITION.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31123  
NASA FMEA #: 05-6BA-2244-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31123  
ITEM: CIRCUIT BREAKER (3 AMP)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE LDG ARM/DN RESET  
CIRCUIT BREAKER.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE:	1/23/87	NASA DATA:
ASSESSMENT ID:	LDGDEC-31123A	BASELINE [ X ]
NASA FMEA #:	05-6BA-2244-2	NEW [   ]

SUBSYSTEM: EPD&C  
MDAC ID: 31123  
ITEM: CIRCUIT BREAKER (3 AMP)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 3 /3 ]	[   ]	[   ]	[   ]	[   ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE LDG ARM/DN RESET CIRCUIT BREAKER.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31126  
NASA FMEA #: 05-6BA-2300-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31126  
ITEM: GENERAL PURPOSE FUSE, (8), 1 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

REMARKS:  
IOA DOES CONCUR WITH NASA'S EVALUATION.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31126A	BASELINE [ X ]
NASA FMEA #: 05-6BA-2300-2	NEW [   ]

SUBSYSTEM: EPD&C  
MDAC ID: 31126  
ITEM: GENERAL PURPOSE FUSE, (8), 1 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[   ]	[   ]	[   ]	[   ] *
IOA	[ 3 / 1R ]	[ P ]	[ F ]	[ P ]	[   ]
COMPARE	[   / N ]	[ N ]	[ N ]	[ N ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION AND DELETING FMEA 2300-2  
BECAUSE IT IS NOT A CREDIBLE FAILURE.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31127  
NASA FMEA #: 05-6BA-2301-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31127  
ITEM: GENERAL PURPOSE FUSE (1 AMP)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31127A	BASELINE [ X ]
NASA FMEA #: 05-6BA-2301-2	NEW [   ]

SUBSYSTEM: EPD&C  
MDAC ID: 31127  
ITEM: GENERAL PURPOSE FUSE (1 AMP)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 3 /3 ]	[   ]	[   ]	[   ]	[   ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[   ]
COMPARE	[   /N ]	[ N ]	[ N ]	[ N ]	[   ]

## RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ] (ADD/DELETE)
-----------	-------	-------	-------	-----------------------

## \* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [   ]  
INADEQUATE [   ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION THAT FMEA 2301-2 BE DELETED BECAUSE ITS FAILURE MODE IS NOT CREDIBLE FOR THIS COMPONENT.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31125  
NASA FMEA #: 05-6BA-2302-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31125  
ITEM: GENERAL PURPOSE FUSE (5 AMP)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ N / ]	[ ]	[ N ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ X ]

## REMARKS:

IOA DOES NOT CONCUR FULLY WITH NASA'S EVALUATION AND IOA  
RECOMMENDS: CHANGING THE REDUNDANCY SCREENS SINCE IT FAILS  
REDUNDANCY SCREEN B, AND DOWNGRADING THE CRITICALITY TO 3/1R.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31137  
NASA FMEA #: 05-6BA-2303-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31137  
ITEM: GENERAL PURPOSE FUSE (2), 5 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ N ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ F ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ X ]

## REMARKS:

IOA DOES NOT CONCUR FULLY WITH NASA'S EVALUATION OF FMEA 2303-1  
AND IOA RECOMMENDS CHANGING THE REDUNDANCY SCREEN B TO CONFORM TO  
NSTS 22206.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31137A  
NASA FMEA #: 05-6BA-2303-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31137  
ITEM: GENERAL PURPOSE FUSE (2), 5 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 2 / 1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ N / N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ ] / [ ] [ ] [ ] [ ] [ ] (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF FMEA 2303-2 THAT THE FMEA AND CIL REFERENCE BE DELETED BECAUSE IT IS NOT A CREDIBLE FAILURE MODE FOR A FUSE.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31130	BASELINE [ X ]
NASA FMEA #: 05-6BA-2351-1	NEW [   ]

SUBSYSTEM: EPD&C  
MDAC ID: 31130  
ITEM: ISOLATION RESISTORS (18); 5.1K, 1/4W

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[   ]	[   ]	[   ]	[   ] *
IOA	[ 3 / 3 ]	[   ]	[   ]	[   ]	[   ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE ISOLATION RESISTORS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31130A  
NASA FMEA #: 05-6BA-2351-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31130  
ITEM: ISOLATION RESISTORS (18); 5.1K, 1/4W

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ] (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR FULLY WITH NASA'S EVALUATION OF THE ISOLATION RESISTORS, RECOMMENDS THAT FMEA 2351-2 BE DELETED BECAUSE FAIL MODE IS NOT CREDIBLE FOR THIS TYPE OF RESISTOR.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31132  
NASA FMEA #: 05-6BA-2352-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31132  
ITEM: RESISTOR (3), 1.2K, 2W

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE CURRENT LIMITING RESISTOR.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31136  
NASA FMEA #: 05-6BA-2353-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31136  
ITEM: RESISTOR (3), 1.2K, 2W

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE CURRENT LIMITING  
RESISTORS IN THE "LDG GR DOWN" STATUS/MONITORING CIRCUITS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31134	BASELINE [ X ]
NASA FMEA #: 05-6BA-2354-1	NEW [   ]

SUBSYSTEM: EPD&C  
MDAC ID: 31134  
ITEM: RESISTOR (2), 1.8K, 1/4W

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A                  B                  C		
NASA	[ 3 / 3 ]	[   ]                  [   ]                  [   ]		[   ] *
IOA	[ 3 / 3 ]	[   ]                  [   ]                  [   ]		[   ]
COMPARE	[   /   ]	[   ]                  [   ]                  [   ]		[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ] (ADD/DELETE)
-----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE BLEED-OFF RESISTOR IN A  
RPC/MDM MONITORING CIRCUIT.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31128  
NASA FMEA #: 05-6BA-2356-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31128  
ITEM: RESISTOR (12), 10.2 OHMS, 2W

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ] (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31129	BASELINE [ X ]
NASA FMEA #: 05-6BA-2356-2	NEW [   ]

SUBSYSTEM: EPD&C  
MDAC ID: 31129  
ITEM: RESISTOR (12), 10.2 OHMS, 2W

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A            B            C		
NASA	[ 3 / 3 ]	[   ]   [   ]   [   ]		[   ] *
IOA	[ 3 / 3 ]	[   ]   [   ]   [   ]		[   ]
COMPARE	[   /   ]	[   ]   [   ]   [   ]		[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ] (ADD/DELETE)
-----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE CURRENT LIMITING RESISTORS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31138  
NASA FMEA #: 05-6BA-2357-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31138  
ITEM: RESISTOR (2), 2.2K, 1/2W

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ . ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE ISOLATION/CURRENT  
LIMITING RESISTORS USED IN RPC OUTPUT/MDM MONITORING CIRCUITS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31133	BASELINE [ X ]
NASA FMEA #: 05-6BA-2360-1	NEW [   ]

SUBSYSTEM: EPD&C  
MDAC ID: 31133  
ITEM: RESISTOR (6), 100K, 1/8W

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY			REDUNDANCY SCREENS			CIL ITEM
	FLIGHT			A	B	C	
	HDW/FUNC						
NASA	[ 3 / 3 ]			[   ]	[   ]	[   ]	[   ] *
IOA	[ 3 / 3 ]			[   ]	[ . ]	[   ]	[   ]
COMPARE	[   /   ]			[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [   ]  
INADEQUATE [   ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE CURRENT LIMITING RESISTOR.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31135  
NASA FMEA #: 05-6BA-2361-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31135  
ITEM: RESISTOR (8), 17.4K, 1/4W

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ] (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

REMARKS:  
IOA CONCURS WITH NASA'S EVALUATION OF THE GSE TEST CURRENT  
LIMITING RESISTORS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31135A  
NASA FMEA #: 05-6BA-2361-2

NASA DATA:  
BASELINE [    ]  
NEW [ X ]

SUBSYSTEM: EPD&C  
MDAC ID: 31135  
ITEM: RESISTOR (8), 17.4K, 1/4W

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[    ]	[    ]	[    ]	[    ] *
IOA	[ 3 / 3 ]	[    ]	[    ]	[    ]	[    ]
COMPARE	[    /    ]	[    ]	[    ]	[    ]	[    ]

RECOMMENDATIONS: (If different from NASA)

[    /    ] [    ] [    ] [    ] [    ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [    ]  
INADEQUATE [    ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE GSE TEST CURRENT  
LIMITING RESISTORS.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31139  
NASA FMEA #: 05-6BA-2362-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31139  
ITEM: RESISTOR, 1.2K, 2W

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE CURRENT LIMITING  
RESISTOR FOR THE LANDING STRUT ACTUATOR POSITION INDICATOR.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31139A  
NASA FMEA #: 05-6BA-2362-2

NASA DATA:  
BASELINE [    ]  
NEW [ X ]

SUBSYSTEM: EPD&C  
MDAC ID: 31139  
ITEM: RESISTOR, 1.2K, 2W

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[    ]	[    ]	[    ]	[    ] *
IOA	[ 3 / 3 ]	[    ]	[    ]	[    ]	[    ]
COMPARE	[    /    ]	[    ]	[    ]	[    ]	[    ]

RECOMMENDATIONS: (If different from NASA)

[    /    ] [    ] [    ] [    ] [    ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [    ]  
INADEQUATE [    ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE CURRENT LIMITING  
RESISTOR FOR THE LANDING STRUT ACTUATOR POSITION INDICATOR.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31131  
NASA FMEA #: 05-6BA-2363-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31131  
ITEM: RESISTOR, 7.5K, 2W

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE CURRENT LIMITING RESISTOR.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31140  
NASA FMEA #: 05-6BA-2400-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31140  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE TYPE 1 HDC (FAILURE MODE: LOSS OF OUTPUT) THAT CONNECTS MAIN DC BUS POWER TO THE WOW CIRCUITS. FMEA DELETED FROM LDG/DECL TRANSFERRED TO NOSE WHEEL STEERING.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31141  
NASA FMEA #: 05-6BA-2400-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31141  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

## RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

## \* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE TYPE 1 HDC (FAILURE MODE: INADVERTENT OUTPUT) THAT CONNECTS MAIN DC BUS POWER TO THE WOW CIRCUITS. FMEA TRANSFERRED TO NOSE WHEEL STEERING SUBSYSTEM.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31149  
NASA FMEA #: 05-6BA-2401-3

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31149  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE TYPE 1 HDC FOR THE  
FAILURE MODE: INADVERTENT OUTPUT.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31143  
NASA FMEA #: 05-6BA-2402-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31143  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF FMEA 2402-1.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31143A	BASELINE [ X ]
NASA FMEA #: 05-6BA-2402-2	NEW [   ]

SUBSYSTEM: EPD&C  
MDAC ID: 31143  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[   ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[   ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]	[   ]	(ADD/DELETE)
-----------	-------	-------	-------	-------	-------	--------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF FMEA 2402-2. NASA INCORPORATED FMEA 2402-2 INTO 2401-1.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31144  
NASA FMEA #: 05-6BA-2402-3

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31144  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ] (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE TYPE 1 HDC FOR THE  
FAILURE MODE: INADVERTENT OUTPUT. NASA INCORPORATED FMEA/CIL  
2402-3 INTO 2401-3.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31148  
NASA FMEA #: 05-6BA-2403-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31148  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF FMEA 2403-1 AND 2403-2, AND IOA RECOMMENDS NASA FMEA 2403-1 AND 2403-2 SHOULD BE COMBINED INTO ONE FMEA TO CONFORM TO NSTS 22206. NASA INCORPORATED 2403-2 INTO 2403-1.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31148A  
NASA FMEA #: 05-6BA-2403-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31148  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF FMEA 2403-1 AND 2403-2, AND IOA RECOMMENDS NASA FMEA 2403-1 AND 2403-2 SHOULD BE COMBINED INTO ONE FMEA TO CONFORM TO NSTS 22206. NASA INCORPORATED 2403-2 INTO 2403-1.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31142  
NASA FMEA #: 05-6BA-2404-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31142  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY		REDUNDANCY SCREENS			CIL ITEM
	FLIGHT	HDW/FUNC	A	B	C	
NASA	[ 3 / 3 ]		[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]		[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]		[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF FMEA'S 2404-1 AND 2404-2 FOR THE TYPE 1 HDC (FAILURE MODES: LOSS OF OUTPUT, INADVERTENT OUTPUT) TO POWER THE DOWN COILS OF THE TWO NLG EVENT INDICATORS. IOA RECOMMENDS COMBINING FMEA 2404-1 AND 2404-2 TOGETHER TO CONFORM TO NSTS 22206. NASA INCORPORATED 2404-1 AND 2404-2 INTO 2403-1.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31142A  
NASA FMEA #: 05-6BA-2404-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31142  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF FMEA'S 2404-1 AND 2404-2 FOR THE TYPE 1 HDC (FAILURE MODES: LOSS OF OUTPUT, INADVERTENT OUTPUT) TO POWER THE DOWN COILS OF THE TWO NLG EVENT INDICATORS. IOA RECOMMENDS COMBINING FMEA 2404-1 AND 2404-2 TOGETHER TO CONFORM TO NSTS 22206. NASA INCORPORATED 2404-1 AND 2404-2 INTO 2403-1.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31142B	BASELINE [ X ]
NASA FMEA #: 05-6BA-2404-3	NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31142  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ]	[ ]	[ ]	[ ]	[ ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ ]
INADEQUATE	[ ]

## REMARKS:

IOA RECOMMENDS THAT FMEA 2404-3 BE DELETED BECAUSE IT IS NOT A CREDIBLE FAILURE MODE FOR THE TYPE 1 HDC. NASA INCORPORATED 2403 INTO 2403-1.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31146  
NASA FMEA #: 05-6BA-2405-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31146  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE TYPE 1 HDC'S, IOA RECOMMENDS COMBINING FMEA'S 2405-1 AND 2405-2 TOGETHER TO CONFORM TO NSTS 22206.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31146A	BASELINE [ X ]
NASA FMEA #: 05-6BA-2405-2	NEW [   ]

SUBSYSTEM: EPD&C  
MDAC ID: 31146  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[   ]	[   ]	[   ]	[   ] *
IOA	[ 3 / 3 ]	[   ]	[   ]	[   ]	[   ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ] (ADD/DELETE)
-----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE TYPE 1 HDC'S. IOA RECOMMENDS COMBINING FMEA'S 2405-1 AND 2405-2 TOGETHER TO CONFORM TO NSTS 22206. NASA INCORPORATE 2405-2 INTO 2405-1.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31146B  
NASA FMEA #: 05-6BA-2405-3

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31146  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

REMARKS:

IOA DOES NOT CONCUR WITH NASA'S EVALUATION OF THE TYPE 1 HDC'S  
AND RECOMMENDS DELETING FMEA 2405-3 BECAUSE ITS FAILURE MODE IS  
NOT A CREDIBLE FAILURE. NASA INCORPORATED 2405-3 INTO 2405-1.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31145  
NASA FMEA #: 05-6BA-2406-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31145  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ /N ]	[ N ]	[ N ]	[ N ]	[ ]

## RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ]	[ ]	[ ]	[ ]	[ ]
----------	-----	-----	-----	-----

(ADD/DELETE)

## \* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES NOT CONCUR FULLY WITH NASA'S EVALUATION OF THE TYPE 1  
HDC, IOA RECOMMENDS: DOWNGRADING CRITICALITY TO 3/3 AND COMBINING  
FMEA'S 2406-1 AN 2406-2 TOGETHER TO CONFORM TO NSTS 22206.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31145A	BASELINE [ X ]
NASA FMEA #: 05-6BA-2406-2	NEW [   ]

SUBSYSTEM: EPD&C  
MDAC ID: 31145  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[   ]	[   ]	[   ]	[   ] *
IOA	[ 3 / 3 ]	[   ]	[   ]	[   ]	[   ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]	(ADD/DELETE)
-----------	-------	-------	-------	-------	--------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE TYPE 1 HDC, IOA RECOMMENDS: COMBINING FMEA'S 2406-1 AN 2406-2 TOGETHER TO CONFORM TO NSTS 22206.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
 ASSESSMENT ID: LDGDEC-31145B  
 NASA FMEA #: 05-6BA-2406-3

NASA DATA:  
 BASELINE [ X ]  
 NEW [ ]

SUBSYSTEM: EPD&C  
 MDAC ID: 31145  
 ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY			REDUNDANCY SCREENS			CIL ITEM
	FLIGHT	HDW/FUNC		A	B	C	
NASA	[ 3 / 3 ]			[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]			[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]			[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
 (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
 INADEQUATE [ ]

## REMARKS:

IOA DOES NOT CONCUR WITH NASA'S EVALUATION OF THE TYPE 1 HDC AND RECOMMENDS DELETING FMEA 2406-3 BECAUSE ITS FAILURE MODE IS NOT A CREDIBLE FAILURE. NASA INCORPORATE 2406-3 INTO 2406-1, WHICH IS NOT VIABLE AS 2406-1 HAS 3/1R CRITICALITY.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31158  
NASA FMEA #: 05-6BA-2407-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31158  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 3 /3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE TYPE 1 HDC'S.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31158A	BASELINE [ X ]
NASA FMEA #: 05-6BA-2407-2	NEW [   ]

SUBSYSTEM: EPD&C  
MDAC ID: 31158  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 3 /3 ]	[   ]	[   ]	[   ]	[   ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE TYPE 1 HDC'S.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31158B  
NASA FMEA #: 05-6BA-2407-3

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31158  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES NOT CONCUR WITH NASA'S EVALUATION OF THE TYPE 1 HDC'S  
FOR FMEA 2407-3 AND RECOMMENDS THAT IT BE DELETED BECAUSE IT IS  
NOT A CREDIBLE FAILURE MODE. NASA INCORPORATE 2407-3 INTO 2407-1

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31156  
NASA FMEA #: 05-6BA-2408-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31156  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE TYPE 1 HDC'S.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31157  
NASA FMEA #: 05-6BA-2408-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31157  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE TYPE ONE HDC'S FOR THE  
FAILURE MODE: INADVERTENT OUTPUT.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE:	1/23/87	NASA DATA:
ASSESSMENT ID:	LDGDEC-31154	BASELINE [ X ]
NASA FMEA #:	05-6BA-2409-1	NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31154  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 3)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ N ]	[ ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ / ]	[ ]	[ F ]	[ ]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ ]
INADEQUATE	[ ]

## REMARKS:

IOA DOES NOT CONCUR WITH NASA'S EVALUATION OF THE TYPE 3 HDC'S.  
IOA RECOMMENDS ADDING THE FMEA TO THE CIL BECAUSE IT FAILS  
REDUNDANCY SCREEN B.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31159  
NASA FMEA #: 05-6BA-2410-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31159  
ITEM: HYBRID DRIVER CONTROLLER (TYPE II)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE TYPE 2 HDC'S.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31160	BASELINE [ X ]
NASA FMEA #: 05-6BA-2410-2	NEW [   ]

SUBSYSTEM: EPD&C  
MDAC ID: 31160  
ITEM: HYBRID DRIVER CONTROLLER (TYPE II)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ / ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[ / ]	[   ]	[   ]	[   ]	[   ]	[   ]	(ADD/DELETE)
-------	-------	-------	-------	-------	-------	--------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE TYPE 2 HDC'S.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31152  
NASA FMEA #: 05-6BA-2413-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31152  
ITEM: HYBRID DRIVER CONTROLLER (TYPE II)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ X ] *
IOA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE TYPE 2 HDC'S.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
 ASSESSMENT ID: LDGDEC-31153  
 NASA FMEA #: 05-6BA-2413-2

NASA DATA:  
 BASELINE [ X ]  
 NEW [ ]

SUBSYSTEM: EPD&C  
 MDAC ID: 31153  
 ITEM: HYBRID DRIVER CONTROLLER (TYPE II)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
 (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
 INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE TYPE 2 HDC'S FOR THE  
 FAILURE MODE: INADVERTENT OUTPUT.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31150  
NASA FMEA #: 05-6BA-2415-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31150  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ X ] *
IOA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE TYPE 1 HDC'S.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31151  
NASA FMEA #: 05-6BA-2415-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31151  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE TYPE 1 HDC'S FOR THE  
FAILURE MODE: INADVERTENT OUTPUT.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31168  
NASA FMEA #: 05-6BA-2501-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31168  
ITEM: LATCHING RELAY (6), LDG GR 'ARM' CONTROL  
CIRCUITS

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF (FMEA 2501-1) THE LATCHING RELAYS FOR THE LANDING GEAR ARM CONTROL CIRCUITS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31168A	BASELINE [ X ]
NASA FMEA #: 05-6BA-2501-2	NEW [   ]

SUBSYSTEM: EPD&C  
MDAC ID: 31168  
ITEM: LATCHING RELAY (6), LDG GR 'ARM' CONTROL  
CIRCUITS

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]
IOA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[ D ] (ADD/DELETE)
-----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

	ADEQUATE [   ]
	INADEQUATE [ X ]

## REMARKS:

FMEA 2501-2 HAS A NONCREDIBLE FAILURE MODE (SHORTS TO GROUND) AND IOA RECOMMENDS THAT THE FMEA AND ITS CIL BE DELETED. NASA INCORPORATED 2501-2 INTO 2501-1.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31169  
NASA FMEA #: 05-6BA-2501-3

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31169  
ITEM: LATCHING RELAY (6), LDG GR 'ARM' CONTROL  
CIRCUITS

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE LDG GR "ARM" LATCHING  
RELAYS FOR THE FAILURE MODE: FAILS CLOSED.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31170	BASELINE [ X ]
NASA FMEA #: 05-6BA-2502-1	NEW [   ]

SUBSYSTEM: EPD&C  
MDAC ID: 31170  
ITEM: LATCHING RELAY (6), LDG GR 'DOWN' CONTROL CIRCUITS

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ X ] *
IOA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ]
COMPARE	[   /   ]	[   ]	[ N ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]	[   ]
					(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF FMEA 2502-1 FOR THE LDG GR "DOWN" RELAYS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31170A  
NASA FMEA #: 05-6BA-2502-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31170  
ITEM: LATCHING RELAY (6), LDG GR 'DOWN' CONTROL CIRCUITS

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ D ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ X ]

## REMARKS:

IOA RECOMMENDS THAT FMEA 2502-2 AND ITS CIL BE DELETED, BECAUSE IT HAS A NONCREDIBLE FAILURE MODE: SHORTS TO GROUND. NASA INCORPORATE 2502-2 INTO 2502-1.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31171	BASELINE [ X ]
NASA FMEA #: 05-6BA-2502-3	NEW [   ]

SUBSYSTEM: EPD&C  
MDAC ID: 31171  
ITEM: LATCHING RELAY (6), LDG GR 'DOWN' CONTROL CIRCUITS

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[   ]
COMPARE	[ N /   ]	[   ]	[   ]	[   ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]	[   ]	(ADD/DELETE)
-----------	-------	-------	-------	-------	-------	--------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[   ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF FMEA 2502-3 FOR THE FAILURE MODE: FAILS CLOSED.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31172  
NASA FMEA #: 05-6BA-2503-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31172  
ITEM: GENERAL PURPOSE RELAY (2)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ /N ]	[ N ]	[ N ]	[ N ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE GENERAL PURPOSE RELAYS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31172A  
NASA FMEA #: 05-6BA-2503-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31172  
ITEM: GENERAL PURPOSE RELAY (2)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE GENERAL PURPOSE RELAYS.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31172B  
NASA FMEA #: 05-6BA-2503-3

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31172  
ITEM: GENERAL PURPOSE RELAY (2)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES NOT CONCUR WITH NASA'S EVALUATION OF THE GENERAL PURPOSE RELAYS. IOA RECOMMENDS DELETING FMEA 2503-3, BECAUSE ITS FAILURE MODE (SHORTS TO GROUND) IS NOT A CREDIBLE FAILURE FOR THESE COMPONENTS. NASA INCORPORATED 2503-3 INTO 2503-1.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
 ASSESSMENT ID: LDGDEC-31173  
 NASA FMEA #: 05-6BA-2550-1

NASA DATA:  
 BASELINE [ X ]  
 NEW [ ]

SUBSYSTEM: EPD&C  
 MDAC ID: 31173  
 ITEM: REMOTE POWER CONTROLLER (2), 3 AMPS

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
 (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
 INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE RPC'S FOR A LOSS OF  
 OUTPUT FAILURE MODE.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31173A  
NASA FMEA #: 05-6BA-2550-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31173  
ITEM: REMOTE POWER CONTROLLER (2), 3 AMPS

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES NOT CONCUR WITH NASA'S EVALUATION OF THE RPC'S, IOA RECOMMENDS DELETING FMEA 2550-2, BECAUSE ITS FAILURE MODE (SHORTS TO GROUND) IS NOT A CREDIBLE FAILURE FOR THE RPC'S.: NASA INCORPORATED 2550-2 INTO 2550-1.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
 ASSESSMENT ID: LDGDEC-31174  
 NASA FMEA #: 05-6BA-2550-3

NASA DATA:  
 BASELINE [ X ]  
 NEW [ ]

SUBSYSTEM: EPD&C  
 MDAC ID: 31174  
 ITEM: REMOTE POWER CONTROLLER (2), 3 AMPS

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	* [ ]
IOA	[ 3 / 1R ]	[ P ]	[ P ]	[ P ]	
COMPARE	[ / N ]	[ N ]	[ N ]	[ N ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
 (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
 INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION FOR THE REDUNDANT SHUTOFF  
 VALVE RPC'S.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31175  
NASA FMEA #: 05-6BA-2575-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31175  
ITEM: PYRO INITIATOR CONTROLLER (6)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION FOR THE PIC'S WITH THE FAILURE  
MODE: LOSS OF OUTPUT.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31176	BASELINE [ X ]
NASA FMEA #: 05-6BA-2575-2	NEW [   ]

SUBSYSTEM: EPD&C  
MDAC ID: 31176  
ITEM: PYRO INITIATOR CONTROLLER (6)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[   ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[   ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]	[   ]	(ADD/DELETE)
-----------	-------	-------	-------	-------	-------	--------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION FOR THE PIC'S WITH THE FAILURE  
MODE: INADVERTENT OUTPUT.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31177  
NASA FMEA #: 05-6BA-2576-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31177  
ITEM: PYRO INITIATOR CONTROLLER (2)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE PIC'S.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31178	BASELINE [ X ]
NASA FMEA #: 05-6BA-2576-2	NEW [   ]

SUBSYSTEM: EPD&C  
MDAC ID: 31178  
ITEM: PYRO INITIATOR CONTROLLER (2)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[   ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[   ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ] (ADD/DELETE)
-----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE PIC'S FOR THE FAILURE MODE: INADVERTENT OUTPUT. NASA TRANSFERRED FMEA TO ANOTHER SUBSYSTEM.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31183  
NASA FMEA #: 05-6BA-2578-1

NASA DATA:  
BASELINE [    ]  
NEW [ X ]

SUBSYSTEM: EPD&C  
MDAC ID: 31183  
ITEM: DIODE, 12 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[    ] *
IOA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ N /    ]	[    ]	[ N ]	[    ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [    ]  
INADEQUATE [    ]

## REMARKS:

IOA RECOMMENDS ADDING THE ISOLATION DIODE TO NASA'S CIL. THE DIODE ISOLATES THE K6 & K7 ARM RELAYS FROM THE K8 DOWN RELAYS; DIODE IS ALSO IN THE CIRCUIT SUPPLYING POWER TO THE LDG GEAR CONTROL VALVE AND THE LDG GEAR DUMP CONTROL VALVE. POSSIBLE LOSS OF CREW/VEHICLE BECAUSE OF LOSS OF POWER TO OPERATE THESE VALVES IF THE DIODE FAILS OPEN.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31184  
NASA FMEA #: 05-6BA-2578-2

NASA DATA:  
BASELINE [ ]  
NEW [ X ]

SUBSYSTEM: EPD&C  
MDAC ID: 31184  
ITEM: DIODE, 12 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH THE NASA FMEA EVALUATION FOR THE ISOLATION DIODE  
FAILURE MODE: SHORTS, LOW RESISTANCE.

C-4

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31185  
NASA FMEA #: 05-6BA-2580-1

NASA DATA:  
BASELINE [    ]  
NEW [ X ]

SUBSYSTEM: EPD&C  
MDAC ID: 31185  
ITEM: DIODE, 12 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[    ]	[    ]	[    ]	[    ] *
IOA	[ 3 / 3 ]	[    ]	[    ]	[    ]	[    ]
COMPARE	[    /    ]	[    ]	[    ]	[    ]	[    ]

RECOMMENDATIONS: (If different from NASA)

[    /    ]    [    ]    [    ]    [    ]    [    ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [    ]  
INADEQUATE [    ]

## REMARKS:

IOA CONCURS WITH THE NASA EVALUATION FOR THE TRANSIENT  
SUPPRESSION DIODE FAILURE MODES: FAILS OPENS, FAILS TO CONDUCT.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31185A  
NASA FMEA #: 05-6BA-2580-2

NASA DATA:  
BASELINE [    ]  
NEW [ X ]

SUBSYSTEM: EPD&C  
MDAC ID: 31185  
ITEM: DIODE, 12 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[    ]	[    ]	[    ]	[    ] *
IOA	[ 3 / 3 ]	[    ]	[    ]	[    ]	[    ]
COMPARE	[    /    ]	[    ]	[    ]	[    ]	[    ]

RECOMMENDATIONS: (If different from NASA)

[    /    ]    [    ]    [    ]    [    ]    [    ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [    ]  
INADEQUATE [    ]

## REMARKS:

IOA CONCURS WITH THE NASA EVALUATION FOR THE TRANSIENT  
SUPPRESSION DIODE FAILURE MODES: FAILS OPENS, FAILS TO CONDUCT.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31213  
NASA FMEA #: 05-6BB-2096-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31213  
ITEM: GENERAL PURPOSE RELAY, NONLATCHING (2)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ X ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ N / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF FMEA 2096-1 FOR THE  
GENERAL PURPOSE RELAYS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31213A	BASELINE [ X ]
NASA FMEA #: 05-6BB-2096-2	NEW [   ]

SUBSYSTEM: EPD&C  
MDAC ID: 31213  
ITEM: GENERAL PURPOSE RELAY, NONLATCHING (2)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]	[   ]	
						(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

IOA RECOMMENDS THAT FMEA 2096-2 BE DELETED, BECAUSE IT IS A NON-CREDIBLE FAILURE MODE (SHORTS TO GROUND) FOR THE NONLATCHING RELAYS. NASA INCORPORATED FMEA 2096-2 INTO 2096-3.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31214  
NASA FMEA #: 05-6BB-2096-3

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31214  
ITEM: GENERAL PURPOSE RELAY, NONLATCHING (2)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ X ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ N / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE GENERAL PURPOSE RELAYS  
FOR THE FAILURE MODE: FAILS OPEN.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31215	BASELINE [ X ]
NASA FMEA #: 05-6BB-2101-1	NEW [   ]

SUBSYSTEM: EPD&C  
MDAC ID: 31215  
ITEM: BLOCKING DIODES (9), 3 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 3 /3 ]	[   ]	[   ]	[   ]	[   ] *
IOA	[ 3 /3 ]	[   ]	[   ]	[   ]	[   ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ] (ADD/DELETE)
-----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE BLOCKING DIODES IN THE ANTI-SKID FAIL LIGHT/CONTROL CIRCUIT.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31211  
NASA FMEA #: 05-6BB-2102-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31211  
ITEM: BLOCKING DIODE (4), 12 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ / ]	[ ]	[ N ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE BLOCKING DIODES.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31212	BASELINE [    ]
NASA FMEA #: 05-6BB-2102-2	NEW [ X ]

SUBSYSTEM: EPD&C  
MDAC ID: 31212  
ITEM: BLOCKING DIODE (4), 12 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ] *
IOA	[ 3 /3 ]	[    ]	[    ]	[    ]	[    ]
COMPARE	[    /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[    /    ]	[    ]	[    ]	[    ]	[    ] (ADD/DELETE)
-------------	--------	--------	--------	------------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ X ]
INADEQUATE	[    ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31222  
NASA FMEA #: 05-6BB-2106-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31222  
ITEM: TOGGLE SWITCH

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION FOR THE BRAKE CONTROL  
CIRCUIT TOGGLE SWITCHES FOR MAINS A, B, AND C

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31222A  
NASA FMEA #: 05-6BB-2106-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31222  
ITEM: TOGGLE SWITCH

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA RECOMMENDS THAT FMEA 2106-2 BE DELETED, BECAUSE IT IS A NON-CREDIBLE FAILURE MODE (SHORTS TO GROUND) FOR THE TOGGLE SWITCHES. NASA INCORPORATED FMEA 2106-2 INTO 2106-1.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31223  
NASA FMEA #: 05-6BB-2106-3

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31223  
ITEM: TOGGLE SWITCH

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION FOR THE BRAKE CONTROL  
TOGGLE SWITCHES FOR MAINS A, B, AND C.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31225  
NASA FMEA #: 05-6BB-2106-3

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31225  
ITEM: TOGGLE SWITCH, DPST

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ /N ]	[ N ]	[ N ]	[ N ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES NOT CONCUR WITH NASA'S EVALUATION OF THE "MAIN C" TOGGLE SWITCH AND IOA RECOMMENDS CHANGING THE CRITICALITY TO 3/3

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31220  
NASA FMEA #: 05-6BB-2107-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31220  
ITEM: TOGGLE SWITCH, DPST

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ N / ]	[ ]	[ ]	[ ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF FMEA 2107-1 FOR THE  
ANTI-SKID CONTROL CIRCUIT TOGGLE SWITCH.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31220A  
NASA FMEA #: 05-6BB-2107-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31220  
ITEM: TOGGLE SWITCH, DPST

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ N / ]	[ ]	[ ]	[ ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA RECOMMENDS THAT FMEA 2107-2 BE DELETED, BECAUSE IT IS A NON-CREDIBLE FAILURE MODE (SHORTS TO GROUND) FOR THE TOGGLE SWITCH. NASA INCORPORATED FMEA 2107-2 INTO 2107-1.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31221  
NASA FMEA #: 05-6BB-2107-3

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31221  
ITEM: TOGGLE SWITCH, DPST

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ /N ]	[ N ]	[ N ]	[ N ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ]	[ ]	[ ]	[ ]	[ ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES NOT CONCUR WITH NASA'S EVALUATION AND RECOMMENDS DOWN-GRADING FMEA 2107-3 TO CRITICALITY 3/3.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31203  
NASA FMEA #: 05-6BB-2111-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31203  
ITEM: ANNUNCIATOR

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE ANTI-SKID FAIL LIGHT.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31201  
NASA FMEA #: 05-6BB-2240-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31201  
ITEM: GENERAL PURPOSE FUSE (8), 3 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE GENERAL PURPOSE FUSE FOR THE "OPENS" FAILURE MODE.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31205	BASELINE [ X ]
NASA FMEA #: 05-6BB-2241-1	NEW [   ]

SUBSYSTEM: EPD&C  
MDAC ID: 31205  
ITEM: GENERAL PURPOSE FUSE (8), 2 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]
COMPARE	[ N /   ]	[   ]	[ N ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[   ]	
					(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

IOA DOES NOT CONCUR FULLY WITH NASA'S EVALUATION OF THE GENERAL PURPOSE FUSES. IOA RECOMMENDS: CHANGING THE REDUNDANCY SCREENS SINCE IT FAILS REDUNDANCY SCREEN B, AND DOWNGRADING FMEA TO A 3/1R..

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31204  
NASA FMEA #: 05-6BB-2242-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31204  
ITEM: FUSE, 1 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE FUSE IN THE BRAKE/SKID CONTROL CIRCUITRY.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31202  
NASA FMEA #: 05-6BB-2246-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31202  
ITEM: FUSE (5 AMP), 2

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ N / ]	[ ]	[ ]	[ ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31206  
NASA FMEA #: 05-6BB-2247-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31206  
ITEM: ISOLATION RESISTOR (3), 5.1K, 1/4 WATT

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE MDM ISOLATION RESISTORS  
FOR THE BRAKE-SKID POWER/SWITCH SCAN MONITORING CIRCUITS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31200  
NASA FMEA #: 05-6BB-2248-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31200  
ITEM: ISOLATION RESISTOR, 5.1K, 1/4W

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE ISOLATION RESISTOR.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31210  
NASA FMEA #: 05-6BB-2249-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31210  
ITEM: CURRENT LIMITING RESISTOR (4), 1.21K, 2W

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ / ]	[ ]	[ N ]	[ ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ F ] [ ] [ A ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES NOT CONCUR FULLY WITH NASA'S EVALUATION OF THE RPC CONTROL CIRCUIT CURRENT LIMITING RESISTORS. IOA RECOMMENDS (1) CHANGING THE REDUNDANCY SCREENS (2) ADDING FMEA 2249-1 TO THE CIL SINCE IT FAILS REDUNDANCY SCREEN B.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31208  
NASA FMEA #: 05-6BB-2249-2

NASA DATA:  
BASELINE [ ]  
NEW [ X ]

SUBSYSTEM: EPD&C  
MDAC ID: 31208  
ITEM: CURRENT LIMITING RESISTOR (4), 1.21K, 2W

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH THE NASA FMEA EVALUATION.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31209  
NASA FMEA #: 05-6BB-2250-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31209  
ITEM: RESISTOR (4), 1.8K, 1/4W

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S ANALYSIS OF THE RPC/MDM MONITOR CIRCUIT  
BLEED-OFF RESISTORS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31207	BASELINE [ X ]
NASA FMEA #: 05-6BB-2253-1	NEW [   ]

SUBSYSTEM: EPD&C  
MDAC ID: 31207  
ITEM: ISOLATION RESISTOR (4), 2.2K, 1/2W

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY		REDUNDANCY SCREENS			CIL ITEM
	FLIGHT	HDW/FUNC	A	B	C	
NASA	[ 3 / 3 ]		[   ]	[   ]	[   ]	[   ] *
IOA	[ 3 / 3 ]		[   ]	[   ]	[   ]	[   ]
COMPARE	[   /   ]		[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE MONITORING CIRCUIT ISOLATION RESISTORS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31226  
NASA FMEA #: 05-6BB-2256-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31226  
ITEM: REMOTE POWER CONTROLLER (4), 10 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE RPC'S THAT CONNECT OR DISCONNECT MAIN DC BUS POWER TO THE ANTI-SKID CONTROL UNITS FOR THE FAILURE MODE: LOSS OF OUTPUT.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31227  
NASA FMEA #: 05-6BB-2256-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31227  
ITEM: REMOTE POWER CONTROLLER (4), 10 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	A	B	C	CIL ITEM
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE RPC'S FOR THE  
FAILURE MODE: INADVERTENT OUTPUT.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31216  
NASA FMEA #: 05-6BB-2262-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31216  
ITEM: HYBRID DRIVER CONTROLLER (3), TYPE 1

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ X ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ N / ]	[ ]	[ N ]	[ ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE TYPE 1 HDC'S.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31218  
NASA FMEA #: 05-6BB-2262-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31218  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 3)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE TYPE 3 HDC FOR THE  
FAILURE MODE: LOSS OF OUTPUT.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31217  
NASA FMEA #: 05-6BB-2262-2

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31217  
ITEM: HYBRID DRIVER CONTROLLER (3), TYPE 1

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	A	B	C	CIL ITEM
NASA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ] *
IOA	[ 3 /3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ X ]  
INADEQUATE [ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE TYPE 1 HDC'S FOR  
THE FAILURE MODE: INADVERTENT OUTPUT.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31219	BASELINE [ X ]
NASA FMEA #: 05-6BB-2262-2	NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31219  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 3)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ]	[ ]	[ ]	[ ]	[ ] (ADD/DELETE)
-------	-----	-----	-----	---------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ ]
INADEQUATE	[ ]

## REMARKS:

IOA DOES CONCUR WITH NASA'S EVALUATION OF THE TYPE 3 HDC FOR THE  
FAILURE MODE: INADVERTENT OUTPUT.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 2/22/88  
ASSESSMENT ID: LDGDEC-31240X  
NASA FMEA #: 05-6BB-2270-1

NASA DATA:  
BASELINE [   X   ]  
NEW [   X   ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 31240  
ITEM: RESISTOR (1 OHM) (2W) ANTI-SKID VLV COIL CURRENT  
MEASUREMENT

LEAD ANALYST: P. BYNUM

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ] *
IOA	[ 2 /1R ]	[ P ]	[ P ]	[ P ]	[ X ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]   [   ]   [   ]   [   ]   [   ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [   X   ]  
INADEQUATE [   ]

## REMARKS:

IOA CONCURS WITH THE NASA FMEA/CIL EVALUATION

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31102	BASELINE [ X ]
NASA FMEA #: 06-6BA-2118-1	NEW [   ]

SUBSYSTEM: EPD&C  
MDAC ID: 31102  
ITEM: PROXIMITY SENSOR BOX (2)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[   ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[   ]
COMPARE	[   /   ]	[   ]	[   ]	[   ]	[   ]

RECOMMENDATIONS: (If different from NASA)

[   /   ]	[   ]	[   ]	[   ]	[   ] (ADD/DELETE)
-----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[   ]
INADEQUATE	[   ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE PROXIMITY SENSOR BOXES  
FOR THE FAILURE MODE: LOSS OF OUTPUT.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31103  
NASA FMEA #: 06-6BA-2200-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31103  
ITEM: ISOLATION DIODE (3), 1 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE ISOLATION DIODES. LOSS OF DIODE FUNCTION NOT CRITICAL TO MISSION/CREW/VEHICLE.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87	NASA DATA:
ASSESSMENT ID: LDGDEC-31103A	BASELINE [    ]
NASA FMEA #: 06-6BA-2200-2	NEW [ X ]

SUBSYSTEM: EPD&C  
MDAC ID: 31103  
ITEM: ISOLATION DIODE (3), 1 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS		CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[    ]	[    ]	[    ]	[    ] *
IOA	[ 3 / 3 ]	[    ]	[    ]	[    ]	[    ]
COMPARE	[    /    ]	[    ]	[    ]	[    ]	[    ]

RECOMMENDATIONS: (If different from NASA)

[    /    ]	[    ]	[    ]	[    ]	[    ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[    ]
INADEQUATE	[    ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION OF THE ISOLATION DIODES. LOSS OF DIODE FUNCTION NOT CRITICAL TO MISSION/CREW/VEHICLE.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31104  
NASA FMEA #: 06-6BA-2201-1

NASA DATA:  
BASELINE [ X ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31104  
ITEM: ISOLATION DIODE (3), 1 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION, LOSS OF DIODE FUNCTION NOT  
CRITICAL TO MISSION/CREW/VEHICLE.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31104A  
NASA FMEA #: 06-6BA-2201-2

NASA DATA:  
BASELINE [ ]  
NEW [ X ]

SUBSYSTEM: EPD&C  
MDAC ID: 31104  
ITEM: ISOLATION DIODE (3), 1 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ / ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ / ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA CONCURS WITH NASA'S EVALUATION, LOSS OF DIODE FUNCTION NOT  
CRITICAL TO MISSION/CREW/VEHICLE.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-10210  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [ ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10210  
ITEM: STEERING DISCONNECT LOCK

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ / ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ A ] (ADD/DELETE)
----------	-------	-------	-------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ X ]

## REMARKS:

NOT CONSIDERED BY THE NASA FMEA/CIL

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
 ASSESSMENT ID: LDGDEC-10212  
 NASA FMEA #: NONE

NASA DATA:  
 BASELINE [ ]  
 NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
 MDAC ID: 10212  
 ITEM: NOSE WHEEL RETAINING BOLT

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ / ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
 INADEQUATE [ ]

## REMARKS:

ADDITIONAL DATA UNCOVERED AFTER STUDY COMPLETION ELIMINATES  
 THIS IOA EVALUATION REPORT

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-10213  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [ ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10213  
ITEM: AXLE

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ / ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ X ]

REMARKS:

NOT EVALUATED BY NASA

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-10220  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [    ]  
NEW [    ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10220  
ITEM: TORQUE TUBE ASSEMBLY

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[    /    ]	[    ]	[    ]	[    ]	[    ] *
IOA	[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [    ]  
INADEQUATE [ X ]

## REMARKS:

THE WORST CASE SCENARIO FOR A BROKEN TORQUE TUBE ASSEMBLY WOULD BE A FAILURE THAT WOULD PREVENT THE NLG FROM LOCKING IN THE EXTENDED POSITION. SIMILAR TO MLG TORQUE TUBE ASSY REF 02-1-010-1.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-10401  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [    ]  
NEW [    ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 10401  
ITEM: NLG EXTEND / RETRACT HYD STRUT ACT

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[    /    ]	[    ]	[    ]	[    ]	[    ] *
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[    ]
COMPARE	[ N / N ]	[ N ]	[ N ]	[ N ]	[    ]

RECOMMENDATIONS: (If different from NASA)

[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [    ]  
INADEQUATE [    ]

## REMARKS:

THERE IS NO NASA FMEA COVERING THIS FAILURE.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-10601	BASELINE [    ]
NASA FMEA #: NONE	NEW [    ]

SUBSYSTEM:            LANDING/DECELERATION SYSTEMS  
MDAC ID:              10601  
ITEM:                  NLG DOOR OVER-CENTER BUNGEE

LEAD ANALYST:        J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY	SCREENS	CIL ITEM
		A	B	C
NASA	[    /    ]	[    ]	[    ]	[    ]
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[    ]
COMPARE	[ N / N ]	[ N ]	[ N ]	[    ]

RECOMMENDATIONS:    (If different from NASA)

[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ A ] (ADD/DELETE)
-----------	--------	--------	--------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[    ]
INADEQUATE	[    ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-11004  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [    ]  
NEW [    ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 11004  
ITEM: NLG UPLOCK ACTUATOR

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[    /    ]	[    ]	[    ]	[    ]	[    ] *
IOA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [    ]  
INADEQUATE [    ]

## REMARKS:

GEAR WILL NOT RELEASE HYDRAULICALLY. THE PYRO BACKUP WILL RELEASE THE GEAR ONE SECOND AFTER THE COMMAND TO DEPLOY IF THE LANDING GEAR HOOK IS NOT OPEN.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-11302  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [ ]  
NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 11302  
ITEM: NLG DOOR BUNGEE ASSIST ASSY

LEAD ANALYST: W.WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ / ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ X ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ 1 /1 ]	[ NA]	[ NA]	[ NA]	[ A ]
----------	-------	-------	-------	-------

(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

BUNGEE COULD POSSIBLY INADVERTENTLY RELEASE CAUSING THE NLG DOOR TO CRACK OPEN.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20205  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [    ]  
NEW [    ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20205  
ITEM: AXLE KIT - MLG

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[    /    ]	[    ]	[    ]	[    ]	[    ] *
IOA	[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [    ]  
INADEQUATE [    ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-20214	BASELINE [    ]
NASA FMEA #: NONE	NEW [    ]

SUBSYSTEM:            LANDING/DECELERATION SYSTEMS  
MDAC ID:              20214  
ITEM:                  WEIGHT ON WHEELS SENSOR - MLG

LEAD ANALYST:        W. WEISSINGER

ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[    /    ]	[    ]	[    ]	[    ]
IOA	[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]
COMPARE	[ N / N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS:    (If different from NASA)

[ 3 / 3 ]	[ NA ]	[ NA ]	[ NA ]	[ A ] (ADD/DELETE)
-----------	--------	--------	--------	-----------------------

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[    ]
INADEQUATE	[    ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20215  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [    ]  
NEW [    ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20215  
ITEM: WEIGHT ON WHEELS SENSOR - MLG

LEAD ANALYST: W. WEISSINGER

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[    /    ]	[    ]	[    ]	[    ]	[    ] *
IOA	[ 3 /3 ]	[ NA ]	[ NA ]	[ NA ]	[    ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[    ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ]	[ NA ]	[ NA ]	[ NA ]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [    ]  
INADEQUATE [    ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86 NASA DATA:  
ASSESSMENT ID: LDGDEC-20401 BASELINE [ ]  
NASA FMEA #: NONE NEW [ ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20401  
ITEM: MLG EXTEND / RETRACT HYD STRUT ACT

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ / ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[ ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ]	[ NA]	[ NA]	[ NA]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[ ]
INADEQUATE	[ ]

## REMARKS:

THERE IS NO NASA FMEA COVERING THIS FAILURE.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-20601  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [    ]  
NEW [    ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 20601  
ITEM: DOOR OVER-CENTER BUNGEE

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[    /    ]	[    ]	[    ]	[    ]	[    ] *
IOA	[ 3 /3 ]	[ NA ]	[ NA ]	[ NA ]	[    ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[    ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ]	[ NA ]	[ NA ]	[ NA ]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [    ]  
INADEQUATE [    ]

REMARKS:

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-21004  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [    ]  
NEW [    ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 21004  
ITEM: MLG UPLOCK ACTUATOR

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[    /    ]	[    ]	[    ]	[    ]	[    ] *
IOA	[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ 2 /1R ]	[ P ]	[ F ]	[ P ]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [    ]  
INADEQUATE [    ]

## REMARKS:

GEAR WILL NOT RELEASE HYDRAULICALLY. THE PYRO BACKUP WILL RELEASE THE GEAR ONE SECOND AFTER THE COMMAND TO DEPLOY IF THE LANDING GEAR HOOK IS NOT OPEN.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86  
ASSESSMENT ID: LDGDEC-30104  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [    ]  
NEW [    ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30104  
ITEM: BRAKE PEDAL TRANSDUCER

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[    /    ]	[    ]	[    ]	[    ]	[    ] *
IOA	[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ X ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ 1 /1 ]	[ NA ]	[ NA ]	[ NA ]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [    ]  
INADEQUATE [    ]

## REMARKS:

CLOSED LVDT WILL RESULT IN HALF-WHEEL LOCKUP ON LANDING WITH ANTISKID OFF, CAUSING POSSIBLE LOSS OF VEHICLE. ANTISKID WILL PROVIDE PROTECTION IF ON.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 12/15/86	NASA DATA:
ASSESSMENT ID: LDGDEC-30127	BASELINE [    ]
NASA FMEA #: NONE	NEW [    ]

SUBSYSTEM: LANDING/DECELERATION SYSTEMS  
MDAC ID: 30127  
ITEM: TRANSDUCERS, SENSORS (INSTRUMENTATION)

LEAD ANALYST: J. COMPTON

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS		CIL ITEM
		A	B	C
NASA	[    /    ]	[    ]	[    ]	[    ]
IOA	[ 3 /3 ]	[ NA ]	[ NA ]	[ NA ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ]	[ NA ]	[ NA ]	[ NA ]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE	[    ]
INADEQUATE	[    ]

REMARKS:



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31161  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [    ]  
NEW [    ]

SUBSYSTEM: EPD&C  
MDAC ID: 31161  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[    /    ]	[    ]	[    ]	[    ]	[    ] *
IOA	[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ X ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /1R ]	[ P ]	[ F ]	[ P ]	[ A ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [    ]  
INADEQUATE [    ]

## REMARKS:

IOA RECOMMENDS ADDING THE UNCOVERED TYPE 1 HDC TO NASA'S FMEA/CIL. THE HDC CONNECTS MAIN BUS DC POWER TO THE "WOW2" CIRCUITS WITHIN BRAKE/SKID CONTROL BOX A.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31162  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [ ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31162  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ / ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 /3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ N /N ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA RECOMMENDS ADDING THE UNCOVERED TYPE 1 HDC TO NASA'S FMEA LIST. THE HDC CONNECTS MAIN BUS DC POWER TO THE "WOW2" CIRCUITS WITHIN BRAKE/SKID CONTROL BOX A.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31167  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [ ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31167  
ITEM: HYBRID DRIVER CONTROLLER (TYPE 1)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ / ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 /3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ N /N ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ]	[ ]	[ ]	[ ]	[ ]
----------	-----	-----	-----	-----

(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA RECOMMENDS ADDING THE UNCOVERED TYPE 1 HDC'S TO NASA'S FMEA/CIL. THE HDC'S POWER LEFT/RIGHT MAIN GEAR EVENT INDICATORS DS3 AND DS2.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31179  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [ ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31179  
ITEM: ANNUNCIATOR LIGHT (4)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ / ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 / 3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ N / N ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 3 / 3 ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA RECOMMENDS ADDING THE ANNUNCIATOR LIGHTS TO NASA'S FMEA LIST.  
THE ANNUNCIATOR LIGHTS PROVIDE VISUAL MONITORING OF THE LANDING  
GEAR "ARM" AND "DOWN" PUSH BUTTON CIRCUITS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31180  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [ ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31180  
ITEM: ANNUNCIATOR CONTROL ASSEMBLY (2)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ / ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 /3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ N /N ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA RECOMMENDS ADDING THE ANNUNCIATOR CONTROL ASSEMBLIES TO NASA'S FMEA LIST. THE ANNUNCIATOR ASSEMBLIES PROVIDE POWER TO THE LANDING GEAR "ARM" AND "DOWN" LIGHTS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31182  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [    ]  
NEW [    ]

SUBSYSTEM: EPD&C  
MDAC ID: 31182  
ITEM: NOSE LANDING GEAR BRAKE UPLOCK RELEASE CIRCUIT  
NO'S 1 & 2

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[    /    ]	[    ]	[    ]	[    ]	[    ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ X ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ N ]

RECOMMENDATIONS: (If different from NASA)

[    /    ]    [    ]    [    ]    [    ]    [    ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [    ]  
INADEQUATE [    ]

## REMARKS:

IOA HAS DONE AN ANALYSIS OF THE NOSE LANDING GEAR BRAKE UPLOCK RELEASE CIRCUITS 1 & 2 TO PROVIDE CONSISTENT COVERAGE OF THE SHUTTLE LANDING GEAR BRAKE UPLOCK RELEASE CIRCUITRY (SEE NASA FMEA 200200-1). IOA AND NASA HAVE ALREADY COVERED THESE COMPONENTS IN THEIR OWN FMEA ANALYSIS. NASA FMEA 200200-1 IS DELETED, IOA RECOMMENDS THAT THIS FMEA SHOULD NOT BE ADDED TO NASA'S FMEA LIST.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31228  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [    ]  
NEW [    ]

SUBSYSTEM: EPD&C  
MDAC ID: 31228  
ITEM: TOGGLE SWITCH (3)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[    /    ]	[    ]	[    ]	[    ]	[    ] *
IOA	[ 3 /3 ]	[    ]	[    ]	[    ]	[    ]
COMPARE	[ N /N ]	[    ]	[    ]	[    ]	[    ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ]	[    ]	[    ]	[    ]	[    ]
----------	--------	--------	--------	--------

(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [    ]  
INADEQUATE [    ]

## REMARKS:

IOA RECOMMENDS ADDING THE TOGGLE SWITCHES (FAILURE MODE: FAILS CLOSED) TO NASA'S FMEA LIST. THE TOGGLE SWITCHES PROVIDE MANUAL SWITCHING FOR DC POWER TO THE BREAK HYDRAULIC LINE HEATERS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31229  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [    ]  
NEW [    ]

SUBSYSTEM: EPD&C  
MDAC ID: 31229  
ITEM: TOGGLE SWITCH (3)

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[    /    ]	[    ]	[    ]	[    ]	[    ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[    ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[    ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[    ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [    ]  
INADEQUATE [    ]

## REMARKS:

IOA RECOMMENDS ADDING THE TOGGLE SWITCHES (FAILURE MODE: FAILS OPEN) TO NASA'S FMEA LIST. THE TOGGLE SWITCHES PROVIDE MANUAL SWITCHING FOR DC POWER TO THE BREAK HYRAULIC LINE HEATERS.



# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31230  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [ ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31230  
ITEM: CURRENT LIMITING RESISTOR (3), 1.21K, 2W

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ / ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
-----------	-------	-------	-------	-----

(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA RECOMMENDS ADDING THE CURRENT LIMITING RESISTORS (FAILURE MODE: ELEMENT OPENS) TO NASA'S FMEA LIST. THE RESISTORS LIMIT CURRENT TO THE RPC CONTROL CIRCUITS IN THE BRAKE HYDRAULIC LINE HEATERS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31231  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [ ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31231  
ITEM: CURRENT LIMITING RESISTOR (3), 1.21K, 2W

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ / ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 /3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ N /N ]	[ ]	[ ]	[ ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ] [ ] [ ] [ ] [ ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA RECOMMENDS ADDING THE CURRENT LIMITING RESISTORS (FAILURE MODE: SHORTS) TO NASA'S FMEA LIST. THE RESISTORS LIMIT CURRENT TO THE RPC CONTROL CIRCUITS IN THE BRAKE HYRAULIC LINE HEATERS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31232  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [ ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31232  
ITEM: REMOTE POWER CONTROLLER (3), 10 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ / ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[ ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /1R ]	[ P ]	[ P ]	[ P ]	[ ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA RECOMMENDS ADDING THE RPC'S (FAILURE MODE: LOSS OF OUTPUT) TO NASA'S FMEA LIST. THE RPC'S CONNECT OR DISCONNECT DC BUS POWER TO THE HYDRAULIC BRAKE HEATER COILS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31233  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [    ]  
NEW [    ]

SUBSYSTEM: EPD&C  
MDAC ID: 31233  
ITEM: REMOTE POWER CONTROLLER (3), 10 AMP

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[    /    ]	[    ]	[    ]	[    ]	[    ] *
IOA	[ 3 / 3 ]	[ P ]	[ P ]	[ P ]	[    ]
COMPARE	[ N / N ]	[ N ]	[ N ]	[ N ]	[    ]

RECOMMENDATIONS: (If different from NASA)

[ 3 / 3 ]	[    ]	[    ]	[    ]	[    ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [    ]  
INADEQUATE [    ]

## REMARKS:

IOA RECOMMENDS ADDING THE RPC'S (FAILURE MODE: INADVERTENT OUTPUT) TO NASA'S FMEA LIST. THE RPC'S CONNECT OR DISCONNECT DC BUS POWER TO THE HYDRAULIC BRAKE HEATER COILS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31234  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [ ]  
NEW [ ]

SUBSYSTEM: EPD&C  
MDAC ID: 31234  
ITEM: ISOLATION RESISTOR (3), 5.1K, 1/4W

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[ / ]	[ ]	[ ]	[ ]	[ ] *
IOA	[ 3 /3 ]	[ ]	[ ]	[ ]	[ ]
COMPARE	[ N /N ]	[ ]	[ ]	[ ]	[ ]
RECOMMENDATIONS: (If different from NASA)					
	[ 3 /3 ]	[ ]	[ ]	[ ]	[ ] (ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [ ]  
INADEQUATE [ ]

## REMARKS:

IOA RECOMMENDS ADDING THE ISOLATION RESISTORS (FAILURE MODE:  
ELEMENT OPENS, SHORTS) TO NASA'S FMEA LIST. THE RESISTORS  
RESTRICT CURRENT BETWEEN HEATER CONTROL CIRCUITS AND MDM  
MONITORING CIRCUITS.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31235  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [    ]  
NEW [    ]

SUBSYSTEM: EPD&C  
MDAC ID: 31235  
ITEM: ANNUNCIATOR CONTROL ASSEMBLY

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[    /    ]	[    ]	[    ]	[    ]	[    ] *
IOA	[ 3 / 3 ]	[    ]	[    ]	[    ]	[    ]
COMPARE	[ N / N ]	[    ]	[    ]	[    ]	[    ]

RECOMMENDATIONS: (If different from NASA)

[ 3 / 3 ]	[    ]	[    ]	[    ]	[    ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [    ]  
INADEQUATE [    ]

## REMARKS:

IOA RECOMMENDS ADDING THE ANNUNCIATOR CONTROL ASSEMBLY (FAILURE MODE: LOSS OF OUTPUT) TO NASA'S FMEA LIST. THE ASSEMBLY PROVIDES POWER TO THE ANTI-SKID FAIL ANNUNCIATOR LIGHT.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31237  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [    ]  
NEW [    ]

SUBSYSTEM: EPD&C  
MDAC ID: 31237  
ITEM: SIGNAL CONDITIONER

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[    /    ]	[    ]	[    ]	[    ]	[    ] *
IOA	[ 3 /3 ]	[ P ]	[ P ]	[ P ]	[    ]
COMPARE	[ N /N ]	[ N ]	[ N ]	[ N ]	[    ]

RECOMMENDATIONS: (If different from NASA)

[ 3 /3 ]    [    ]    [    ]    [    ]    [    ]  
(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [    ]  
INADEQUATE [    ]

## REMARKS:

IOA RECOMMENDS ADDING THE SIGNAL CONDITIONER (FAILURE MODE: LOSS OF OUTPUT, INADVERTENT OUTPUT) TO NASA'S FMEA LIST. THE SIGNAL CONDITIONER PROVIDES VOLTAGE REDUCTION AND MDM-OF1 MONITORING TO BRAKE/SKID CONTROL BOX A.

# APPENDIX C ASSESSMENT WORKSHEET

ASSESSMENT DATE: 1/23/87  
ASSESSMENT ID: LDGDEC-31239  
NASA FMEA #: NONE

NASA DATA:  
BASELINE [    ]  
NEW [    ]

SUBSYSTEM: EPD&C  
MDAC ID: 31239  
ITEM: SIGNAL CONDITIONER

LEAD ANALYST: G. BEAIRD

## ASSESSMENT:

	CRITICALITY FLIGHT HDW/FUNC	REDUNDANCY SCREENS			CIL ITEM
		A	B	C	
NASA	[    /    ]	[    ]	[    ]	[    ]	[    ] *
IOA	[ 3 / 3 ]	[ P ]	[ P ]	[ P ]	[    ]
COMPARE	[ N / N ]	[ N ]	[ N ]	[ N ]	[    ]

RECOMMENDATIONS: (If different from NASA)

[ 3 / 3 ]	[    ]	[    ]	[    ]	[    ]
				(ADD/DELETE)

\* CIL RETENTION RATIONALE: (If applicable)

ADEQUATE [    ]  
INADEQUATE [    ]

## REMARKS:

IOA RECOMMENDS ADDING THE SIGNAL CONDITIONER (FAILURE MODE: LOSS OF OUTPUT, INADVERTENT OUTPUT) TO NASA'S FMEA LIST. THE SIGNAL CONDITIONER PROVIDES VOLTAGE REDUCTION AND MDM-OF2 MONITORING TO BRAKE/SKID CONTROL BOX B.



APPENDIX D

CRITICAL ITEMS

**APPENDIX D  
POTENTIAL CRITICAL ITEMS**

NASA FMEA	MDAC-ID	ITEM	FAILURE MODE
02-1-001-1	20201	SHOCK STRUT STRUCTURE	STRUCTURAL FAILUR
02-1-001-2	20202	SHOCK STRUT PISTON	INTERNAL / EXTERN
02-1-001-2	20203	SHOCK STRUT PISTON	INTERNAL / EXTERN
02-1-002-1	20101	TIRES, MLG TYPE I	RUPTURE
02-1-003-1	20206	LOWER DRAG BRACE STRU	STRUCTURAL FAILUR
02-1-003-1	20207	UPPER DRAG BRACE TRUN	STRUCTURAL FAILUR
02-1-003-1	20219	UPPER DRAG BRACE STRU	STRUCTURAL FAILUR
02-1-003-1	20220	CENTER DRAG BRACE TRU	STRUCTURAL FAILUR
02-1-003-1	20221	LOWER DRAG BRACE TRUN	STRUCTURAL FAILUR
02-1-004-1	20204	TORQUE ARM ASSEMBLY	STRUCTURAL FAILUR
02-1-005-1	20208	LOCK BRACE ASSEMBLY	STRUCTURAL FAILUR
02-1-005-1	20222	LOCK BRACE CENTER TRU	STRUCTURAL FAILUR
02-1-006-1	20216	UPLOCK ROLLER RETAINI	STRUCTURAL FAILUR
02-1-008-1	20209	DOWN LOCK BUNGEE	PHYSICAL BINDING
02-1-008-1	20210	DOWN LOCK BUNGEE	STRUCTURAL FAILUR
02-1-010-1	20217	TORQUE TUBE ASSEMBLY	STRUCTURAL FAILUR
02-1-012-1	20501	DOOR EXTEND / RETRACT	STRUCTURAL FAILUR
02-1-012-1	20701	MLG UPLOCK HOOK ASSEM	STRUCTURAL FAILUR
02-1-012-1	20901	DOOR HOOK ACTUATUON	STRUCTURAL FAILUR
02-1-013-1	20501	DOOR EXTEND / RETRACT	STRUCTURAL FAILUR
02-1-013-1	20701	MLG UPLOCK HOOK ASSEM	STRUCTURAL FAILUR
02-1-013-1	20901	DOOR HOOK ACTUATUON	STRUCTURAL FAILUR
02-1-014-1	20501	DOOR EXTEND / RETRACT	STRUCTURAL FAILUR
02-1-014-1	20701	MLG UPLOCK HOOK ASSEM	STRUCTURAL FAILUR
02-1-014-1	20901	DOOR HOOK ACTUATUON	STRUCTURAL FAILUR
02-1-015-1	21101	MLG PYRO UPLOCK RELEA	INADVERTANT FIRIN
02-1-015-2	21102	MLG PYRO UPLOCK RELEA	FAIL TO FIRE
02-1-017-1	20218	SHOCK STRUT ATTACHING	STRUCTURAL FAILUR
02-1-018-1	20223	SUPPORT BEAM	STRUCTURAL FAILUR
02-1-019-1	20501	DOOR EXTEND / RETRACT	STRUCTURAL FAILUR
02-1-019-1	20701	MLG UPLOCK HOOK ASSEM	STRUCTURAL FAILUR
02-1-019-1	20901	DOOR HOOK ACTUATUON	STRUCTURAL FAILUR
02-1-020-1	20501	DOOR EXTEND / RETRACT	STRUCTURAL FAILUR
02-1-020-1	20701	MLG UPLOCK HOOK ASSEM	STRUCTURAL FAILUR
02-1-020-1	20901	DOOR HOOK ACTUATUON	STRUCTURAL FAILUR
02-1-021-1	20501	DOOR EXTEND / RETRACT	STRUCTURAL FAILUR
02-1-021-1	20701	MLG UPLOCK HOOK ASSEM	STRUCTURAL FAILUR
02-1-021-1	20901	DOOR HOOK ACTUATUON	STRUCTURAL FAILUR
02-1-023-2	30117	SELECTOR VALVE	LEAKAGE
02-1-023-3	30118	SELECTOR VALVE	JAMMED CLOSED
02-1-024-1	30101	ANTI-SKID SELECT SW	SWITCH MALFUNCTION
02-1-025-1	30106	BRAKE CIRCUIT	OPEN OR SHORT CIR
02-1-025-2	30105	BRAKE CIRCUIT	OPEN OR SHORT CIR
02-1-026-2	30107	SKID CIRCUIT	OPEN OR SHORT CIR
02-1-027-1	30109	ANTI-SKID FAIL CIRCUIT	OPEN OR SHORT CIR
02-1-027-2	30109	ANTI-SKID FAIL CIRCUIT	OPEN OR SHORT CIR

NASA FMEA	MDAC-ID	ITEM	FAILURE MODE
02-1-028-2	30121	BRAKE / SKID CONTROL	JAMMED OPEN
02-1-029-2	30131	BY - PASS VALVE, HYD	FAILS TO OPEN
02-1-030-1	30112	INLET FILTER, HYD MOD	CLOGGED FILTER
02-1-030-1	30130	INLET FILTER, HYD MOD	FILTER CLOGGED
02-1-032-2	30102	BRAKE PEDAL TRANSDUCE	NO TRANSDUCER DEF
02-1-033-2	30111	HYD PRESS REG (SYS 2	FAILS CLOSED
02-1-033-2	30129	HYD PRESS REG (SYS 1)	FAILS CLOSED
02-1-034-1	21301	MLG DOOR BOOSTER BUNG	FAILS TO FUNCTION
02-1-044-1	30125	RUDDER / BRAKE PEDAL	STRUCTURAL FAILUR
02-1-044-1	30126	RUDDER / BRAKE PEDAL	BINDING / JAMMING
02-1-050-1	30123	EXCITER RING - WHEEL	SHORT OR NO INPUT
02-1-051-1	30123	EXCITER RING - WHEEL	SHORT OR NO INPUT
02-1-051-2	30123	EXCITER RING - WHEEL	SHORT OR NO INPUT
02-1-053-1	30123	EXCITER RING - WHEEL	SHORT OR NO INPUT
02-1-066-2	30124	STATORS, ROTORS, CLIP	STRUCTURAL FAILUR
02-1-075-1	10215	SHOCK STRUT	INTERNAL / EXTERN
02-1-075-1	10216	SHOCK STRUT	INTERNAL / EXTERN
02-1-076-1	10211	TORQUE ARM ASSEMBLY	STRUCTURAL FAILUR
02-1-077-1	10202	DRAG BRACE	STRUCTURAL FAILUR
02-1-077-1	10203	DRAG BRACE TRUNION	STRUCTURAL FAILUR
02-1-077-1	10221	DRAG BRACE	STRUCTURAL FAILUR
02-1-077-1	10222	DRAG BRACE TRUNION	STRUCTURAL FAILUR
02-1-077-1	10223	DRAG BRACE TRUNION	STRUCTURAL FAILUR
02-1-077-1	10224	SUPPORT BEAM	STRUCTURAL FAILUR
02-1-078-1	10204	LOCK BRACE ASSEMBLY	STRUCTURAL FAILUR
02-1-079-1	10205	DOWNLOCK BUNGEE	PHYSICAL BINDING
02-1-079-1	10206	DOWNLOCK BUNGEE	STRUCTURAL FAILUR
02-1-080-1	10214	WEIGHT ON WHEELS SENS	ERONEOUS OUTPUT
02-1-082-1	10217	UPLOCK ROLLER RETAINI	STRUCTURAL FAILUR
02-1-082-1	10701	NLG UPLOCK HOOK ASSEM	STRUCTURAL FAILUR
02-1-083-1	10217	UPLOCK ROLLER RETAINI	STRUCTURAL FAILUR
02-1-085-1	10209	STEERING COLLAR ASSEM	STRUCTURAL FAILUR
02-1-097-1	11102	NLG B/U PYRO UPLOCK	FAIL TO FIRE
02-1-097-2	11101	NLG B/U PYRO UPLOCK	INADVERTANT FIRIN
02-1-098-1	10501	NLG DOOR EXTEND / RET	STRUCTURAL FAILUR
02-1-098-1	10901	NLG DOOR HOOK ACT LIN	STRUCTURAL FAILUR
02-1-099-1	10501	NLG DOOR EXTEND / RET	STRUCTURAL FAILUR
02-1-099-1	10901	NLG DOOR HOOK ACT LIN	STRUCTURAL FAILUR
02-1-102-1	11301	NLG DOOR BUNGEE ASSIS	STRUCTURAL FAILUR
02-1-104-1	11202	NLG EXTENSION BOOSTER	FAIL TO FIRE
02-1-104-2	11201	NLG EXTENSION BOOSTER	INADVERTANT FIRIN
02-1-109-1	10201	NOSE LANDING GEAR TRU	STRUCTURAL FAILUR
02-1-110-1	10101	TIRES, NLG TYPE II	RUPTURE
02-6-G08-A01	21005	MLG UPLOCK ACTUATOR	RUPTURE
02-6-G08-A02	21003	MLG UPLOCK ACTUATOR	LEAK EXTERNAL
02-6-G09-A01	20404	MLG EXTEND / RETRACT	RUPTURE
02-6-G09-A02	20402	MLG EXTEND / RETRACT	LEAK, EXTERNAL
02-6-G09-A04	20416	MLG EXTEND / RETRACT	TEMPERATURE TRANS
02-6-G09-B02	20413	MLG EXTEND / RETRACT	SHUTTLE VALVE
02-6-G09-F01	20414	MLG EXTEND / RETRACT	TIMING ORIFICE

NASA FMEA	MDAC-ID	ITEM	FAILURE MODE
02-6-H01-A01	10404	NLG EXTEND / RETRACT	RUPTURE
02-6-H01-A02	10402	NLG EXTEND / RETRACT	LEAK, EXTERNAL
02-6-H01-A04	10416	NLG EXTEND / RETRACT	TEMPERATURE TRANS
02-6-H01-B02	10413	NLG EXTEND / RETRACT	SHUTTLE VALVE
02-6-H01-F01	10414	NLG EXTEND / RETRACT	TIMING ORIFICE
02-6-H03-1	11005	NLG UPLOCK ACTUATOR	RUPTURE
02-6-H03-2	11003	NLG UPLOCK ACTUATOR	LEAK EXTERNAL
05-6BA-200200-1	31166	HYBRID DRIVER CONTROL	INADVERTENT OUTPU
05-6BA-2115-1	31113	PUSHBUTTON SWITCH (2)	FAILS OPEN
05-6BA-2117-1	31117	PUSHBUTTON SWITCH, LD	FAILS OPEN
05-6BA-2118-4	31101	PROXIMITY SENSOR BOX	INADVERTENT OUTPU
05-6BA-2206-1	31107	BLOCKING DIODE (2) 12	ELEMENT OPENS
05-6BA-2207-1	31109	BLOCKING DIODE (2) 12	ELEMENT OPENS
05-6BA-2244-2	31123	CIRCUIT BREAKER (3 AM	FAILS OPEN
05-6BA-2300-1	31126	GENERAL PURPOSE FUSE	OPENS, PREMATURE
05-6BA-2301-1	31127	GENERAL PURPOSE FUSE	OPENS, PREMATURE
05-6BA-2302-1	31125	GENERAL PURPOSE FUSE	OPENS, PREMATURE
05-6BA-2303-1	31137	GENERAL PURPOSE FUSE	OPENS, PREMATURE
05-6BA-2356-1	31128	RESISTOR (12)	ELEMENT OPENS
05-6BA-2401-3	31149	HYBRID DRIVER CONTROL	INADVERTENT OUTPU
05-6BA-2402-3	31144	HYBRID DRIVER CONTROL	INADVERTENT OUTPU
05-6BA-2407-1	31158	HYBRID DRIVER CONTROL	LOSS OF OUTPUT
05-6BA-2407-2	31158	HYBRID DRIVER CONTROL	LOSS OF OUTPUT
05-6BA-2408-1	31156	HYBRID DRIVER CONTROL	LOSS OF OUTPUT
05-6BA-2409-1	31154	HYBRID DRIVER CONTROL	LOSS OF OUTPUT
05-6BA-2410-1	31159	HYBRID DRIVER CONTROL	LOSS OF OUTPUT
05-6BA-2410-2	31160	HYBRID DRIVER CONTROL	INADVERTENT OUTPU
05-6BA-2413-1	31152	HYBRID DRIVER CONTROL	LOSS OF OUTPUT
05-6BA-2413-2	31153	HYBRID DRIVER CONTROL	INADVERTENT OUTPU
05-6BA-2415-1	31150	HYBRID DRIVER CONTROL	LOSS OF OUTPUT
05-6BA-2415-2	31151	HYBRID DRIVER CONTROL	INADVERTENT OUTPU
05-6BA-2501-1	31168	LATCHING RELAY (6)	FAILS OPEN
05-6BA-2502-1	31170	LATCHING RELAY (6)	FAILS OPEN
05-6BA-2502-3	31171	LATCHING RELAY (6)	FAILS CLOSED
05-6BA-2576-1	31177	PYRO INITIATOR CNTRL	LOSS OF OUTPUT
05-6BA-2578-1	31183	DIODE, 12 AMP	ELEMENT OPENS
05-6BB-2096-1	31213	GENERAL PURPOSE RELAY	FAILS OPEN
05-6BB-2096-2	31213	GENERAL PURPOSE RELAY	FAILS OPEN
05-6BB-2096-3	31214	GENERAL PURPOSE RELAY	FAILS CLOSED
05-6BB-2102-2	31212	BLOCKING DIODE (4)	SHORTS
05-6BB-2107-1	31220	TOGGLE SWITCH, DPST	FAILS OPEN
05-6BB-2107-2	31220	TOGGLE SWITCH, DPST	FAILS OPEN
05-6BB-2240-1	31201	GENERAL PURPOSE FUSE	OPENS, PREMATURE
05-6BB-2241-1	31205	GENERAL PURPOSE FUSE	OPENS, PREMATURE
05-6BB-2246-1	31202	FUSE (5 AMP), 2	OPENS, PREMATURE
05-6BB-2249-1	31210	CURRENT LIMITING RESI	ELEMENT OPENS
05-6BB-2262-1	31216	HYBRID DRIVER CONTROL	LOSS OF OUTPUT
05-6BB-2262-2	31217	HYBRID DRIVER CONTROL	INADVERTENT OUTPU
05-6BB-2270-1	31240	RESISTOR (1 OHM)(2W)	OPEN (ELECTRICAL)
NONE	10210	STEERING DISCONNECT	STRUCTURAL FAILUR
NONE	10212	NOSE WHEEL RETAINING	CORROSION, STRUCT

NASA FMEA	MDAC-ID	ITEM	FAILURE MODE
-----	-----	-----	-----
NONE	10213	AXLE	CORROSION, STRUCT
NONE	10220	TORQUE TUBE ASSEMBLY	STRUCTURAL FAILUR
NONE	10401	NLG EXTEND / RETRACT	BROKEN ROD / LINK
NONE	10601	NLG DOOR OVER-CENTER	STRUCTURAL FAILUR
NONE	11004	NLG UPLOCK ACTUATOR	BROKEN ROD / LINK
NONE	11302	NLG DOOR BUNGEE ASSIS	STRUCTURAL FAILUR
NONE	20205	AXLE KIT - MLG	STRUCTURAL FAILUR
NONE	20214	WEIGHT ON WHEELS SENS	SHORTED OPEN
NONE	20215	WEIGHT ON WHEELS SENS	SHORTED CLOSED
NONE	20401	MLG EXTEND / RETRACT	BROKEN ROD / LINK
NONE	20601	DOOR OVER-CENTER BUNG	STRUCTURAL FAILUR
NONE	21004	MLG UPLOCK ACTUATOR	BROKEN ROD / LINK
NONE	30104	BRAKE PEDAL TRANSDUCE	SHORT / CLOSED
NONE	30127	TRANSDUCERS, SENSORS	FAILURE - NO DATA
NONE	31161	HYBRID DRIVER CONTROL	LOSS OF OUTPUT



## APPENDIX E DETAILED ANALYSIS

This appendix contains the IOA analysis worksheets supplementing previous results reported in STSEOS Working Paper 1.0-WP-VA86001-25, Analysis of the Landing/Deceleration Subsystem, (19 January 1987). Prior results were obtained independently and documented before starting the FMEA/CIL assessment activity. Supplemental analysis was performed to address failure modes not previously considered by the IOA. Each sheet identifies the hardware item being analyzed, parent assembly and function performed. For each failure mode possible causes are identified, and hardware and functional criticality for each mission phase are determined as described in NSTS 22206, Instructions for Preparation of FMEA and CIL, 10 October 1986. Failure mode effects are described at the bottom of each sheet and worst case criticality is identified at the top.

### LEGEND FOR IOA ANALYSIS WORKSHEETS

-----

#### Hardware Criticalities:

- 1 = Loss of life or vehicle
- 2 = Loss of mission or next failure of any redundant item (like or unlike) could cause loss of life/vehicle
- 3 = All others

#### Functional Criticalities:

- 1R = Redundant hardware items (like or unlike) all of which, if failed, could cause loss of life or vehicle.
- 2R = Redundant hardware items (like or unlike) all of which, if failed, could cause loss of mission.

#### Redundancy Screen A:

- 1 = Is Checked Out PreFlight
- 2 = Is Capable of Check Out PreFlight
- 3 = Not Capable of Check Out PreFlight
- NA = Not Applicable

#### Redundancy Screens B and C:

- P = Passed Screen
- F = Failed Screen
- NA = Not Applicable

INDEPENDENT ORBITER ASSESSMENT  
ORBITER SUBSYSTEM ANALYSIS WORKSHEET

DATE: 2/22/88 HIGHEST CRITICALITY HDW/FUNC  
SUBSYSTEM: LANDING/DECELERATION SYSTEMS FLIGHT: 2/1R  
MDAC ID: 31240 ABORT: 2/1R

ITEM: RESISTOR (1 OHM) (2W) ANTI-SKID VLV COIL CURRENT  
MEASUREMENT  
FAILURE MODE: OPEN (ELECTRICAL)

LEAD ANALYST: P. BYNUM SUBSYS LEAD: SCHMECKPEPER

BREAKDOWN HIERARCHY:

- 1) BRAKE AND ANTI-SKID
- 2) FWD LCA-1,2,3
- 3) RESISTOR (1 OHM) (2W) CURRENT SENSORS (4)
- 4)
- 5)
- 6)
- 7)
- 8)
- 9)

FLIGHT PHASE	HDW/FUNC	CRITICALITIES ABORT	HDW/FUNC
PRELAUNCH:	/NA	RTLS:	/NA
LIFTOFF:	/NA	TAL:	2/1R
ONORBIT:	/NA	AOA:	2/1R
DEORBIT:	2/1R	ATO:	2/1R
LANDING/SAFING:	2/1R		

REDUNDANCY SCREENS: A [ P ] B [ P ] C [ P ]

LOCATION: 40V51A2  
PART NUMBER:

CAUSES: MECHANICAL SHOCK, STRUCTURAL FAILURE, THERMAL SHOCK,  
VIBRATION

EFFECTS/RATIONALE:  
POSSIBLE LOSS OF ANTI-SKID PROTECTION DUE TO LOCKED WHEEL/BRAKE.

REFERENCES:



## APPENDIX F

### NASA FMEA TO IOA WORKSHEET CROSS REFERENCE/RECOMMENDATIONS

This section provides a cross reference between the NASA FMEA and corresponding IOA analysis worksheet(s) included in Appendix E. The Appendix F identifies: NASA FMEA Number, IOA Assessment Number, NASA criticality and redundancy screen data, and IOA recommendations.

#### Appendix F Legend

##### Code Definition

1. IOA recommends downgrading the criticality.
2. IOA recommends upgrading the criticality.
3. IOA recommends changing the effects field.
4. IOA concurs with NASA's evaluation of the FMEA.
5. IOA generated a non-credible failure mode.
6. IOA recommends that the FMEA be deleted because it is not a credible failure for that particular component or components.
7. IOA recommends changing NASA's redundancy screen fields to conform to NSTS 22206
8. IOA recommends generating a new FMEA for an uncovered component and/or failure mode.
9. IOA recommends combining FMEA's together that are criticality 3, to conform to NSTS 22206.
10. NASA's FMEA/CIL revaluation deleted this failure mode as a non-credible failure.
11. IOA recommends deletion of this item from the CIL.
12. IOA recommends the addition of this item to the CIL listing.
13. NASA transferred the FMEA/CIL to another subsystem.

ORIGINAL PAGE IS  
OF POOR QUALITY

APPENDIX F

NASA FMEA TO IDA WORKSHEET CROSS REFERENCE / RECOMMENDATIONS

IDENTIFIERS		NASA			IDA RECOMMENDATIONS *						
NASA	IDA	CRIT	SCREENS			CRIT	SCREENS			OTHER	ISSUE
FMEA NUMBER	ASSESSMENT NUMBER	HW/F	A	B	C	HW/F	A	B	C	(SEE LEGEND CODE)	
02-1-001-1	LDGDEC-20201	1/1	NA	NA	NA	/				8.12	
02-1-001-2	LDGDEC-20202	3/3	NA	NA	NA	1/1	NA	NA	NA	8.12	X
	LDGDEC-20203	3/3	NA	NA	NA	3/1R	NA	NA	NA	2	X
02-1-002-1	LDGDEC-20101	1/1	NA	NA	NA	1/1	NA	NA	NA	4	
02-1-003-1	LDGDEC-20206	1/1	NA	NA	NA	1/1	NA	NA	NA	8	X
	LDGDEC-20207	1/1	NA	NA	NA	1/1	NA	NA	NA	8.12	X
	LDGDEC-20219	1/1	NA	NA	NA	1/1	NA	NA	NA	8.12	X
	LDGDEC-20220	1/1	NA	NA	NA	1/1	NA	NA	NA	8.12	X
	LDGDEC-20221	1/1	NA	NA	NA	1/1	NA	NA	NA	8.12	X
02-1-004-1	LDGDEC-20204	1/1	NA	NA	NA	/				4	
02-1-005-1	LDGDEC-20208	1/1	NA	NA	NA	/				4	
	LDGDEC-20222	1/1	NA	NA	NA	1/1	NA	NA	NA	8.12	X
02-1-006-1	LDGDEC-20216	1/1	NA	NA	NA	/				4	
02-1-007-1	LDGDEC-20211	3/3	NA	NA	NA	/				4	
02-1-008-1	LDGDEC-20209	3/1R	P	NA	P	1/1	NA	NA	NA	2,8.12	X
	LDGDEC-20210	3/1R	P	NA	P	1/1	NA	NA	NA	2,8.12	X
02-1-009-1	LDGDEC-20212	3/3	NA	NA	NA	/				4	
	LDGDEC-20213	3/3	NA	NA	NA	/				4	
02-1-010-1	LDGDEC-20217	1/1	NA	NA	NA	/				4	
02-1-012-1	LDGDEC-20501	1/1	NA	NA	NA	/				4	
	LDGDEC-20701	1/1	NA	NA	NA	/				4	
	LDGDEC-20901	1/1	NA	NA	NA	/				4	
02-1-013-1	LDGDEC-20501A	1/1	NA	NA	NA	/				4	
	LDGDEC-20701A	1/1	NA	NA	NA	/				4	
	LDGDEC-20901A	1/1	NA	NA	NA	/				4	
02-1-014-1	LDGDEC-20501B	1/1	NA	NA	NA	/				4	
	LDGDEC-20701B	1/1	NA	NA	NA	/				4	
	LDGDEC-20901B	1/1	NA	NA	NA	/				4	
02-1-015-1	LDGDEC-21101	1/1	NA	NA	NA	/				4,13	
02-1-015-2	LDGDEC-21102	2/1R	NA	NA	NA	2/1R	N	N	N	2,13	
02-1-017-1	LDGDEC-20218	1/1	NA	NA	NA	/				4	
02-1-018-1	LDGDEC-20223	1/1	NA	NA	NA	/				4	
02-1-019-1	LDGDEC-20501C	1/1	NA	NA	NA	/				4	
	LDGDEC-20701C	1/1	NA	NA	NA	/				4	
	LDGDEC-20901C	1/1	NA	NA	NA	/				4	
02-1-020-1	LDGDEC-20501D	1/1	NA	NA	NA	/				4	
	LDGDEC-20701D	1/1	NA	NA	NA	/				4	
	LDGDEC-20901D	1/1	NA	NA	NA	/				4	
02-1-021-1	LDGDEC-20501E	1/1	NA	NA	NA	/				4	
	LDGDEC-20701E	1/1	NA	NA	NA	/				4	
	LDGDEC-20901E	1/1	NA	NA	NA	/				4	
02-1-022-1	LDGDEC-30114	3/1R	P	P	P	/				4	
02-1-022-2	LDGDEC-30113	2/1R	P	P	P	3/1R	P	P	P	1,11	
02-1-023-1	LDGDEC-30119	3/1R	P	P	P	3/3	NA	NA	NA	1	
02-1-023-2	LDGDEC-30117	1/1	NA	NA	NA	/				4	

ORIGINAL PAGE IS  
OF POOR QUALITY

IDENTIFIERS		NASA			IOA RECOMMENDATIONS					ISSUE	
NASA FMEA NUMBER	IOA ASSESSMENT NUMBER	CRIT H4/F	SCREENS A B C			CRIT H4/F	SCREENS A B C				OTHER (SEE LEGEND CODE)
02-1-023-3	LDGDEC-30118	1/1	NA	NA	NA	3/1R	P	P	P	1	
02-1-024-1	LDGDEC-30101	3/1R	P	P	P	3/3	NA	NA	NA	1	
02-1-025-1	LDGDEC-30106	3/1R	P	P	P	/				4	
02-1-025-2	LDGDEC-30105	2/1R	P	F	P	1/1	NA	NA	NA	2	X
02-1-026-1	LDGDEC-30108	3/1R	P	F	P	3/3	NA	NA	NA	1	
02-1-026-2	LDGDEC-30107	2/1R	P	F	P	3/3	NA	NA	NA	11	
02-1-027-1	LDGDEC-30109	2/1R	P	F	P	3/3	NA	NA	NA	1.11	
02-1-027-2	LDGDEC-30109A	2/1R	P	F	P	3/3	NA	NA	NA	1.11	
02-1-028-1	LDGDEC-30122	3/1R	P	P	P	/				4	
02-1-028-2	LDGDEC-30121	1/1	NA	NA	NA	/				4	
02-1-028-3	LDGDEC-30120	3/1R	P	P	P	/				4	
02-1-029-1	LDGDEC-30115	3/3	NA	NA	NA	/				4	
02-1-029-2	LDGDEC-30116	3/3	NA	NA	NA	2/1R	P	F	F	8.12	X
	LDGDEC-30131	3/3	NA	NA	NA	2/1R	P	P	P	2.3,8.12	X
02-1-030-1	LDGDEC-30112	3/1R	P	P	P	2/1R	P	P	P	2,8.12	X
	LDGDEC-30130	3/1R				2/1R	P	F	P	2.3,8.12	X
02-1-032-1	LDGDEC-30103	3/1R	P	P	P	/				4	
02-1-032-2	LDGDEC-30102	2/1R	P	P	P	/				4	
02-1-033-1	LDGDEC-30110	3/1R	P	P	P	/				4	
02-1-033-2	LDGDEC-30111	3/1R	P	P	P	2/1R	P	P	P	2,8.12	X
	LDGDEC-30129	3/1R	P	P	P	2/1R	P	F	P	2.3	X
02-1-034-1	LDGDEC-21301	1/1	NA	NA	NA	/				4	
02-1-044-1	LDGDEC-30125	1/1	NA	NA	NA	/				4	
	LDGDEC-30126	1/1	NA	NA	NA	/				4	
02-1-050-1	LDGDEC-30123	2/1R	P	P	P	3/3	NA	NA	NA	1.11	
02-1-051-1	LDGDEC-30123A	2/1R	P	P	P	3/3	NA	NA	NA	1.11	
02-1-051-2	LDGDEC-30123B	2/1R	P	P	P	3/3	NA	NA	NA	1.11	
02-1-053-1	LDGDEC-30123C	2/1R	P	P	P	3/3	NA	NA	NA	1.11	
02-1-066-2	LDGDEC-30124	3/1R	P	P	P	1/1	P	P	P	8.12	X
02-1-075-1	LDGDEC-10215	1/1	NA	NA	NA	/				4	
	LDGDEC-10216	1/1	NA	NA	NA	3/1R	NA	NA	NA	8.12	
02-1-076-1	LDGDEC-10211	3/1R	P	P	P	1/1	NA	NA	NA	2	X
02-1-077-1	LDGDEC-10202	1/1	NA	NA	NA	1/1	NA	NA	NA	8.12	X
	LDGDEC-10203	1/1	NA	NA	NA	1/1	NA	NA	NA	8.12	X
	LDGDEC-10221	1/1	NA	NA	NA	/				8.12	X
	LDGDEC-10222	1/1	NA	NA	NA	1/1	NA	NA	AN	8.12	X
	LDGDEC-10223	1/1	NA	NA	NA	1/1	NA	NA	AN	8.12	X
	LDGDEC-10224	1/1	NA	NA	NA	1/1	NA	NA	AN	8.12	X
02-1-078-1	LDGDEC-10204	1/1	NA	NA	NA	/				4	
02-1-079-1	LDGDEC-10205	3/1R	P	NA	P	1/1	NA	NA	NA	8.12	X
	LDGDEC-10206	3/1R	P	NA	P	1/1	NA	NA	NA	8.12	X
02-1-080-1	LDGDEC-10214	3/1R	P	NA	P	3/3	P	P	P	1	
	LDGDEC-10218	3/1R	P	NA	P	3/3	P	P	P	1	
	LDGDEC-10219	3/1R	P	NA	P	3/3	P	P	P	1	
02-1-081-1	LDGDEC-10207	3/3	NA	NA	NA	/				4	
	LDGDEC-10208	3/3	NA	NA	NA	/				4	
02-1-082-1	LDGDEC-10217	1/1	NA	NA	NA	/				4	
	LDGDEC-10701	1/1	NA	NA	NA	/				4	
02-1-083-1	LDGDEC-10217A	1/1	NA	NA	NA	1/1	NA	NA	NA	8.12	
02-1-085-1	LDGDEC-10209	3/1R	P	P	P	/				4	

ORIGINAL PAGE IS  
OF POOR QUALITY

IDENTIFIERS		NASA			IOA RECOMMENDATIONS					ISSUE	
NASA	IOA	CRIT	SCREENS			CRIT	SCREENS				OTHER
FMEA NUMBER	ASSESSMENT NUMBER	HW/F	A	B	C	HW/F	A	B	C	(SEE LEGEND CODE)	
02-1-097-1	LDGDEC-11102	2/1R	NA	NA	NA	1/1	NA	NA	NA	2.13	
02-1-097-2	LDGDEC-11101	1/1	NA	NA	NA	1/1				13	
02-1-098-1	LDGDEC-10501	1/1	NA	NA	NA	1/1				4	
	LDGDEC-10901	1/1	NA	NA	NA	1/1				4	
02-1-099-1	LDGDEC-10501A	1/1	NA	NA	NA	1/1				9	
	LDGDEC-10901A	1/1	NA	NA	NA	1/1	NA	NA	NA	3.12	
02-1-102-1	LDGDEC-11301	1/1	NA	NA	NA	1/1				4	
02-1-104-1	LDGDEC-11202	1/1	NA	NA	NA	1/1				4.13	
02-1-104-2	LDGDEC-11201	1/1	NA	NA	NA	1/1				4.13	
02-1-109-1	LDGDEC-10201	1/1	NA	NA	NA	1/1	NA	NA	NA	B.12	X
02-1-110-1	LDGDEC-10101	1/1	NA	NA	NA	1/1				4	
02-6-608-A01	LDGDEC-21005	2/1R	P	P	P	1/1		F		3	
02-6-608-A02	LDGDEC-21003	2/1R	P	P	P	1/1		F		3.4	
02-6-608-A03	LDGDEC-21006	3/3	NA	NA	NA	1/1				4	
02-6-609-A01	LDGDEC-20404	1/1	NA	NA	NA	1/1				4	
02-6-609-A02	LDGDEC-20402	3/3	NA	NA	NA	2/1R	P	F	P	2.9.12	X
02-6-609-A03	LDGDEC-20403	3/3	NA	NA	NA	1/1				4	
02-6-609-A04	LDGDEC-20416	2/1R	P	P	P	2/1R	P	F	P	3	
02-6-609-B01	LDGDEC-20412	3/3	NA	NA	NA	1/1				4	
02-6-609-B02	LDGDEC-20413	1/1	NA	NA	NA	1/1				4	
02-6-609-C01	LDGDEC-20411	3/3	NA	NA	NA	1/1				4	
02-6-609-D01	LDGDEC-20409	3/3	NA	NA	NA	1/1				4	
02-6-609-D02	LDGDEC-20410	3/3	NA	NA	NA	1/1				4	
02-6-609-E01	LDGDEC-20408	3/3	NA	NA	NA	1/1				4	
02-6-609-E02	LDGDEC-20408A	3/3	NA	NA	NA	1/1				4	
02-6-609-F01	LDGDEC-20414	1/1	NA	NA	NA	1/1				4	
02-6-609-G01	LDGDEC-20406	3/3	NA	NA	NA	1/1				4	
02-6-609-H01	LDGDEC-20415	3/3	NA	NA	NA	1/1				4	
02-6-609-H02	LDGDEC-20415A	3/3	NA	NA	NA	1/1				4	
02-6-609-J01	LDGDEC-20407	3/3	NA	NA	NA	1/1				4	
02-6-609-J02	LDGDEC-20407A	3/3	NA	NA	NA	1/1				4	
02-6-609-K01	LDGDEC-20405	3/3	NA	NA	NA	1/1				4	
02-6-612-1	LDGDEC-30128	3/1R	P	P	P	1/1				4	
02-6-H01-A01	LDGDEC-10404	1/1	NA	NA	NA	1/1				4	
02-6-H01-A02	LDGDEC-10402	3/3	NA	NA	NA	2/1R	P	F	P	2	
02-6-H01-A03	LDGDEC-10403	3/3	NA	NA	NA	1/1				4	
02-6-H01-A04	LDGDEC-10416	2/1R	P	P	P	2/1R	P	F	P	3	
02-6-H01-B01	LDGDEC-10412	3/3	NA	NA	NA	1/1				4	
02-6-H01-B02	LDGDEC-10413	1/1	NA	NA	NA	1/1				4	
02-6-H01-C01	LDGDEC-10411	3/3	NA	NA	NA	1/1				4	
02-6-H01-D01	LDGDEC-10409	3/3	NA	NA	NA	1/1				4	
02-6-H01-D02	LDGDEC-10410	3/3	NA	NA	NA	1/1				4	
02-6-H01-E01	LDGDEC-10408	3/3	NA	NA	NA	1/1				4	
02-6-H01-E02	LDGDEC-10408A	3/3	NA	NA	NA	1/1				4	
02-6-H01-F01	LDGDEC-10414	1/1	NA	NA	NA	1/1				4	
02-6-H01-G01	LDGDEC-10406	3/3	NA	NA	NA	1/1				4	
02-6-H01-H01	LDGDEC-10415	3/3	NA	NA	NA	1/1				4	
02-6-H01-H02	LDGDEC-10415A	3/3	NA	NA	NA	1/1				4	
02-6-H01-J01	LDGDEC-10407	3/3	NA	NA	NA	1/1				4	
02-6-H01-J02	LDGDEC-10407A	3/3	NA	NA	NA	1/1				4	

ORIGINAL PAGE IS  
OF POOR QUALITY

IDENTIFIERS		NASA			IDA RECOMMENDATIONS *						
NASA	IDA	CRIT	SCREENS			CRIT	SCREENS			OTHER	ISSUE
FMEA NUMBER	ASSESSMENT NUMBER	HW/F	A	B	C	HW/F	A	B	C	(SEE LEGEND CODE)	
02-6-H01-K01	LDGDEC-10405	3/3	NA	NA	NA	/				4	
02-6-H03-1	LDGDEC-11005	2/1R	P	P	P	2/1R	P	F	P	3	
02-6-H03-2	LDGDEC-11003	2/1R	NA	NA	NA	2/1R	P	F	P	8.12	
02-6-H03-3	LDGDEC-11006	3/3	NA	NA	NA	/				4	
05-68A-2401-1	LDGDEC-31147	3/1R	P	P	P	/				4	
05-68A-2401-2	LDGDEC-31147A	3/1R	P	P	P	/				4	
05-2106-1	LDGDEC-31224	3/1R	P	P	P	/	P	F	P	4	
05-2409-2	LDGDEC-31155	3/3				/				4	
05-68A-200200-1	LDGDEC-31163	3/1R	P	P	P	/				6.10	
	LDGDEC-31164	3/1R	P	P	P	/				6.10	
	LDGDEC-31165	3/1R	P	P	P	/				6.10	
	LDGDEC-31166	3/1R	P	P	P	/		F		6.10	
	LDGDEC-31181	3/1R	P	P	P	/				6.10	
05-68A-2113-1	LDGDEC-31119	3/3				/				4	
05-68A-2115-1	LDGDEC-31113	2/1R	P	P	P	/				4	
05-68A-2115-2	LDGDEC-31114A	/				/				5.10	
05-68A-2115-3	LDGDEC-31114	1/1				3/3				5.11	X
05-68A-2116-1	LDGDEC-31115	3/3				/				4	
05-68A-2116-2	LDGDEC-31115A	3/3				/				5.10	
05-68A-2116-3	LDGDEC-31115B	2/1R	P	F	P	3/3				1.9,11	X
05-68A-2117-1	LDGDEC-31117	2/1R	P	P	P	3/1R	P	F	P	1.7	X
05-68A-2117-2	LDGDEC-31118A	3/1R	NA	NA	NA	/				6.10	
05-68A-2117-3	LDGDEC-31118	2/1R	P	P	P	3/3				1.11	X
05-68A-2118-4	LDGDEC-31101	2/1R	P	P	P	/				4	
05-68A-2204-1	LDGDEC-31100	3/3				/				4	
05-68A-2204-2	LDGDEC-31100A	3/3				/				4	
05-68A-2205-1	LDGDEC-31105	3/3				/				4	
05-68A-2205-2	LDGDEC-31105A	3/1R	F	F	P	3/3				1.11	X
05-68A-2206-1	LDGDEC-31107	3/1R	P	F	P	/				4	
05-68A-2206-2	LDGDEC-31108	3/3				/				4	
05-68A-2207-1	LDGDEC-31109	3/1R	P	F	P	/				4	
05-68A-2207-2	LDGDEC-31110	3/3				/				4	
05-68A-2208-1	LDGDEC-31122	3/3				/				4	
05-68A-2208-2	LDGDEC-31122A	3/3				/				4	
05-68A-2209-1	LDGDEC-31121	3/1R	P	P	P	/				4	
05-68A-2209-2	LDGDEC-31120	3/3				/				4	
05-68A-2243-1	LDGDEC-31111	3/1R	P	P	P	/				4	
05-68A-2243-2	LDGDEC-31112	3/3				/				4	
05-68A-2244-1	LDGDEC-31123	3/3				/				4	
05-68A-2244-2	LDGDEC-31123A	2/1R	P	P	P	/				4	
05-68A-2300-1	LDGDEC-31126	3/1R	P	F	P	/				4	
05-68A-2300-2	LDGDEC-31126A	3/3				/				5.10	
05-68A-2301-1	LDGDEC-31127	3/1R	P	F	P	/				4	
05-68A-2301-2	LDGDEC-31127A	3/3				/				5.10	
05-68A-2302-1	LDGDEC-31125	2/1R	P	P	P	3/1R	P	F	P	1.7	X
05-68A-2303-1	LDGDEC-31127	2/1R	P	P	P	/		F		7	X
05-68A-2303-2	LDGDEC-31137A	3/3				/				6.10	
05-68A-2351-1	LDGDEC-31130	3/3				/				4	
05-68A-2351-2	LDGDEC-31130A	3/3				/				6.10	
05-68A-2352-1	LDGDEC-31132	3/3				/				4	

ORIGINAL PAGE IS  
OF POOR QUALITY

IDENTIFIERS		NASA			IDA RECOMMENDATIONS						
NASA	IDA	CRIT	SCREENS			CRIT	SCREENS			OTHER	ISSUE
FMEA NUMBER	ASSESSMENT NUMBER	HW/F	A	B	C	HW/F	A	B	C	(SEE LEGEND CODE)	
05-68A-2353-1	LDGDEC-31136	3/3				/				4	
05-68A-2354-1	LDGDEC-31134	3/3				/				4	
05-68A-2356-1	LDGDEC-31128	3/1R	P	F	P	/				4	
05-68A-2356-2	LDGDEC-31129	3/3				/				4	
05-68A-2357-1	LDGDEC-31138	3/3				/				4	
05-68A-2360-1	LDGDEC-31133	3/3				/				4	
05-68A-2361-1	LDGDEC-31135	3/3				/				4	
05-68A-2361-2	LDGDEC-31135A	3/3				/				4	
05-68A-2362-1	LDGDEC-31139	3/3				/				4	
05-68A-2362-2	LDGDEC-31139A	3/3				/				4	
05-68A-2363-1	LDGDEC-31131	3/3				/				4	
05-68A-2400-1	LDGDEC-31140	3/1R	P	P	P	/				4.13	
05-68A-2400-2	LDGDEC-31141	3/3				/				4.13	
05-68A-2401-3	LDGDEC-31149	2/1R	P	P	F	/				4	
05-68A-2402-1	LDGDEC-31143	3/1R	P	P	P	/				4	
05-68A-2402-2	LDGDEC-31143A	3/1R	P	P	P	/				4	
05-68A-2402-3	LDGDEC-31144	2/1R	P	P	P	/				4	
05-68A-2403-1	LDGDEC-31146	3/3				/				4.9	
05-68A-2407-2	LDGDEC-31148A	3/3				/				4.9	
05-68A-2404-1	LDGDEC-31142	3/3				/				4.9	
05-68A-2404-2	LDGDEC-31142A	3/3				/				4.9	
05-68A-2404-3	LDGDEC-31142B	3/3				/				6	X
05-68A-2405-1	LDGDEC-31146	3/3				/				4.9	
05-68A-2405-2	LDGDEC-31146A	3/3				/				4.9	
05-68A-2405-3	LDGDEC-31146B	3/3				/				6	X
05-68A-2406-1	LDGDEC-31145	3/1R	P	P	P	3/3				1.9	X
05-68A-2406-2	LDGDEC-31145A	3/3				/				4.9	
05-68A-2406-3	LDGDEC-31145B	3/3				/				6	X
05-68A-2407-1	LDGDEC-31158	2/1R	P	P	P	/				4	
05-68A-2407-2	LDGDEC-31158A	2/1R	P	P	P	/				4	
05-68A-2407-3	LDGDEC-31158B	3/3				/				6	X
05-68A-2408-1	LDGDEC-31156	3/1R	P	F	P	/				4	
05-68A-2408-2	LDGDEC-31157	3/3				/				4	
05-68A-2409-1	LDGDEC-31154	3/1R	P	P	P	/		F		7.12	X
05-68A-2410-1	LDGDEC-31159	3/1R	P	F	P	/				4	
05-68A-2410-2	LDGDEC-31160	3/1R	P	F	P	/				4	
05-68A-2413-1	LDGDEC-31152	2/1R	P	F	P	/				4	
05-68A-2413-2	LDGDEC-31153	3/1R	P	F	P	/				4	
05-68A-2415-1	LDGDEC-31150	2/1R	P	F	P	/				4	
05-68A-2415-2	LDGDEC-31151	3/1R	P	F	P	/				4	
05-68A-2501-1	LDGDEC-31166	2/1R	P	P	P	/				4	
05-68A-2501-2	LDGDEC-31168A	2/1R	P	P	P	/				6	X
05-68A-2501-3	LDGDEC-31169	3/1R	P	P	P	/				4	
05-68A-2502-1	LDGDEC-31170	2/1R	P	F	P	/				4	
05-68A-2502-2	LDGDEC-31170A	2/1R	F	P	P	/				6	X
05-68A-2502-3	LDGDEC-31171	2/1R	P	P	P	/				4	
05-68A-2503-1	LDGDEC-31172	3/1R	F	P	P	/				4	
05-68A-2503-2	LDGDEC-31172A	3/3				/				4	
05-68A-2503-3	LDGDEC-31172B	3/3				/				6	X
05-68A-2550-1	LDGDEC-31173	3/1R	P	P	P	/				4	

ORIGINAL PAGE IS  
OF POOR QUALITY

IDENTIFIERS		NASA			IOA RECOMMENDATIONS *							ISSUE
NASA	IOA	CRIT	SCREENS			CRIT	SCREENS			OTHER (SEE LEGEND CODE)		
FMEA NUMBER	ASSESSMENT NUMBER		HW/F	A	B		C	HW/F	A		B	
05-68A-2550-2	LDGDEC-31173A	3/1R	P	P	P	/				6	X	
05-68A-2550-3	LDGDEC-31174	3/3				/				4		
05-68A-2575-1	LDGDEC-31175	3/1R	P	P	P	/				4		
05-68A-2575-2	LDGDEC-31175	3/1R	P	P	P	/				4		
05-68A-2576-1	LDGDEC-31177	2/1R	P	P	P	/				4		
05-68A-2576-2	LDGDEC-31178	3/1R	P	P	P	/				4.13		
05-68A-2578-1	LDGDEC-31183	3/1R	P	P	P	2/1R	P	F	P	2.12	X	
05-68A-2578-2	LDGDEC-31184	3/3				/				4		
05-68A-2580-1	LDGDEC-31185	3/3				/				4		
05-68A-2580-2	LDGDEC-31185A	3/3				/				4		
05-68B-2096-1	LDGDEC-31213	2/1R	P	F	P	/				4		
05-68B-2096-2	LDGDEC-31213A	3/1R	P	F	P	/				6	X	
05-68B-2096-3	LDGDEC-31214	2/1R	P	F	P	/				4		
05-68B-2101-1	LDGDEC-31215	3/3				/				4		
05-68B-2102-1	LDGDEC-31211	3/1R	P	F	P	/				4		
05-68B-2102-2	LDGDEC-31212	3/1R	P	F	P	/				4		
05-68B-2106-1	LDGDEC-31222	3/1R	P	P	P	/				4		
05-68B-2106-2	LDGDEC-31222A	3/1R	P	P	P	/				6	X	
05-68B-2106-3	LDGDEC-31223	3/3				/				4		
	LDGDEC-31225	3/1R	P	P	P	3/3				1	X	
05-68B-2107-1	LDGDEC-31220	2/1R	P	P	P	/				4		
05-68B-2107-2	LDGDEC-31220A	2/1R	P	P	P	/				6	X	
05-68B-2107-3	LDGDEC-31221	3/1R	P	P	P	3/3				1	X	
05-68B-2111-1	LDGDEC-31203	3/3				/				4		
05-68B-2240-1	LDGDEC-31201	3/1R	P	F	P	/				4		
05-68B-2241-1	LDGDEC-31205	2/1R	P	P	P	3/1R	P	F	P	1.7	X	
05-68B-2242-1	LDGDEC-31204	3/3				/				4		
05-68B-2246-1	LDGDEC-31202	2/1R	P	P	P	/				4		
05-68B-2247-1	LDGDEC-31206	3/3				/				4		
05-68B-2248-1	LDGDEC-31200	3/3				/				4		
05-68B-2249-1	LDGDEC-31210	3/1R	P	P	P	/		F		7.12	X	
05-68B-2249-2	LDGDEC-31208	3/3				/				4		
05-68B-2250-1	LDGDEC-31209	3/3				/				4		
05-68B-2253-1	LDGDEC-31207	3/3				/				4		
05-68B-2256-1	LDGDEC-31226	3/1R	P	P	P	/				4		
05-68B-2256-2	LDGDEC-31227	3/3				/				4		
05-68B-2262-1	LDGDEC-31216	2/1R	P	F	P	/				4		
	LDGDEC-31218	3/1R	P	P	P	/				4		
05-68B-2262-2	LDGDEC-31217	3/1R	P	F	P	/				4		
	LDGDEC-31219	3/3				/				4		
05-68B-2270-1	LDGDEC-31240X	2/1R	P	P	P	/				4		
06-68A-2118-1	LDGDEC-31102	3/1R	P	P	P	/				4		
06-68A-2200-1	LDGDEC-31103	3/3				/				4		
06-68A-2200-2	LDGDEC-31103A	3/3				/				4		
06-68A-2201-1	LDGDEC-31104	3/3				/				4		
06-68A-2201-2	LDGDEC-31104A	3/3				/				4		
NCNE	LDGDEC-10210	/				1/1	NA	NA	NA	2.12	X	
	LDGDEC-10212	/				1/1	NA	NA	NA	5		
	LDGDEC-10213	/				1/1	NA	NA	NA	8.12	X	
	LDGDEC-10220	/				1/1	NA	NA	NA	8.12	X	

ORIGINAL PAGE IS  
OF POOR QUALITY

IDENTIFIERS		NASA			IOA RECOMMENDATIONS *					
NASA	IOA	CRIT			SCREENS			OTHER		ISSUE
EMEA NUMBER	ASSESSMENT NUMBER	HW/F	A	B	C	HW/F	A	B	C	(SEE LEGEND CODE)
NONE	LDGDEC-10401	/				3/3	NA	NA	NA	B
	LDGDEC-10601	/				3/3	NA	NA	NA	A
	LDGDEC-11004	/				2/1R	P	F	P	8.12
	LDGDEC-11302	/				1/1	NA	NA	NA	8.12
	LDGDEC-20245	/				1/1	NA	NA	NA	4.9
	LDGDEC-20214	/				3/3	NA	NA	NA	
	LDGDEC-20215	/				3/3	NA	NA	NA	
	LDGDEC-20401	/				3/3	NA	NA	NA	B
	LDGDEC-20601	/				3/3	NA	NA	NA	
	LDGDEC-21004	/				2/1R	P	F	P	9.12
	LDGDEC-30104	/				1/1	NA	NA	NA	8.12
	LDGDEC-30127	/				3/3	NA	NA	NA	5
	LDGDEC-31161	/				3/1R	P	F	P	8.12
	LDGDEC-31162	/				3/3				8
	LDGDEC-31167	/				3/3				8
	LDGDEC-31179	/				3/3				8
	LDGDEC-31180	/				3/3				8
	LDGDEC-31182	/				/				6.10
	LDGDEC-31228	/				3/3				8
	LDGDEC-31229	/				3/1R	P	P	P	8
	LDGDEC-31230	/				3/1R	P	P	P	8
	LDGDEC-31231	/				3/3				8
	LDGDEC-31232	/				3/1R	P	P	P	8
	LDGDEC-31233	/				3/3				9
	LDGDEC-31234	/				3/3				8
	LDGDEC-31235	/				3/3				9
	LDGDEC-31237	/				3/3				8
	LDGDEC-31239	/				3/3				9









**MCDONNELL DOUGLAS ASTRONAUTICS COMPANY –  
ENGINEERING SERVICES  
16055 SPACE CENTER BLVD, HOUSTON, TEXAS 77062**

Independent Orbiter Assessment  
Assessment of the Landing/Deceleration FMEA/CIL

M  
1.0 EXECUTIVE SUMMARY

The McDonnell Douglas Astronautics Company (MDAC) was selected in June 1986 to perform an Independent Orbiter Assessment (IOA) of the Failure Modes and Effects Analysis (FMEA) and Critical Items List (CIL). ~~Direction was given by the STS Orbiter and GFE Projects Office to perform the hardware analysis using the instructions and ground rules defined in NSTS 22206, Instructions for Preparation of FMEA and CIL, 10 October 1986.~~

The IOA effort first completed an analysis of the Landing / Deceleration (LDG/DEC) hardware, generating draft failure modes and potential critical items. To preserve independence, this analysis was accomplished without reliance upon the results contained within the NASA FMEA/CIL documentation. The IOA results were then compared to the NASA FMEA/CIL baseline with proposed Post 51-L updates included. A resolution of each discrepancy from the comparison is provided through additional analysis as required. This report documents the results of that comparison for the Orbiter LDG/DEC hardware.

The IOA product for the LDG/DEC analysis consisted of 259 failure mode "worksheets" that resulted in 124 potential critical items being identified. Comparison was made to the NASA baseline ~~(as of 19 November 1986)~~ which consisted of 267 FMEA's and 120 CIL items. ~~The comparison determined if there were any results which had been found by the IOA but were not in the NASA baseline.~~ This comparison produced agreement on all but 75 FMEA's which caused differences in 51 CIL items. ~~Figure 1 presents a comparison of the proposed Post 51-L NASA baseline, with the IOA recommended baseline, and any issues.~~

The issues arose due to differences between the NASA and IOA FMEA/CIL preparation instructions. NASA had used an older ground rules document which has since been superseded by the NSTS 22206 used by the IOA. After comparison, there were no discrepancies found that were not already identified by NASA, and the remaining issues may be attributed to differences in ground rules.



**NASA  
FORMAL  
REPORT**





due 3/16/89

responsible NASA Project Officer, Technical Monitor, or other appropriate presentations, reports, papers, and proceedings that contain scientific information. Explanations are on the back of this form and are presented in NHB 2200.2, "NASA Scientific and Technical Information Handbook."

☐ Original  
☐ Modified

(Facility Use Only)

Control No.

Date

on Sept 3/3/89

(PROJECT IDENTIFICATION (Information contained on report documentation page should not be repeated except title, date and contract number))

1 Independent Orbiter Assessment

Sponsoring NASA Organization: JSC VA

Sponsoring Organization (if different) McDonnell - Douglas

Contract/Grant/Interagency/Project Number(s) NAS9-17650

CR-185570

Document Date: Various

Document Number(s) Various

(For presentations or externally published documents, enter appropriate information on the intended publication such as name, place, and date of conference, periodical or journal title, or book title and publisher:

These documents must be routed to NASA Headquarters, International Affairs Division for approval. (See Section VII)

## II. AVAILABILITY CATEGORY

Check the appropriate category(ies):

Security Classification: ☐ Secret ☐ Secret RD ☐ Confidential ☐ Confidential RD ☒ Unclassified

Export Controlled Document - Documents marked in this block must be routed to NASA Headquarters International Affairs Division for approval.

☐ ITAR ☐ EAR

NASA Restricted Distribution Document

☐ FEDD ☐ Limited Distribution ☐ Special Conditions-See Section III

Document disclosing an invention

☐ Documents marked in this block must be withheld from release until six months have elapsed after submission of this form, unless a different release date is established by the appropriate counsel. (See Section IX).

Publicly Available Document

☒ Publicly available documents must be unclassified and may not be export-controlled or restricted distribution documents.☐ Copyrighted ☐ Not copyrighted

IN-16

21/6/89

## III. SPECIAL CONDITIONS

Check one or more of the applicable boxes in each of (a) and (b) as the basis for special restricted distribution if the "Special Conditions" box under NASA Restricted Distribution Document in Section II is checked. Guidelines are provided on reverse side of form.

a. This document contains:

☐ Foreign government information☐ Commercial product test or evaluation results☐ Preliminary information☐ Information subject to special contract provision☐ Other-Specify \_\_\_\_\_

b. Check one of the following limitations as appropriate:

☐ U.S. Government agencies and U.S. Government agency contractors only☐ NASA contractors and U.S. Government agencies only☐ U.S. Government agencies only☐ NASA personnel and NASA contractors only☐ NASA personnel only☐ Available only with approval of issuing office; \_\_\_\_\_

## IV. BLANKET RELEASE (OPTIONAL)

All documents issued under the following contract/grant/prc

The blanket release authorization granted \_\_\_\_\_ Date \_\_\_\_\_

☐ Rescinded - Future documents must have individual

Card Made

No Stock

processed as checked in Sections II and III.

Instructions for all documents processed in the STI system under the blanket be changed to conform to blocks as checked in Section II.

## V. PROJECT OFFICER/TECHNICAL MONITOR

Typed Name of Project Officer/Technical Monitor  
R. BERRY

Date

## VI. PROGRAM OFFICE REVIEW

☐ Appro

Typed Name of Program Office Representative

Program Office and Code

Signature

Date

## VII. INTERNATIONAL AFFAIRS DIVISION REVIEW

☐ Open, domestic conference presentation approved.☐ Export controlled limitation is not applicable.☐ Foreign publication/presentation approved.☐ The following Export controlled limitation (ITAR/EAR) is assigned to this document: \_\_\_\_\_☐ Export controlled limitation is approved.

International Affairs Div. Representative

Title

Date

## VIII. EXPIRATION OF REVIEW TIME

The document is being released in accordance with the availability category and limitation checked in Section II since no objection was received from the Program Office within 20 days of submission, as specified by NHB 2200.2, and approval by the International Affairs Division is not required.

Name &amp; Title

Office Code

Date

Note: This release procedure cannot be used with documents designated as Export Controlled Documents, conference presentations or foreign publications.

## IX. DOCUMENTS DISCLOSING AN INVENTION

a. This document may be released on \_\_\_\_\_ Date \_\_\_\_\_ Installation Patent or Intellectual Property Counsel \_\_\_\_\_ Date \_\_\_\_\_

b. This document was processed on \_\_\_\_\_ Date \_\_\_\_\_ In accordance with Sections II and III as applicable. NASA STI Facility \_\_\_\_\_ Date \_\_\_\_\_

## X. DISPOSITION

Completed forms should be forwarded to the NASA Scientific and Technical Information Facility, P.O. Box 8757, BWI Airport, Maryland 21240, with either (check box):

☐ Printed or reproducible copy of document enclosed☐ Abstract or Report Documentation Page enclosed. The issuing or sponsoring NASA installation should provide a copy of the document, when complete, to the NASA Scientific and Technical Information Facility at the above listed address.

